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Journal of Comparative Physiology A

The arm skeleton and shoulder girdle of *Falco peregrinus*, *Falco tinnunculus*, *Accipiter nisus*, and *Columba livia domestica* --Manuscript Draft--

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Full Title:	The arm skeleton and shoulder girdle of <i>Falco peregrinus</i> , <i>Falco tinnunculus</i> , <i>Accipiter nisus</i> , and <i>Columba livia domestica</i>	
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Abstract:	<p>During a high-speed dive peregrine falcons (<i>Falco peregrinus</i>) can reach a flight velocity of up to 320 kmh⁻¹. In consequence the bones of the wings and the shoulder girdle of peregrine falcons most likely experience large mechanical forces. We investigated the bones of the arm skeleton and the shoulder girdle of peregrine falcons. For comparison, we also investigated the comparable bones in European kestrels (<i>Falco tinnunculus</i>), sparrow hawks (<i>Accipiter nisus</i>) and pigeons (<i>Columba livia domestica</i>). The normalized bone mass of the entire arm skeleton and the shoulder girdle (coracoid, scapula, furcula) was significantly higher in <i>F. peregrinus</i> than in the other three species investigated. The midshaft cross-section of the humerus of <i>F. peregrinus</i> had the highest specific bending stiffness per body mass and the highest second moment of area. The mineral densities of the humerus, radius, ulna, and sternum were highest in <i>F. peregrinus</i>, indicating again a larger overall stability of these bones. Furthermore, the bones of the arm and shoulder were strongest in peregrine falcons. Computational fluid dynamics simulations suggest, that the forces that pull on the wings of a peregrine can reach up to three times the falcon's body mass at a stoop velocity of 80 ms⁻¹.</p>	
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Author Comments:	We would like to publish this MS about peregrine falcons in JCP. The MS is about numerical simulations of the forces acting on the wings of a diving	

	peregrine and morphological adaptations regarding the wing bones.
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**The arm skeleton and shoulder girdle of *Falco peregrinus*, *Falco tinnunculus*,
Accipiter nisus, and *Columba livia domestica***

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Keywords Bird flight, Bones, Bone mineral density, Computational fluid dynamics, Mechanical
forces

Abbreviations

BM Body mass

BMD Bone mineral density

CA Area of compact bone in a cross section

CFD Computation fluid dynamics

E Youngs modulus

HA Hydroxyapatite

HU Hounsfield unit

I Second moment of area

J Polar moment of area

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M Specific bending stiffness

SD standard deviation

Abstract

During a high-speed dive peregrine falcons (*Falco peregrinus*) can reach a flight velocity of up to 320 kmh⁻¹. In consequence the bones of the wings and the shoulder girdle of peregrine falcons most likely experience large mechanical forces. We investigated the bones of the arm skeleton and the shoulder girdle of peregrine falcons. For comparison, we also investigated the comparable bones in European kestrels (*Falco tinnunculus*), sparrow hawks (*Accipiter nisus*) and pigeons (*Columba livia domestica*). The normalized bone mass of the entire arm skeleton and the shoulder girdle (coracoid, scapula, furcula) was significantly higher in *F. peregrinus* than in the other three species investigated. The midshaft cross-section of the humerus of *F. peregrinus* had the highest specific bending stiffness per body mass and the highest second moment of area. The mineral densities of the humerus, radius, ulna, and sternum were highest in *F. peregrinus*, indicating again a larger overall stability of these bones. Furthermore, the bones of the arm and shoulder were strongest in peregrine falcons. Computational fluid dynamics simulations suggest, that the forces that pull on the wings of a peregrine can reach up to three times the falcon's body mass at a stoop velocity of 80 ms⁻¹.

Introduction

The peregrine falcon (*Falco peregrinus*) is the world's fastest bird. While attacking its bird prey in midair (Mebs and Schmidt 2005) a diving peregrine can reach velocities of up to 320 kmh⁻¹ (Tucker and Parrott 1970; Orton 1975; Tucker 1990; Savage 1992; Clark 1995; Peter and Kestenholz 1998; Franklin 1999, 2011). To learn more about the flight of peregrines, wing contours and flight trajectories of diving peregrines were investigated by Ponitz et al. (2014a, b). These investigations suggest that during a dive, but also while pulling out of a dive, the arm skeleton and the wings and tail feathers of *F. peregrinus* are exposed to large mechanical forces (Ponitz et al. 2014a). To uncover possible adaptations that allow peregrines to cope with these forces, Schmitz et al. (2015) studied the morphology and material properties of the wing and tail feathers of *F. peregrinus* (primary 10, alula one, central tail feather). For comparison Schmitz et al. (2015) also investigated the corresponding feathers in sparrow hawks (*Accipiter nisus*), European kestrels (*Falco tinnunculus*) and pigeons (*Columba livia domestica*). The latter three

1 species were chosen because they differ markedly in flight style but do not reach the high flight
2 velocities of a diving peregrine. According to Schmitz et al. (2015) the tail feathers of *F.*
3 *peregrinus* are more stable than the corresponding feathers of the other three bird species.
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6 During a dive, peregrines alter the shape of their wings; while accelerating, they move them
7 closer and closer to their body (Franklin 1999). At top velocities they build a wrap dive vacuum
8 pack, *i.e.* the wings are completely folded against the elongated body (Seitz 1999). Peregrines are
9 not only extremely fast flyers but also maintain a remarkable maneuverability at high flight
10 speeds. For instance, during courtship they often turn sharply from a fast vertical dive into a steep
11 climb (H.Bleckmann, pers. observation).
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18 Out of the four bird species investigated in the present study, *C. livia domestica* is the most
19 sustained flyer (Heinzel et al. 1992) In this species bones show no striking morphological
20 adaptations (Pennycuick 1968a; Dial 1992; Bachmann et al. 2007; Berg and Biewener 2010).
21 Since the forces acting on the wings of a peregrine falcon during a dive can hardly be measured
22 directly, we used finite element analysis to calculate these forces. Our calculations indicate that
23 up to 350 g may pull on the wings of a peregrine diving with a velocity of 80ms⁻¹ (288 kmh⁻¹).
24 Due to the expected high forces experienced by the wings of a diving peregrine (or while pulling-
25 out of a dive), the wing and shoulder bones should be significantly stronger in peregrines than in
26 other bird species (Norberg 1981, 1985; Selker and Carter 1989). We tested this hypothesis by
27 investigating the bones of the arm skeleton and the shoulder girdle of *F. peregrinus*. For
28 comparison we also studied the comparable bones in *F. tinnunculus*, *A. nisus* and *C. livia*
29 *domestica*.
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44 **Material and Methods**

45 **Animals**

46 Four bird species were investigated: *F. peregrinus* (Tunstall 1771), *F. tinnunculus* (Linneus
47 1758), *A. nisus* (Linneus 1758) and captive bred *C. livia domestica* (Gmelin 1789). In all species,
48 four males and four females were investigated. Mean body masses were 546 g ± 58 (male *F.*
49 *peregrinus*) and 849 g ± 34 (female *F. peregrinus*), respectively. The corresponding values for *F.*
50 *tinnunculus* and *A.nisus* were 167 g ± 12 and 218 g ± 11 (males) and 165 g ± 2 and 224 g ± 16
51 (females). Mean body mass of *C.livia domestica* was 479 g ± 16, values for males and females
52 were not significantly different.
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Animals were provided by Walter Bednarek (two *F. tinnunculus*; CitesNr. DE-COE121306201, two *A. nisus*; DE-COE121206202 and one *F. peregrinus*; DE-COE080818171); another *F. peregrinus* was supplied by Daniel Müller (DE-HF98050800001). Carcasses of six *F. peregrinus*, six *F. tinnunculus*, six *A. nisus* and eight *C. livia domestica* were provided by the Clinic for Birds, Reptiles, Amphibians and Fish at the Justus-Liebig-University of Giessen, Giessen, Germany in cooperation with the Society to Support Avian Medicine in Giessen (Verein zur Förderung der Vogelmedizin in Gießen e.V.). All animals included in our study were submitted by private persons immediately after being found or by veterinarians and rehabilitation centers in the first 48 hours upon admission. Birds which had been kept in captivity for more than two days were excluded from the study. All individuals were clinically examined including radiographic examination in latero-lateral and ventro-dorsal view. Dead animal were inspected during routine necropsy. Therefore, only injured animals with a presumptive acute cause of injury were included in the study without findings of an underlying subacute or chronic course of disease. Examination (preparation of the carcasses, detailed investigation of muscles and bones) of these birds was authorized by the Regierungspräsidium Giessen, Dezernat 53.2. No bird was killed for our study. Therefore, no permits for animal experiments were needed. All animals were stored at -18°C and thawed 16-24 hours before dissection. Animals were kept in a freezer for less than six months before investigation. All animals of a given species and sex had a comparable body mass (BM); their muscles did not show any signs of autolysis. From each animal BM and body length were obtained. Feathers and skin were removed unilaterally. Four individuals per species were fixed in 4% formaldehyde in phosphate buffer for two weeks. After fixation, animals were stored in a solution that contained 0.5% formaldehyde. Four individuals from each species were prepared immediately after thawing. Freezing of specimens prior to investigation may increase the strength of the bones slightly but does not change any other bone properties (Turner and Burr 1993). All bone mineral density (BMD) measurements (see below) were performed on individuals prior to fixation.

49 Bones

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Bone integrity was verified radiographically before further analysis using a high-frequency digital diagnostic x-ray unit (Gierth HF 400A, GIERTH X-Ray international GmbH, Riesa, Germany). Only one side of the body from each animal was used for analysing the arm skeleton and the thorax. All muscles were removed and the bones were cleaned manually. Finally, bones were air-dried and the length of the arm skeleton was measured from the shoulder joint (reference point middle of epiphysis) to the tip of the *Digitus major*. Finally, data were related to BM. The

1 masses of the arm and shoulder skeleton (masses per BM) were determined after rewetting the
2 bone pieces in a physiological saline solution (0.9% NaCl = 9 g NaCl/L) for 24 h, since wet
3 bones are more similar to the bones of a living animal than dry bones (Zysset et al. 1999). The
4 shoulder skeleton consisted of the coracoid, the scapula and the halved furcula. Measurements
5 were conducted on eight specimens of each species.
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10 The midshaft cross-sections of the humeri (Fig. 1) were chosen as focal point of our analyses
11 because the humerus has to cope with the highest bending stresses (Biewener and Taylor 1986a,
12 b). For our investigations the humeri of four specimens per species were immersed in a
13 physiological saline solution for 24 hours. From the midpoint of the humeri, 2-3 mm thick slices
14 were cut off with a coping saw. Bone pieces were air-dried, embedded in epoxy resin (Toolcraft,
15 epoxy resin L and hardener L), and polished with a diamond point. The trimmed bone areas were
16 used for the determination of the second moment of area (I) and the Young's modulus (E). I was
17 determined by point counting of the trimmed bone areas (Purslow and Vincent 1978; Schmitz et
18 al. 2015). Drying and rewetting had no effect on I as the bone pieces were embedded in Epon.
19 For the determination of I, the bones were bend in dorso-ventral (I_{dv}), lateral (I_{lat}) and diagonal
20 (I_{diag}) direction (Fig. 1e). Finally, I was normalized to the BM of the respective animal. This
21 allowed estimating the resistance of a bone to bending loads (Boresi and Schmidt 2002; Simons
22 et al. 2011; Brassey et al. 2013).
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36 For the determination of the Youngs modulus, samples were rewetted as the dry status of bones is
37 far from the physiological condition of the bones of a living animal (Zysset et al. 1999).
38 Rewetting was done 2 hours prior to measurements. With a nanoindenter (*Hysitron Triboscope*,
39 *D3100*), 72 indents were performed per trimmed bone area using a Berkovich diamond tip with a
40 load of 5000 μ N. This load caused a contact depth of about 400 nm. The specific bending
41 stiffness M is:
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$$49 \quad M = (E * I) * d^{-1} \quad (1)$$

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52 with E = Young's modulus (Nm^{-2}), I = second moment of area (m^4) and d = sample width (m).
53 Hardness and E were calculated with the Hysitron software from the unloading portions of the
54 load-displacement curves following a procedure given by Oliver and Pharr (1992). M was finally
55 normalized to the BM of the respective bird.
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The polar moment of area J was calculated according to:

$$J = I_{lat} + I_{dv} \quad (2)$$

with I_{lat} and I_{dv} being orthogonal to each other (Boresi and Schmidt 2002; Simons et al. 2011), J - normalized to BM - is a measure for the resistance of a bone to torsion (Brassey et al. 2013).

The cortical area (CA) of compact bone in a given cross section (Fig. 1), determined by point counting, was used to estimate the resistance of a bone to compressional loading (Boresi and Schmidt 2002; Simons et al. 2011; Brassey et al. 2013). The thickness of the cortex can be described by the dimensionless parameter K, which is the inner diameter divided by the outer diameter of a bone cross section (Selker and Carter 1989; Simons et al. 2011). Measurements were obtained from four specimens (2 males and 2 females) of each species. Finally CA was normalized to BM.

According to Currey (2002), the maximum stress that BM can exert on the bones of a given bird species is proportional to $BM^{1/3}$. Moreover, the force at which a bone will break (F) is proportional to $BM^{2/3}$.

Bone mineral density

Bone mineral density (BMD) measurements were performed on freshly thawed birds and after computer tomographic scanning. Bones and muscles were separated and bones were prepared as described above. Computed tomographic scans were obtained with a 3rd generation 16-slice helical scanner (PHILIPS Brilliance, Fig. 2) using a standard protocol: 0.7mm slice thickness, 140kV, 200mA. Three specimens per species were investigated. System calibration was done with a customized liquid dipotassium hydrogenphosphate phantom consisting of five cylindrical tubes with permanent reference densities (distilled water as well as hydrogen-phosphate diluted in distilled water: 1:100, 1:200, 1:400, 1:800), acting as bone mineral and water equivalent. The phantom was placed on the animal during the scan (Kalender et al. 1995; Cann 1988) and a calibration curve was prepared for each scan. BMD was measured within stacks of manually drawn regions of interest and within standardized regions of the long bones covering a predefined percentage of the entire bone length. For the long bones (humerus, radius, ulna, coracoid and clavícula) a mid-diaphyseal stack covering 10% of the entire bone length and a proximal and distal metaphyseal stack sparing the former physis covering 5% of the entire bone length,

1 respectively, were obtained. For the scapula, 10% of the entire bone length was obtained in the
2 region of the neck and at its widest cranio-caudal dimension. Mean, range and standard deviation
3 (SD) of the Hounsfield units (HU) within the regions of interest were used for analysis. BMD
4 was determined according to phantom based quantitative computer tomography using a
5 conventional multislice computer tomograph scanner and a customized K_2HPO_4 phantom.
6 Quantitative computer tomography is an established method to assess BMD and relative risk of
7 osteoporotic fracture in people and does not implement added filtration. To address the
8 systematic error potentially introduced by the polychromatic nature of the x-ray beam and beam
9 hardening effects, strict standardization of the study set up was observed (Rueggsegger et al. 1976;
10 Cann 1985; Cann et al. 1985). Moreover, the influence of beam hardening on BMD is assumed to
11 be small and negligible in diaphyseal areas of the bone, as these areas are composed of a thin
12 cortex and an "empty" pneumatized medullary cavity only. BMD measurements comprised
13 cortical bone, blood, fatty tissue, and pneumatized regions of the medullary cavity. BMD values
14 are expressed as equivalent densities in milligrams of K_2HPO_4 per millilitre of bone tissue (HA,
15 $mg\ cm^{-3}$) using a calibration phantom and a linear regression model.
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28 Computation Fluid Dynamics

29 We calculated the forces acting on the wings of a diving peregrine using computational fluid
30 dynamics (CFD), from which the pressure distribution and shear stresses along the body and
31 wing contours can be derived. The present study builds on previous detailed field studies of life
32 bird and wind-tunnel experiments of realistic models of peregrine falcons (Ponitz et al. 2014a, b),
33 which is the base for the 3D geometry of the wing and body shape used herein, i.e. the cupped
34 wing configuration (see Ponitz et al. 2014a). The present method follows a procedure described in
35 Ponitz et al. (2014a). The three-dimensional CAD model of the falcon was transferred into a
36 computational unstructured grid using a grid generation tool ICEM CFD 14.5 (ANSYS, Inc.,
37 Canonsburg, PA, USA). The computational domain includes the inflow region, the falcon region,
38 and the downstream wake region of the flow. Special attention was paid to the meshing of the
39 falcon. Refinements toward near-wall regions were taken into consideration. The grid consists in
40 total of 6.5 million unstructured tetrahedron cells and 1.5 million prism cells on the falcon
41 surface. A mesh independency check for the results of lift and drag coefficients was done for up
42 to 10 million cells. Simulation stability was investigated in respect to different grid parameters
43 and following settings leads to stable results: The height of the first prisms layer on the falcon
44 surface was set to 0.1 mm with a growth factor of 1.1 for the following layer perpendicular to the
45 wall and a total number of 10 layers. For these simulation parameters the numerical flow
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1 simulations delivered stable values which furthermore matched the experimental results of lift
 2 and drag forces obtained from the wind-tunnel tests (see Ponitz et al. 2014a, b). The numerical
 3 flow simulation was performed using the open source CFD software OpenFOAM (OpenCFD
 4 Ltd., Bracknell, UK). The code numerically solves the conservation equations of mass and
 5 momentum by means of a finite volume approach ([https://www.openfoam.com/documentation/
 6 tutorial-guide/](https://www.openfoam.com/documentation/tutorial-guide/)):
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$$10 \quad \frac{\partial}{\partial x_i}(u_i) = 0 \quad , \quad \frac{\partial}{\partial x_i}(u_j u_i) = -\frac{1}{\rho} \frac{\partial p}{\partial x_i} + \frac{1}{\rho} \frac{\partial \tau_{ij}}{\partial x_j} \quad (3)$$

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 19 with p = pressure, u = velocity vector, x = Cartesian coordinate system, and ij = stress tensor in
 20 Einstein notation. The calculation of the resulting forces F on the body segments were done by
 21 integrating the pressure over the outer surface exposed to wind.
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$$26 \quad \vec{F} = -\iint (p \cdot \vec{n}) dA \quad (4)$$

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 31 This excluded the part of the surface where the segment is separated from the body.
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34 In our simulations air was treated as a single-phase, incompressible (0.07 Ma), isothermal (20°C)
 35 Newtonian fluid with constant density (1.189 kgm⁻³) and viscosity (18.232 ×10⁻⁶ Pas). Three
 36 different stoop velocities were simulated, beginning with 22.5 ms⁻¹, as observed in our
 37 experiments on a dam wall (Ponitz et al. 2014a). We then increased the stoop speed in the
 38 simulations to 40 ms⁻¹ and to a maximum of 80 ms⁻¹, a speed that diving peregrines most likely
 39 can reach (Franklin 1999, 2011; Orton 1975; Tucker and Parrott 1970). The Reynolds number
 40 Re , based on the body length of a peregrine (400 mm), is for all three cases > 5×10⁵. Therefore,
 41 turbulent flow was taken into account by a Reynolds averaged approach (Spalart-Almaras
 42 turbulence model). The no slip boundary condition was applied to the body surface. For
 43 calculating the forces acting on the wings, we segmented the body in 4 parts: the two wings, the
 44 tail and the center body. Integration of the pressure and shear-stress along the surface segments
 45 then provide the caudal, dorsal, and distal forces on the wing. In aerodynamic notation these
 46 forces on the wing correspond to the drag, the lift force and the side-force of an aerofoil.
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1 The wing configuration chosen was the “cupped wing configuration”, one which a peregrine
2 adapts late in a stoop as it starts to pull-out (Ponitz et al. 2014a, b). The formulation “cupped
3 wings” was introduced by Tucker and Parrott (1970) to describe the shape of the downward tilted
4 tips of the wing which is a typical falcon shape during a dive. In the cupped wing formation, air
5 can enter the space between the wing and the body which affects the lateral forces on the wings.
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7 The transition to pull-out is when we expect the largest forces acting on the body and the wing as
8 this is the moment of change from a straight flight path into a curved one. This requires an
9 increase in lift forces to overcome the centrifugal forces at such high speeds.
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16 Statistics

17 Mean values and standard deviations (SD) were calculated. Differences between species were
18 compared using SPSS (IBM, version 22). Distribution of the data was tested with a one-sample
19 Kolmogorov-Smirnov test. One-way analysis of variance (ANOVA) with Bonferroni test as post-
20 hoc evaluation was used to compare mean values of different species if values were normally
21 distributed. If data did not follow a Student’s t-distribution, a Mann-Whitney-U test was applied.
22 If not otherwise stated significance level was $P < 0.05$. In most figures significant differences are
23 only shown for *F. peregrinus*.
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31 Results

32 Bones

33 Normalized to body mass (BM), *F. peregrinus* and *C. livia domestica* had the shortest arm
34 skeleton (Fig. 3a). In *F. peregrinus* the mass of the arm and shoulder skeleton, also normalized to
35 BM, was higher than in the other three species (Fig. 3 b, c). *C. livia domestica* had the largest
36 pectoral muscles (not shown), but a relatively light weight arm and shoulder skeleton. The
37 Young’s modulus E of the humeri (Table 1) was highest in *F. peregrinus*, followed by *F.*
38 *tinnunculus*, *A. nisus* and *C. livia domestica*. The maximum and minimum of I was greatest in *F.*
39 *peregrinus* (Table 1). If normalized to BM, I was greatest in *F. peregrinus* for all bending
40 directions applied (Fig. 4a). The calculated M value, if normalized to body mass, was also higher
41 in *F. peregrinus* than in the other three bird species (Fig. 4b). In the normalized data sex
42 differences were not found.
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58 The maximum stress that the body mass can exert on the bones of a bird is proportional to body
59 mass^{1/3}. The force F is proportional to body mass^{2/3} (Currey 2002). Using these relations, the
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1 maximum stress to which the bones will be exposed is 1.2 times larger in *F. peregrinus* than in *C.*
2 *livia domestica* (males 1.04, females 1.2) and 1.6 times larger in *F. peregrinus* than in *F.*
3 *tinnunculus* and *A. nisus* (1.48 and 1.57). Moreover, the force at which the humeri of *F.*
4 *peregrinus* will break is 1.3 times larger in *F. peregrinus* than in *C. livia domestica* (males 1.08,
5 females 1.44) and 2.3 times larger in *F. peregrinus* than in *A. nisus* and *F. tinnunculus*. (2.18 and
6 2.45).
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12 The calculated J value and CA were greatest in *F. peregrinus* (Figs. 5, 6a). If normalized to body
13 mass, differences between raptor species vanished, but the difference to *C. livia domestica*
14 remained (Fig. 6e). All species possessed thin-walled humeri (as indicated by the K-values in
15 Fig. 7). Humeri of *F. peregrinus* had the lowest K-value, i.e. the humeri of peregrines had the
16 thickest walls. However, only the difference to *C. livia domestica* was significant.
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23 BMD, expressed as the amount of hydroxyapatite (Cann 1988; Damilakis et al. 2007), was
24 highest in *F. peregrinus* (humerus, radius, ulna, and sternum) (Fig. 8b). The scapula and furcula
25 of *C. livia domestica* revealed the highest values, followed by the values of *F. peregrinus*. The
26 coracoid had a similar HA-value in all species (Fig. 8b). The calculated HA-values showed the
27 same relationships as the values on the Hounsfield scale (HU). HU was used to calculate the
28 statistics, as shown in Fig. 8a.
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36 Forces acting on the wings

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38 Our morphological data show that the humerus, ulna, radius and sternum of peregrine falcons are
39 extraordinarily strong (Figs. 4, 5, 6, and 8). This supports our working hypothesis that the wings
40 of a diving peregrine are exposed to extraordinary large forces. To estimate these forces we
41 performed CFD simulations. Figure 9 shows the side and front view of the “cupped wings”
42 configuration of a diving peregrine. This configuration was used for our calculations (Fig. 10).
43 The initial assumed stoop speed was 22.5ms^{-1} , the angle of incidence was 5° . Aerodynamic
44 conditions were derived from high-resolution tracking and imaging of life birds (Ponitz et al.
45 2014a). In Fig. 10, the segment of the wing that was used for the integration of the forces is
46 coloured in blue. Forces are given in a body-related coordinate system. Figure 10 shows that the
47 distal force acting on the wings pull the cupped part of the wings away from the body. This force
48 is of the same order of magnitude as the dorsal force, which corresponds to the lift force acting
49 on the wing.
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1 Calculations for three diving speeds show that the forces acting on the wings of a diving
2 peregrine scale proportional to the square of diving velocity (Fig. 11). For an adult peregrine
3 falcon (assumed mass 500 g) and a diving velocity of 80 m/s (288 km/h) the flight muscles of the
4 falcon must develop a force of -11.5 Newton in dorsal direction and -9 Newton in distal direction
5 (the negative sign hints that the muscle forces are pointing towards the body axis to counter-act
6 the aerodynamic forces). Thus, compared to the weight of a peregrine falcon the forces that the
7 wings may experience are in total about 3-times higher.
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13 **Discussion**

14 Our results are consistent with the data reported for other bird species. The second moment of
15 area I in *Larus californicus* (California gull) of the humerus varies between 1 and 40 mm⁴ (BM
16 of the animals was 40 to 700 g) (Carrier and Leon 1990). This agrees well with our values for *C.*
17 *livia domestica* (body mass about 450 g, I=23-38 mm⁴). To our knowledge the Young' modulus
18 of the humerus has been measured by nanoindentation for two species of penguins; the values
19 obtained are 19.5-22.1 GPa (Currey 1988). In volant birds, the thickness of the cortices (K-
20 values) of the humeri varies between 0.68 and 0.86: the K-value for the humerus of *C. livia*
21 *domestica* is 0.83 (Currey and Alexander 1985). These values are similar to our value for *F.*
22 *peregrinus* (0.85).
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34 Most previous studies that assessed the mechanical properties of bones used a Berkovich tip and
35 the Oliver and Pharr calculation (Oliver and Pharr 1992). There is some evidence that E of a bone
36 can be overestimated by using this method (Rodriguez-Florez et al. 2013). Bones are anisotropic
37 and viscoelastic. To obtain realistic values with a Berkovich tip (Oliver and Pharr 1992), the
38 material under investigation must be isotropic and elasto-plastic. To get realistic values we
39 corrected for viscoelastic effects as described by Tang and Ngang (2004) and Tang et al. (2007).
40 Based on these studies, Ngan (University of Hong Kong, Department of Mechanical
41 Engineering) calculated, that our data must be multiplied with the correction factor 0.78.
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51 **Bones**

52 A long bone rarely fractures in a living animal due to pure axial loads (Carter and Spengler
53 1982). Most fractures are caused by stresses created by bending and torsion (Rubin and Lanyon
54 1982; Biewener et al. 1983; Biewener and Taylor 1986a, b), forces that especially affect the
55 humerus and the forearm of a bird (Bou et al. 1991; Biewener and Dial 1995). The resistance
56 against bending and torsion is higher in thin walled bones with a large diameter (de Margerie et
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1 al. 2005). For bending and torsional loading, the force that is required to fracture a bone is
2 proportional to $BM^{2/3}$ (Selker and Carter 1989; Currey 2002). In the bird species investigated in
3 this study, forces that may lead to bone fracture were more than twice as high in *F. peregrinus*
4 than in *A. nisus* and *F. tinnunculus* and 1.3 times higher than in *C. livia domestica*. The large
5 body mass of the peregrine falcon is one factor that requires solid bones, i.e. bones that resist
6 fractures. A second factor are probably the forces to which the wing bones of a peregrine are
7 exposed during a high-speed dive (Figs. 10, 11). The humerus, radius and ulna of peregrines
8 probably resist these loads more effectively than the corresponding bones of the other bird
9 species investigated.

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18 Bone geometry is the primary variable that determines bone strength (Selker and Carter 1989).
19 The midshaft cross-section of long bones can resist the highest bending stresses (Biewener and
20 Taylor 1986; Beer et al. 2006). A circular structure with a large diameter and a thin cortex (e.g.
21 the humeri of alate birds) has a higher I value and a greater resistance against bending and
22 especially against torsion than a structure with a small diameter and a thick cortex (Biewener
23 1982; Alexander 1983; Currey and Alexander 1985; Swartz et al. 1992; Swartz 1997; de
24 Margerie et al. 2005; Habib and Ruff 2008; Dumont 2010). All humeri of the bird species
25 investigated in the present study had thin cortices and a high K-value. This indicates a potential
26 bending in multiple planes and/or high amounts of torsion in the humerus (Habib and Ruff 2008).
27 The relatively low K-value of *F. peregrinus* coincides with the higher mass and a higher M value
28 of its arm skeleton. Even though the difference to the other species is small it may indicate an
29 adaptation to fast flight manoeuvres.

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41 Circular humeri are equally resistant against bending in all directions, whereas elliptical humeri
42 have the highest resistance to bending in the direction of the largest cross section (Simons et al.
43 2011). The cross sections of the humeri of the bird species investigated in this study were round
44 or elliptical. Because of the high second moments of area I and polar moments of area J in *F.*
45 *peregrinus*, the bending loads and the resistance to torsion were higher in *F. peregrinus* than in
46 the other three species (Figs. 4, 5). If normalized to body mass (Fig. 4a), the humeri of *F.*
47 *peregrinus* still have the highest I value in the bending directions tested. The round shaped cross
48 section of the humerus together with the high I and J values might ascertain that this bone can
49 resist high forces from all directions.

1 The resistance to compressional loads (CA) was higher in *F. peregrinus* than in the other species
2 investigated. However, if normalized to body mass, values no longer were significantly different
3 from the value of *C. livia domestica* (Fig. 6b). CA most likely is not an important parameter for
4 coping with the high forces during a dive. BMD derived from quantitative CT as measured by
5 dipotassium hydrogenphosphate equivalents per volume unit reflect bone strength and correlates
6 with the relative risk of fracture (Marshall et al. 1996). Besides the higher M-value of the humeri,
7 BMD-values of the entire wing bones also indicate that the wing bones of *F. peregrinus* are more
8 stable than the wing bones of the other bird species investigated. In *F. peregrinus*, the scapula
9 and furcula are less mineralised than in *C. livia domestica*. The high mineralization of the scapula
10 and furcula of *C. livia domestica* most likely is an adaptation to the forces produced by the great
11 breast muscles (especially the *M. pectoralis*) of pigeons.
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21 Forces acting on the wings

22 Our CFD simulations show that the aerodynamic lift force component acting on the wings (in
23 dorsal direction) is the highest, followed by the side-force component (distal) that tries to pull the
24 cupped part of the wings away from the body. Both are of the same order of magnitude, while the
25 drag force component (caudal) is considerably lower. The flight muscles of a diving peregrine
26 have to counter-act these aerodynamic forces to keep the wings close to the body. The
27 calculations for three diving speeds show that the forces acting on the wings scale proportional to
28 the square of diving velocity. The flight muscles of a peregrine falcon of 500 g weight and an
29 assumed diving velocity of 80 ms^{-1} (288 kmh^{-1}) must develop a force of -11.5 Newton in dorsal
30 direction and -9 Newton in distal direction (the negative sign indicates that the muscle forces are
31 pointing towards the body axis to counter-act the aerodynamic forces). Thus, compared to the
32 weight of the falcon the forces that the wings experience at the end of a dive are about 3-times
33 higher. Note that these aerodynamic forces may even be larger if the bird is going into a higher
34 angle of incidence during pull-out. This probably explains while the bones of the arm and
35 shoulder are so strong in peregrine falcons.
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50 The present study shows, that the forces acting on the wings of a diving bird largely depend on
51 flight velocity (Fig. 11). Kestrels dive after a windhovering bout to stoop on their ground
52 dwelling prey, however, the estimated height from which they start a dive usually is below 60
53 meters (personal observation). From this height even a diving peregrine falcon does not reach
54 speeds larger than 60 kmh^{-1} (Ponitz et al. 2014a). Thus the velocity of a diving kestrel is far less
55 than the velocity of a diving peregrine falcon. Consequently, the maximal forces acting on the
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1 wings of a diving kestrel are much smaller than the forces acting on the wings of a diving
2 peregrine. During courtship or prior to prey capture many eagles also dive. To our knowledge the
3 maximum diving speed of eagles has, however, never been measured. So far, no bird known can
4 match the flying speed of a peregrine falcon in its hunting dive. Further investigations should
5 analyse the bone architecture of peregrine falcons in more detail as the ultrastructure may also be
6 crucial in determining the biomechanical properties of bones adapted to potentially extreme
7 loads.
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Figure legends

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3 **Fig. 1** Cross-sections through the humeri of *F. peregrinus* (**a**), *Columba livia domestica* (**b**), *F.*
4 *tinnunculus* (**c**), and *A. nisus* (**d**). All humeri are oriented with dorsal side up (dorsal refers to the
5 dorsal side of the wing when fully extended). (**e**) directions of measurement. d dorsal, diag
6 diagonal, lat lateral, v ventral.
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12 **Fig. 2** BMD measurements. **a** General set up with the phantom (P) placed directly on the chest of
13 a bird. The gap between the phantom and the animal was reduced to a minimal extent (range 1 to
14 10 mm depending on species). The picture in **a** shows the transverse image of an entire bird
15 through the breast and wings. Right: thorax of the bird in dorso-ventral and left-lateral view,
16 respectively. Muscles and inner organs are coloured in grey. Example for the measurement of the
17 diaphysis of the ulna of *F. tinnunculus* (**b**) and *F. peregrinus* (**c**). The histograms in **d** and **e** show
18 the distribution of the Hounsfield units measured within the region of interest. Size relationships
19 are also indicated. H humerus, Mp M. pectoralis, R radius, U ulna, VC vertebral column, W
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37 **Fig. 3** Length and mass of the arm skeleton (**a, b**) and mass of the shoulder skeleton (scapula,
38 coracoid and half of furcula) (**c**) normalized to body mass. Significant differences ($P < 0.001$) are
39 marked by ** (ANOVA, $N = 8$ per species).
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44 **Fig. 4** Second moment of area (I) of cross-sections of the humeri (**a**) and specific bending
45 stiffness M calculated for the cross sections (**b**). All data are normalized to body mass. Data are
46 given for (from left to right) the diagonal, lateral and dorso-ventral bending direction (c.f. Fig.
47 1e). In all cases four animals per species were investigated. Significant differences between *F.*
48 *peregrinus* and the other species are marked by ** (ANOVA, $P < 0.001$).
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55 **Fig. 5** Polar moment of area J of the humerus. Polar moment of area (**a**) and polar moment of
56 area normalized to body mass (**b**) for four individuals (2 males and 2 females) of each species
57 (measured in the centre of the humeri). Significant differences are marked by ** (ANOVA,
58 $P < 0.001$).
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1 **Fig. 6** Area of the middle humeral cortex (a) and the middle humeral cortex normalized to body
2 mass (b). Four individuals (2 males and 2 females) per species were investigated. Significant
3 differences are marked by ** (ANOVA, $P < 0.001$).
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8 **Fig. 7** K-values of the humeri of the four bird species investigated. Significant differences are
9 marked by ** (ANOVA, $P < 0.001$).
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13 **Fig. 8 a** Hounsfield units HU (mg cm^{-3}) - measured directly on the bones - on the Hounsfield
14 scale with SD (vertical bars). Values are given for the wing and shoulder skeleton. Significant
15 differences are indicated ($P < 0.01$, ANOVA, post-hoc Bonferroni). **b** HA values for the bones of
16 the wing and shoulder skeleton. HA-values were calculated by using the Hounsfield unit data and
17 calibration data obtained with the phantom.
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25 **Fig. 9.** Computer aided design model of the cupped wing shape geometry of a peregrine falcon.
26 Contours were generated from multi-view high-resolution camera recordings of a life bird (adult
27 peregrine falcon, mass 500g) in diving motion along a dam wall (c.f. Ponitz et al. 2014a). Note
28 the gap between the inner and outer side of the wing, which allows for aerodynamic side-forces
29 to build up in addition to the lift forces.
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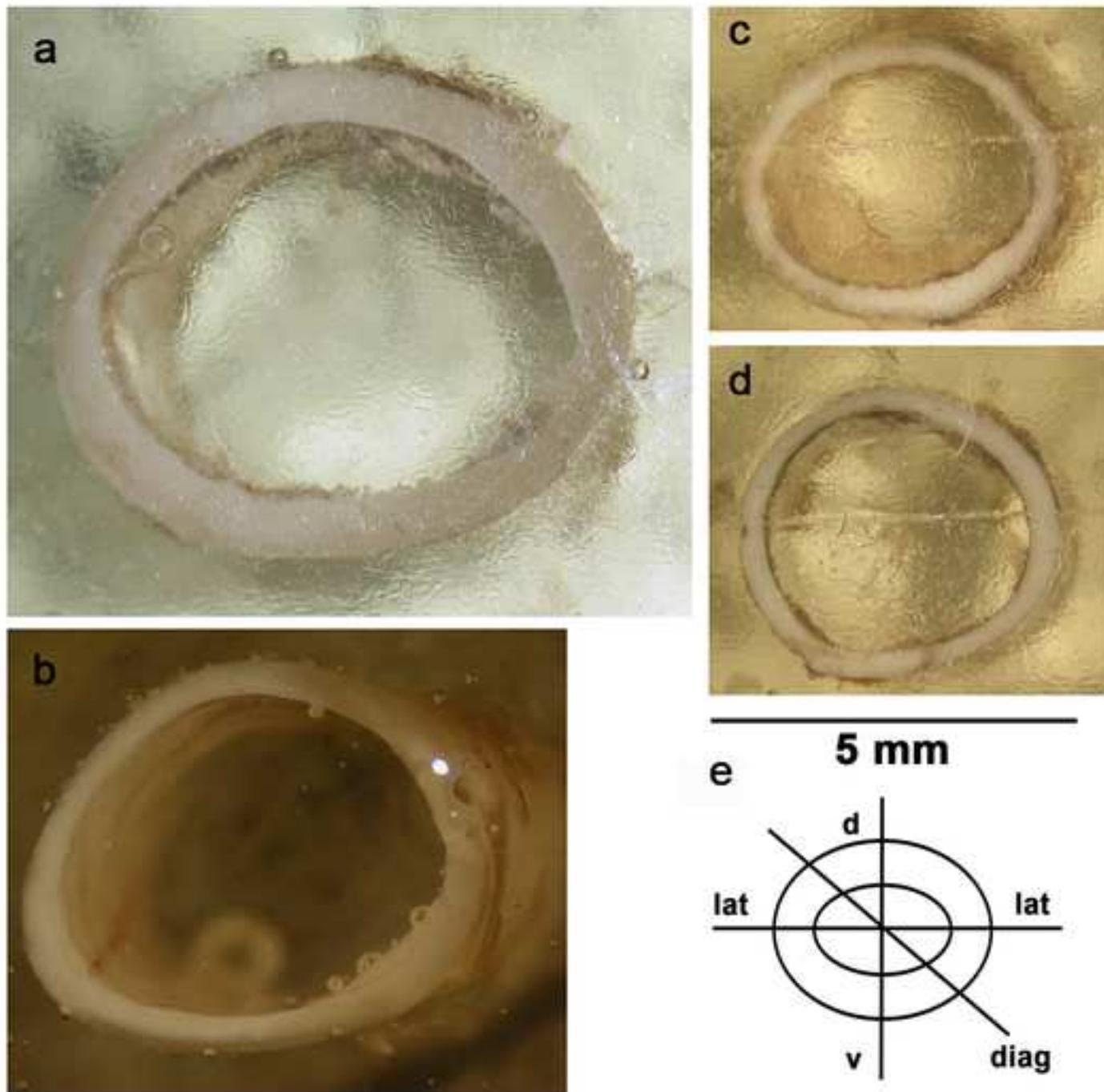
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36 **Fig. 10.** Isolated forces on the wings for the cupped wing geometry obtained from CFD
37 simulations. Assumed flight speed was $22,5 \text{ ms}^{-1}$, angle of incidence was 5° . The calculated
38 aerodynamic forces are given in a body-related-coordinate-systems (x-direction for caudal, y-
39 direction for dorsal, and z-direction for distal).
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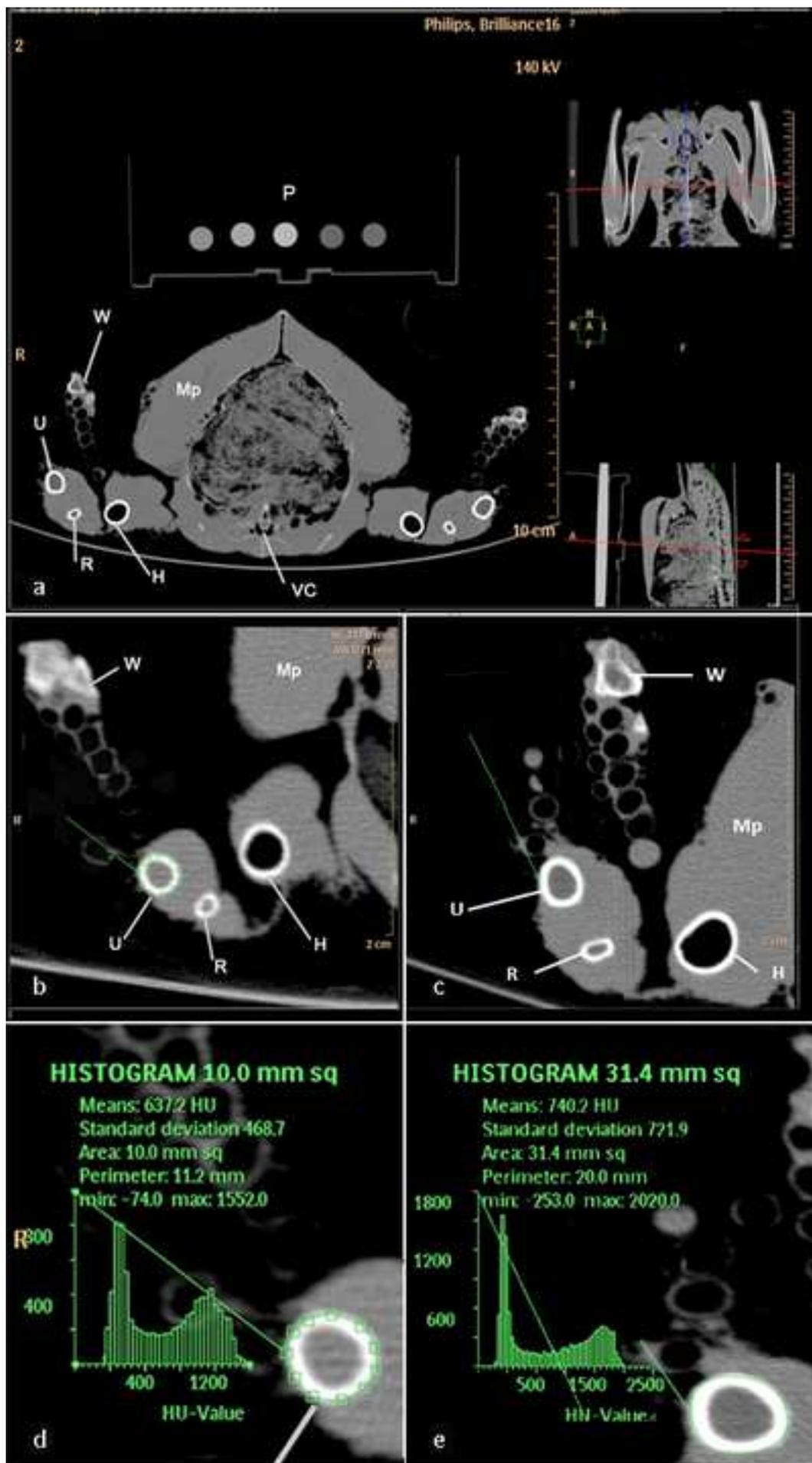
45 **Fig. 11.** Scaling of forces with the diving velocity U . For CFD-simulations assumed speeds were
46 $22,5 \text{ ms}^{-1}$, 40 ms^{-1} , and 80 ms^{-1} . The resulting forces are given in a body-related coordinate
47 system (x caudal, y dorsal, and z distal) (c.f. Fig. 10).
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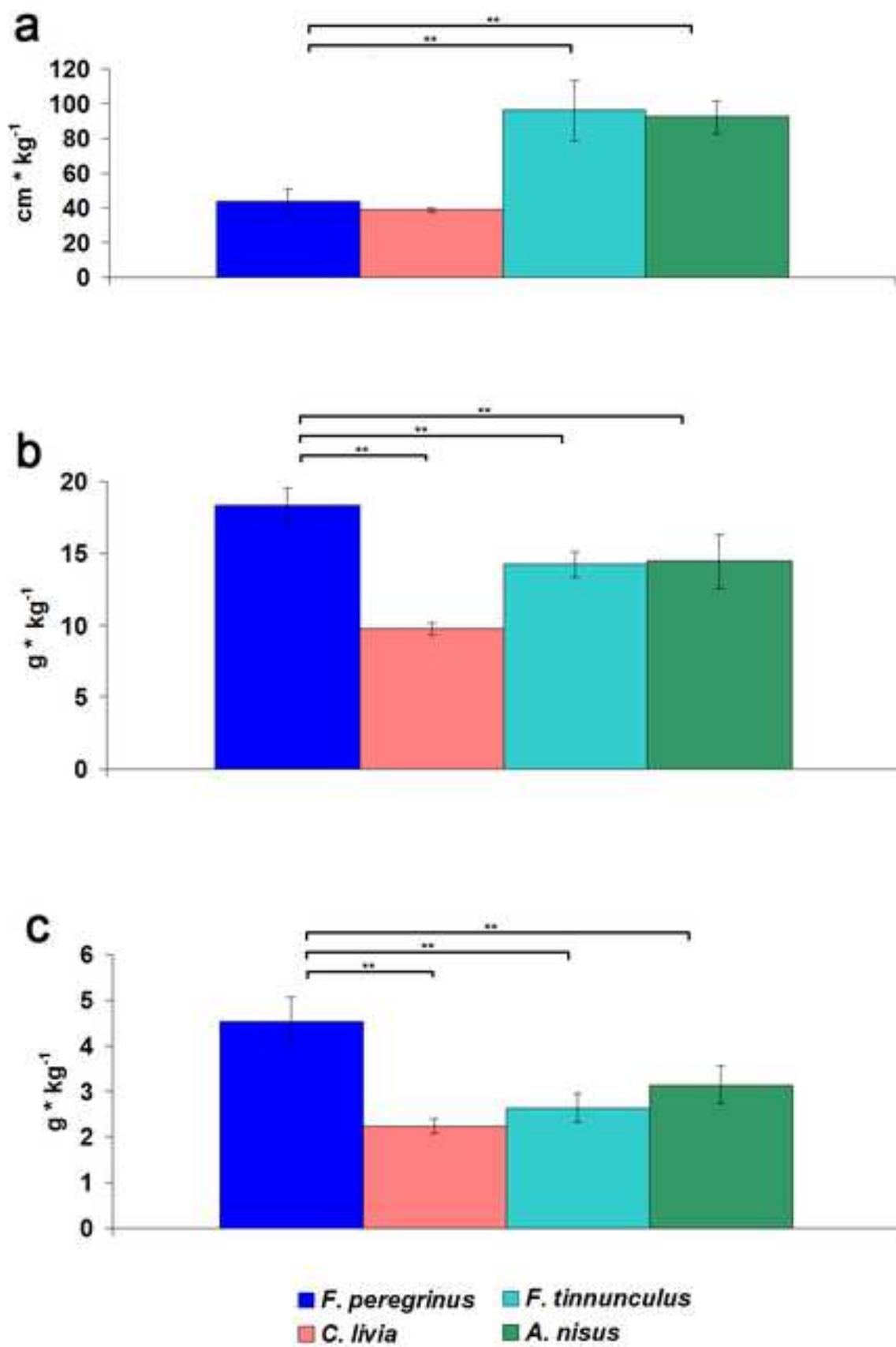
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52 **Table 1:** Maxima and minima of the second moment of area (I) and Young's modulus E of the
53 humeri of the bird species investigated. Results are given as arithmetic mean (\pm SD). Significant
54 differences between *F. peregrinus* and the other bird species are marked by * (ANOVA, post-hoc
55 Bonferroni, $P < 0.01$; Mann-Whitney U-test for Young's modulus).
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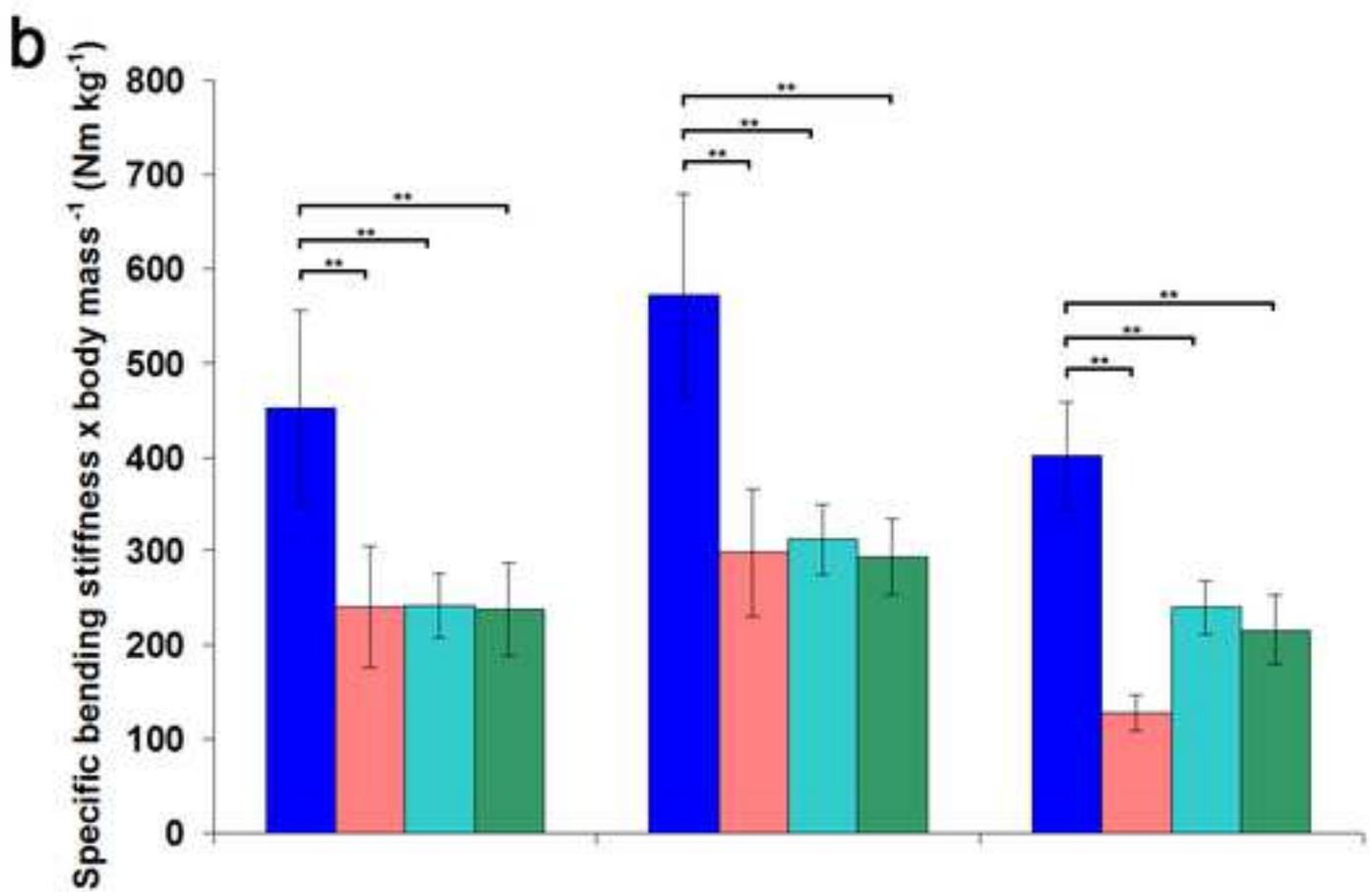
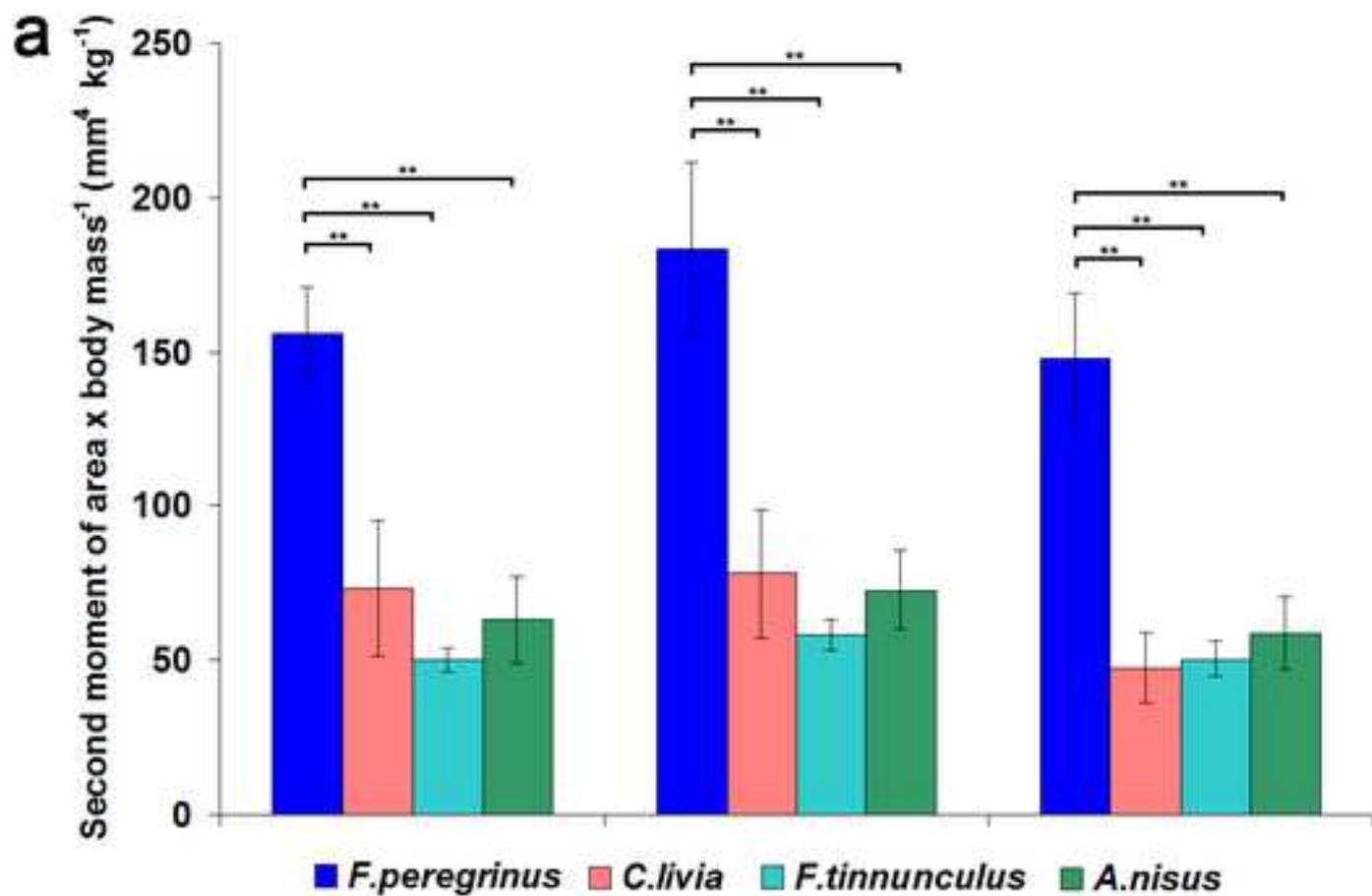
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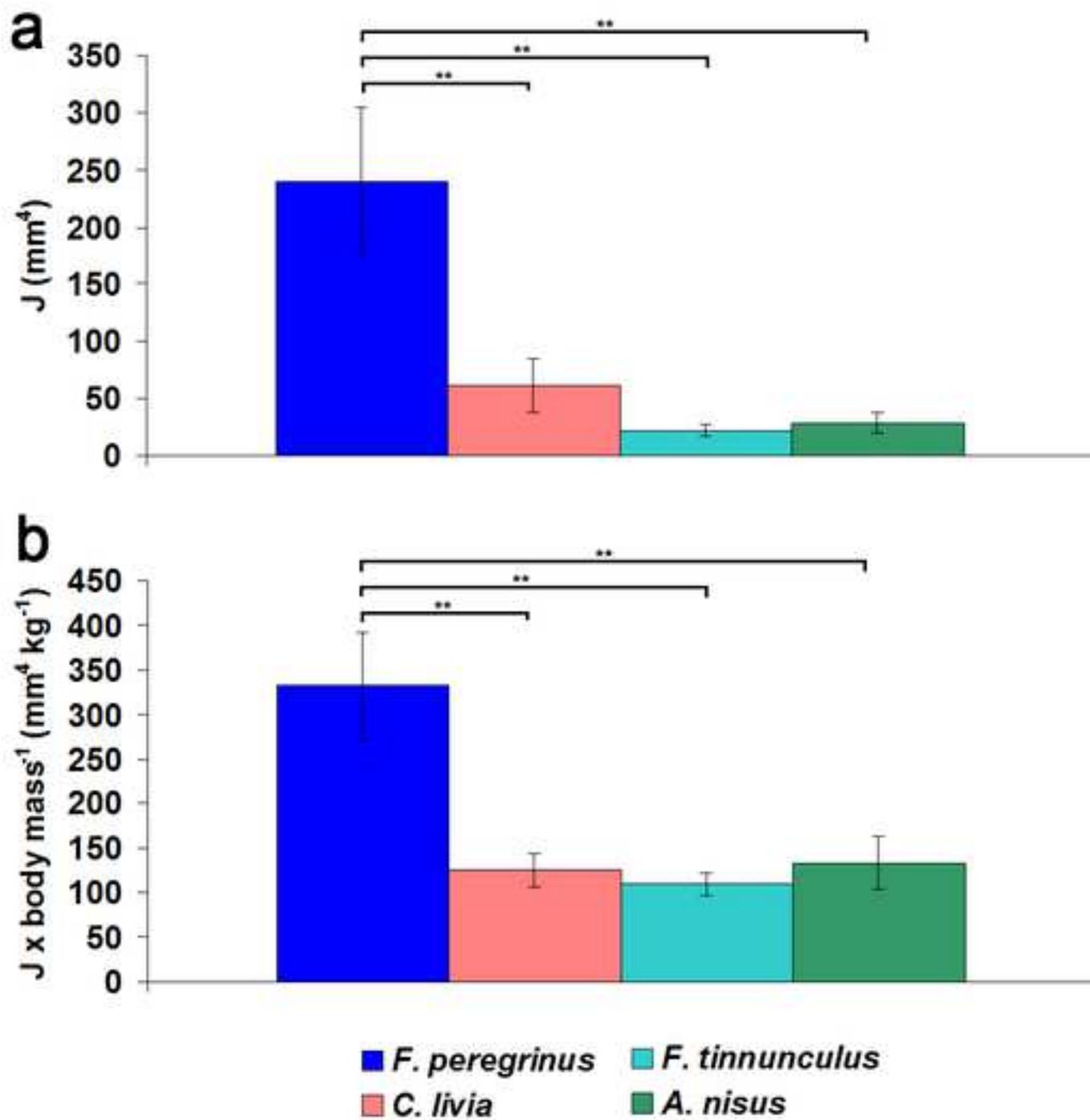
Species	I_{\max} (\pm SD) mm ⁴	I_{\min} (\pm SD) mm ⁴	Young's modulus (GPa)
<i>F. peregrinus</i>	132.98 (\pm 35.5)	107.29 (\pm 29.2)	22.7 (\pm 3.9)
<i>C. livia domestica</i>	37.95 (\pm 14.8) *	23.26 (\pm 8.1) *	18.7 (\pm 2.4)
<i>F. tinnunculus</i>	11.76 (\pm 2.27) *	10.21 (\pm 2.15) *	22.6 (\pm 4.3)
<i>A. nisus</i>	15.81 (\pm 5.0) *	12.76 (\pm 4.4) *	19.6 (\pm 3.6)

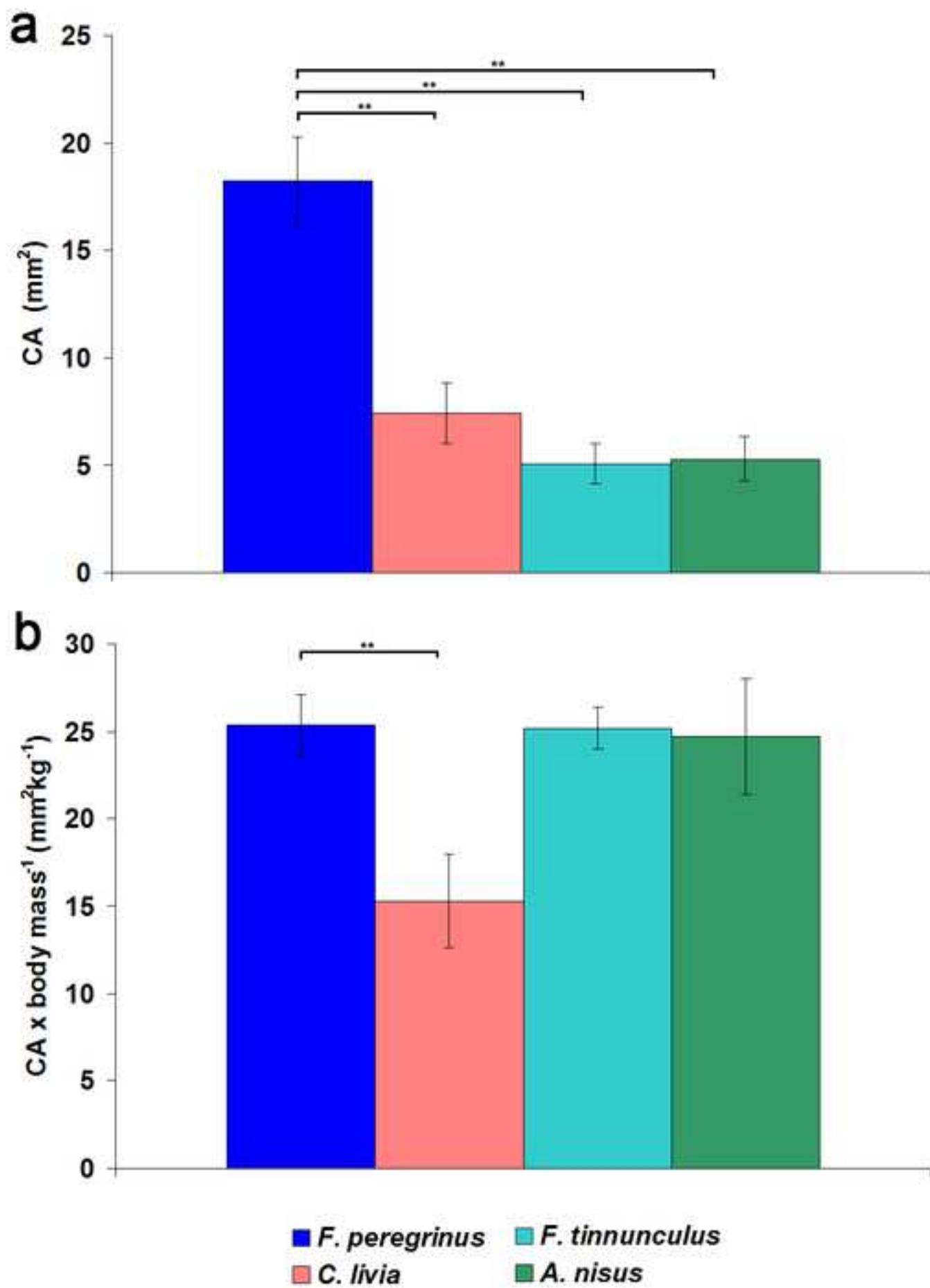


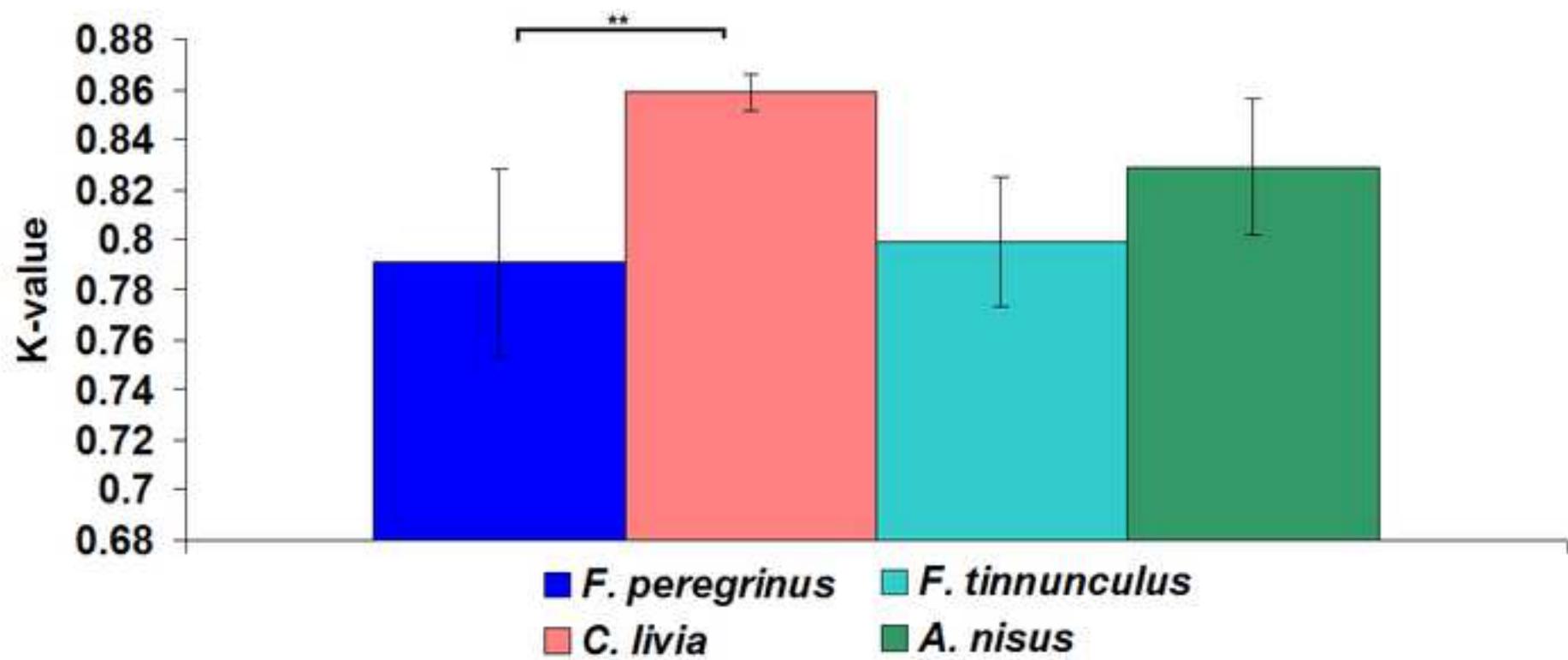


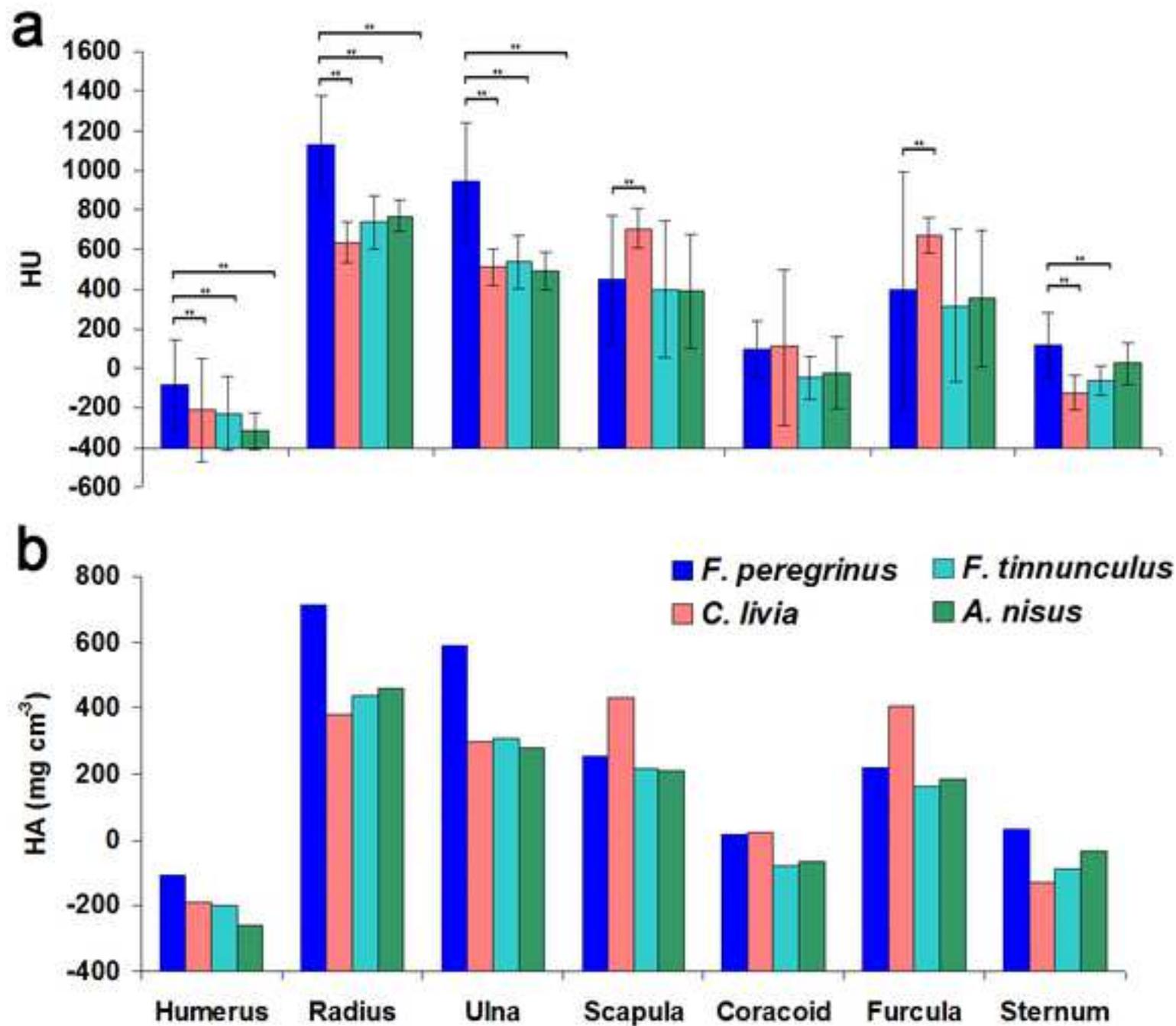


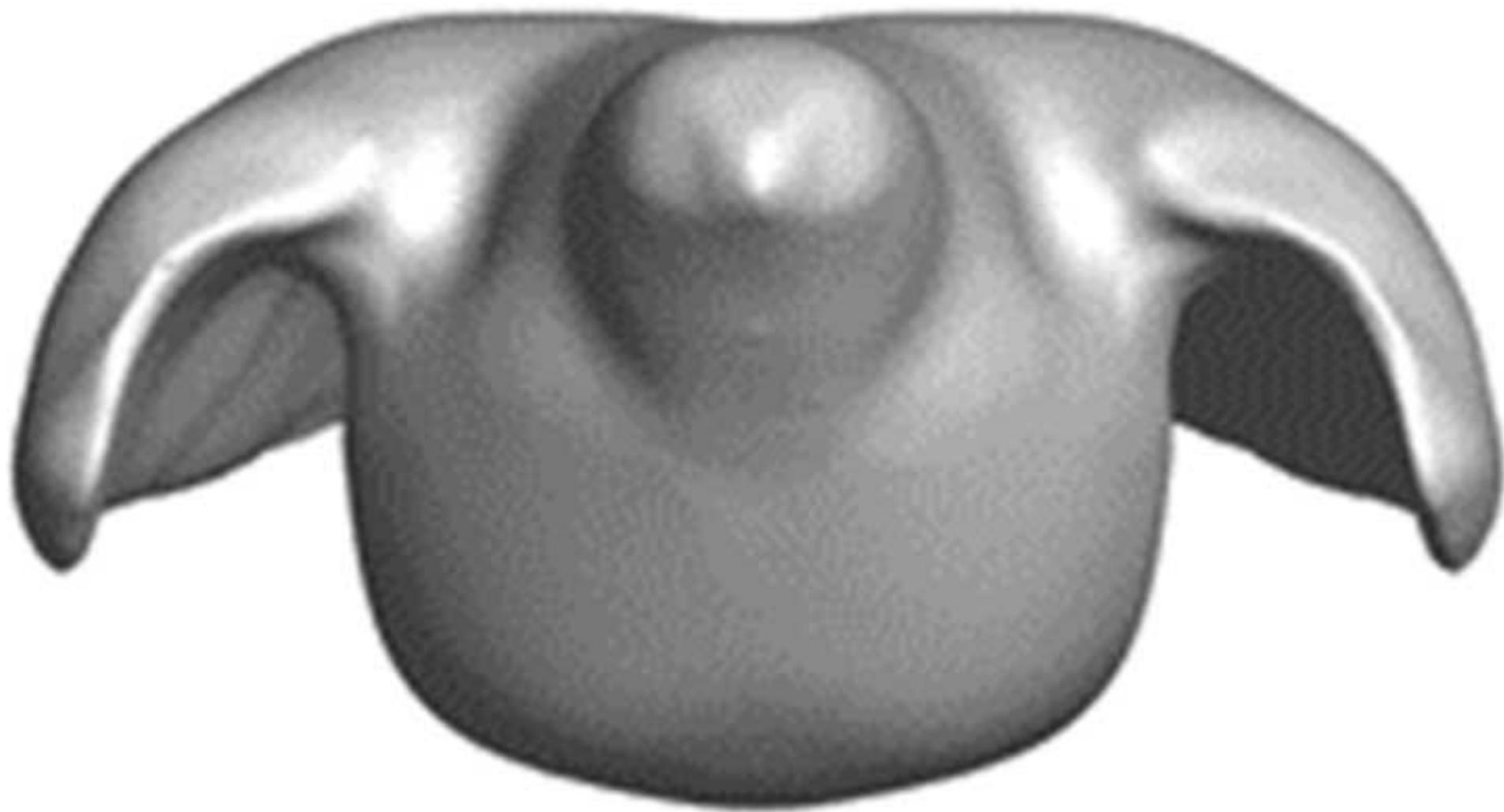




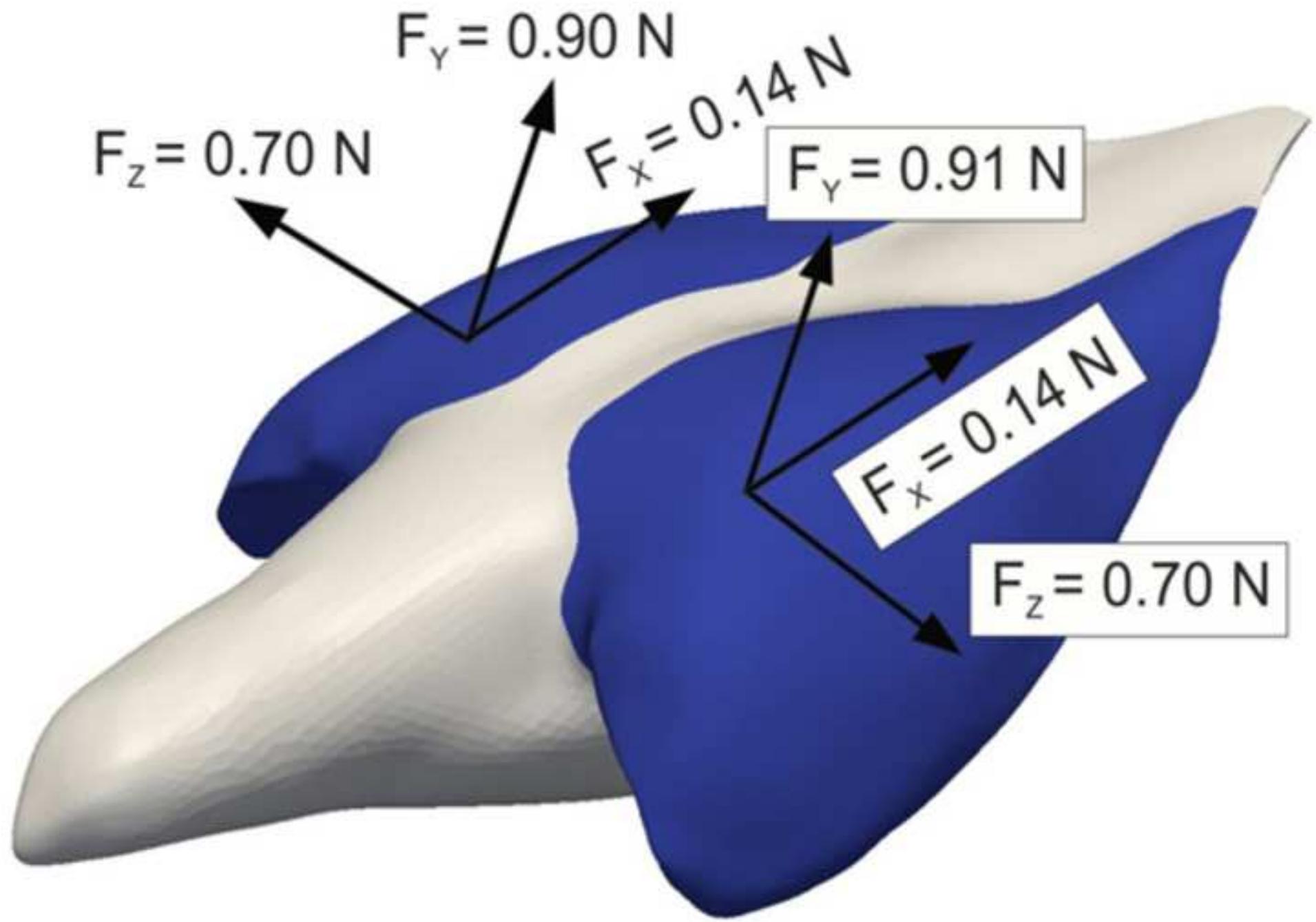












wing forces as a function of speed

