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Citation: Khan, K., Chen, Z., Liu, J. & Tsavdaridis, K. D. (2023). Experimental and analytical investigations on compression behaviors of rotary-connected sway column-supported steel modular interior frames. *Journal of Building Engineering*, 78, 107692. doi: 10.1016/j.jobe.2023.107692

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1 **Experimental and analytical investigations on compression** 2 **behaviors of rotary-connected sway column-supported steel** 3 **modular interior frames**

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10 **Abstract:** This study examines the compressive behaviors of sway column-supported
11 steel modular interior frames (SCSMIFs) using rotary-type vertical and horizontal inter-
12 modular connections (IMCs). The compression behavior of SCSMIFs was investigated
13 through experimental, numerical parametric, and analytical techniques. Findings
14 indicate that the relative rigidity of beam-to-column connections primarily influences
15 lateral translation. Adjacent upper columns displayed symmetrically inward or outward
16 elastic and plastic S-shaped local buckling without IMC failures. A finite element
17 model (FEM) was developed and validated, achieving a 1% average prediction error
18 for compressive resistance. The examination of 87 SCSMIFs with a validated FEM
19 revealed the effects of different parameters on compressive resistance, initial stiffness,
20 and pre-and post-ultimate ductility. Based on member stiffnesses and rotary-type IMCs
21 in semi-rigid and pinned conditions, theoretical models predicted sub-assembled
22 CMSIF buckling loads. The average theory-to-FEM results for pinned and semi-rigid
23 IMCs were 0.70 and 0.95, indicating that incorporating the stiffness of rotary-type
24 IMCs resulted in more accurate and less scattered buckling load predictions.
25 Considering their unique characteristics, the study's findings contribute to ensuring the
26 structural integrity and design of SCSMIFs with IMCs under compressive loads.

27 **Keywords:** Compression behaviors; Inter-modular connections; Column-supported
28 frames; Sub-assembled testing; Finite element analysis; Buckling load models

29

30 **1 Introduction**

31 Modular steel buildings (MSBs) employ prefabricated volumetric modules as a
32 practical, superior, secure, and sustainable alternative to conventional steel buildings.
33 These modules are assembled with intra-modular connections (ITCs) and vertical and
34 horizontal inter-modular connections (IMCs) [1,2]. Successful MSB applications, such
35 as the 44-story Croydon, 32-story B2, and 29-story SOHO and Apex, highlight the
36 practicality of these structures [3–5]. Compared to braced, hybrid, or wall-supported
37 systems, column-supported steel modules offer simple connectivity, a distinct load
38 transfer path, and high prefabrication levels [6,7]. The applications and benefits of these
39 modules and MSBs were apparent when the first government-approved MSB in China
40 was produced in just one month using 314 modules [2,8–10].

41 The structural performance of MSBs is mainly reliant on ITCs and IMCs, particularly
42 vertical installation and horizontal connectivity. While welded ITCs [11–15] are
43 preferred for beams and columns over bolted [16,17] or fin-plate [7] due to their
44 increased resistance [2], IMCs transfer forces vertically and horizontally between
45 modules and considerably influence structural stability, robustness, and behavior [18].
46 Thus, they have been the subject of extensive research, as reviewed in [4,5,19–23].
47 Recent literature has proposed varying reliable bolted [24–27], welded [11–15], shear-
48 keyed [28–31], automatic [32–35], and pre-and post-tensioned [36–39] IMCs. A wealth
49 of research has investigated IMCs' mechanical performances under various loadings.
50 Shear loading research is summarized in [40–42], tensile in [33,40,43], bending in
51 [9,44,45], and seismic in [46–48]. Despite this, calculating buckling length and load in
52 frame columns subjected to compressive loads remains challenging. This is owing to
53 the properties of IMCs, which include semi-rigidity and discontinuity, distinguishing
54 IMCs from the semi-rigid joints of regular frame systems and affecting the structural

55 rigidity, capacity, and resilience [49].

56 There has been extensive study on the compressive behavior of ordinary steel columns.

57 Liu et al. [50] observed that reducing the width-to-thickness ratio improved the ductility

58 and bearing capacity of the tube. Nie et al. [51] examined multiple dimensions,

59 slenderness ratios, and eccentricities while applying eccentric compression to tubular

60 columns. They discovered that the columns exhibited global buckle and significant

61 lateral deflections. Fratamico et al. [52] revealed that most of the instability in duplex

62 composite columns was caused by local buckling. While designing for compressive

63 loads, stability demands are crucial for ensuring the structure's resistance to buckle and

64 overall integrity [53]. Thus, stability design largely depends on the effective length

65 factor, based on the degree of elastic restraint at the column's ends like in IS800 [54,55],

66 NZS 3404[56], EC3:1-1 [57], CSA S16-19 [58], AISC360-16 [59], and GB 50017-

67 2017 [60]. While these works provide valuable insights, these computations might not

68 apply to the design of MSBs due to semi-rigid IMCs between columns, which alter the

69 constraint conditions at both ends of the columns and consequently impact the buckling

70 length, load and overall stability. Existing practices using alignment charts [61] and

71 simplified equations [62] may result in the design of MSBs that are non-conservative,

72 excessively conservative, economically inefficient, or inadequate, with engineers

73 assuming substantial risks due to the unpredictability and variance of the behavior to

74 conventional systems [63].

75 Despite the extensive research conducted on conventional columns, there are limited

76 investigations on the compressive behavior of MSBs. Lawson et al. [64] addressed

77 these issues with compressive tests on multi-column walls and a second-order analysis

78 method integrating hypothesized horizontal forces for module column stability analysis.

79 Hou et al. [65] and Khan et al. [66–68] investigated the buckling behavior of multi-

80 column walls and discovered that concrete cladding reduced buckling and that
81 GB50017 estimates were most reliable. Their research, however, assumed uniform load
82 distribution across all columns and disregarded the effect of adjacent modular units,
83 IMCs, and complex joint zones. Deng et al. [69], Chen et al. [70], and Khan et al. [71–
84 73] carried out investigations on single and grouped columns with shear-keyed IMCs.
85 They formulated theoretical buckling load equations and modified code predictions for
86 conservative design. However, these studies assumed complete column-to-endplate
87 welding, impractical for interior IMCs, or shear-keyed columns, which cannot design
88 other columns and do not involve semi-rigid connections between adjacent modules.
89 Zhang [74] established a simplified analysis model to calculate the column buckling
90 length; however, this model was limited to single-story and single-module columns,
91 and its relevance to the design of multi-story MSB columns requires further
92 confirmation.

93 Further studies, including those conducted by Li et al. [75,76], Farajian et al. [49], Zhai
94 et al. [77], and Wang and Su [78], examined stability calculations in sway and non-
95 sway semi-rigid steel frames with corner connections using simplified modeling
96 techniques. These studies developed alignment charts for columns K -factors and
97 proposed simplified formulas following the French Rules [79]. However, these studies
98 lacked experimental support for particular types of IMCs, missed to account for the
99 rotational stiffness of vertical and horizontal inter-modular connection and joint design
100 separately, relied on simplified formulas with limited data for fitting, and failed to
101 account for variable story heights and varied height-to-span ratios. As these alignment
102 charts rely on the designer's visual interpretation, they are also susceptible to error,
103 indicating the need for more precise and straightforward approaches [77]. Besides,
104 classification methods similar to those put forward by Farajian et al. [80,81] and He et

105 al. [82] primarily characterize the response characteristics of connections regarding
106 their strength and rotational stiffness. They provide design recommendations and
107 validate their proposed systems but neglect further discussion on aspects such as non-
108 linear analyses and the post-buckling behavior of structures under multiple limit states.
109 The current practice calculates the buckling length and load for MSB as abnormal
110 values because the actual structural mechanism cannot be correctly identified with
111 simplified connections or modeling techniques [83]. Indeed, compressive tests must
112 account for the P -delta effect, the relative stiffness of module members and IMCs, and
113 the stiffness of vertical and horizontal IMCs to produce accurate FEMs and replicate
114 the actual behavior of rotary-connected sway column-supported steel modular interior
115 frames (SCSMIFs) [2,21,84–87]. Such analyses should then focus on the non-linear
116 behavior of SCSMIFs. Consequently, a comprehensive approach featuring compressive
117 testing, accurate modeling, analysis, and design of MSB is required to address these
118 deficiencies and provide a conservative method employing equations to evaluate
119 buckling load from semi-rigid to pinned boundaries, thereby eliminating the need for
120 charts. Considering the compressive performance of these systems, specific types of
121 IMCs [88], and the stability-relevant mechanical properties of IMCs [2], it is crucial to
122 investigate global stability and reliable design methods in greater detail [83].
123 The present study intends to contribute to this field by investigating the compressive
124 behavior of SCSMIF using rotary-type IMCs, as described in Ref. [9]. Two sub-
125 assembled interior module frames were compressed as sway frames [88]. Validated
126 FEMs explored the effects of varying parameters. Experimental and FEM data verified
127 theoretical models assuming semi-rigid and pinned IMCs for predicting sub-assembled
128 rotary-connected SCSMIF buckling loads to design cost-effective, secure, and
129 sustainable MSBs.

130 **2 Compression tests on rotary-connected SCSMIFs**

131 Compressive behavior of rotary-connected SCSMIFs involves testing sub-assembled
132 interior frames designed to represent the sway frame behavior.

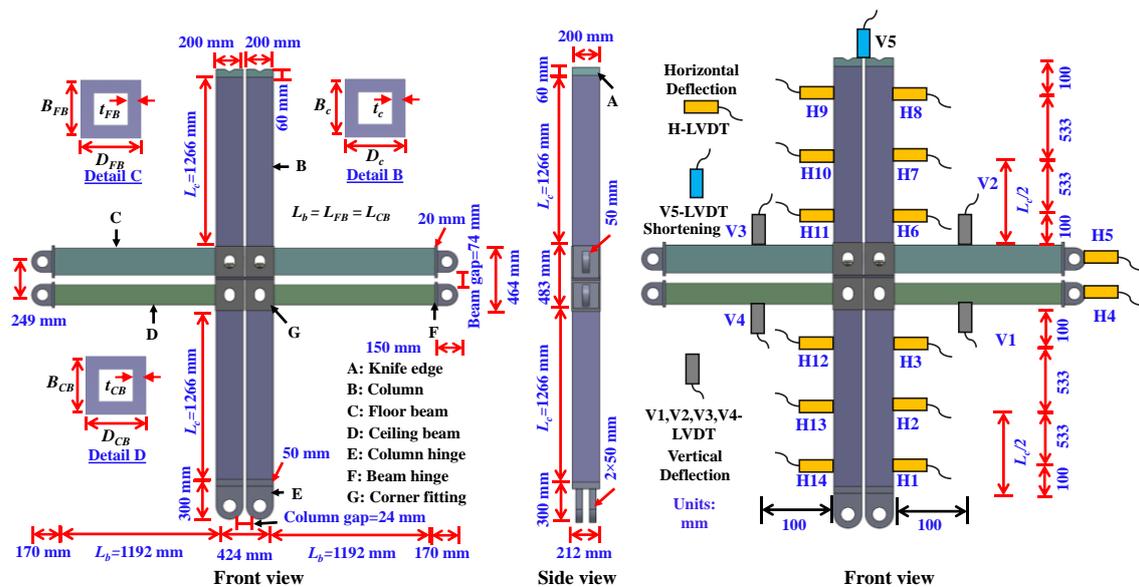
133 **2.1 Specimens design**

134 As the engineering basis for the study's members and IMC designs, the selected
135 prototype comprised the design of 8.5×3.0×3.0 and 6.7×3.0×3.0 m rotary-connected
136 modules for the construction of 5-story Ziya Shanglinyuan MSBs [2,8–10,40,89]. The
137 design was executed in compliance with GB50017-2017 [60]. Two sub-assembled
138 interior frames with unique roller support on beam ends were designed to examine the
139 compressive behavior and failure response of rotary-connected SCSMIF [88]. The
140 testing was intended to investigate the compressive behavior and failure response of
141 SCSMIFs, collect empirical data for FEM validation, and then carry out parametric and
142 theoretical research to develop buckling load models utilizing rotary-type IMCs. The
143 current investigation used sub-assembled specimens to achieve results comparable to
144 those of full-frame studies [32,33,36,46,88,90,91]. Groove welding was used to create
145 butt joints in the middle of the section of columns and beams. A gap of 74 mm was
146 maintained between floor and ceiling beams, and a gap of 24 mm between adjacent
147 columns, following the design of the prototype project to allow for access and
148 installation of MEP facilities [88].

149 **2.2 Specimens geometry**

150 The assembly process of SCSMIFs featuring rotary-type IMCs followed Refs. [2,8–
151 10,40,89]. The specimens' geometry, boundary configurations, and IMC details are
152 illustrated in **Fig. 1(a~c)** and **Table 1**. Because of the primary load-bearing member,
153 floor beams' flexural stiffness was kept higher than ceiling beams', resulting in floor
154 beams (B_{FB}) having a deeper cross section than ceiling beams (B_{CB}). While RDD-2
155 selected identical thicknesses of 8 mm, RDD-1 opted for thicker floor beams (t_{FB}) of 8

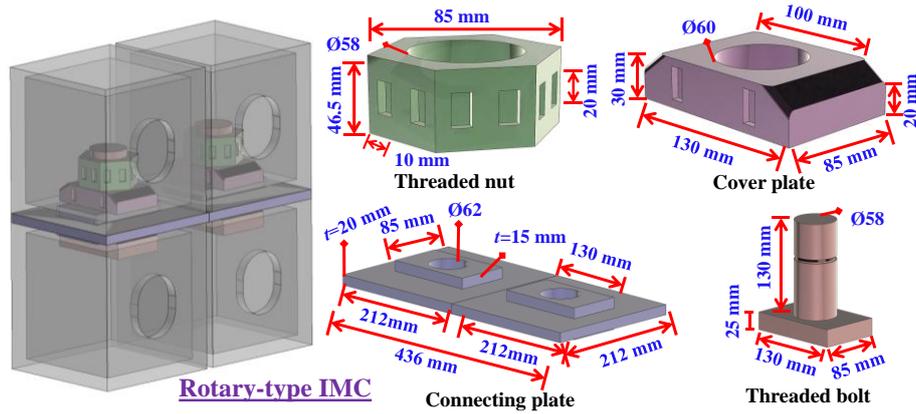
156 mm and thinner ceiling beams (t_{CB}) of 6 mm [90]. In RDD-2, t_{CB} was raised to 8 mm.
 157 To analyze the impact of beam and ITCs' relative stiffness on the compressive
 158 properties of SCSMIF, different t_{FB} and t_{CB} were tested while ensuring consistency with
 159 the original prototype design. The IMC details and the members' cross-sectional sizes,
 160 columns height (L_c), and beam lengths (L_{FB} and L_{CB}) were kept unchanged. The sub-
 161 assemblage is recognized as a precise and standard approach for determining the height
 162 and length of members at the inflection point [88]. The specimens were created with
 163 consistent dimensions of 3375 mm in height and 3160 mm in width to satisfy design
 164 and laboratory specifications. The clear height and length of the upper and lower
 165 columns (L_c) were maintained at 1266 mm, while the floor (L_{FB}) and ceiling (L_{CB})
 166 beams were kept at 1192 mm for both the left and right modules. The chosen column
 167 was a 200×200×8 mm size, with a length of 200 mm (D_c), width of 200 mm (B_c), and
 168 thickness of 8 mm (t_c). For the floor beams, a width (D_{FB}) and depth (B_{FB}) of 150 and
 169 200 mm were utilized, while a width (D_{CB}) of 150 mm and depth (B_{CB}) of 150 mm were
 170 selected for the ceiling beams.



(a) Specimen geometry and LVDTs installation details

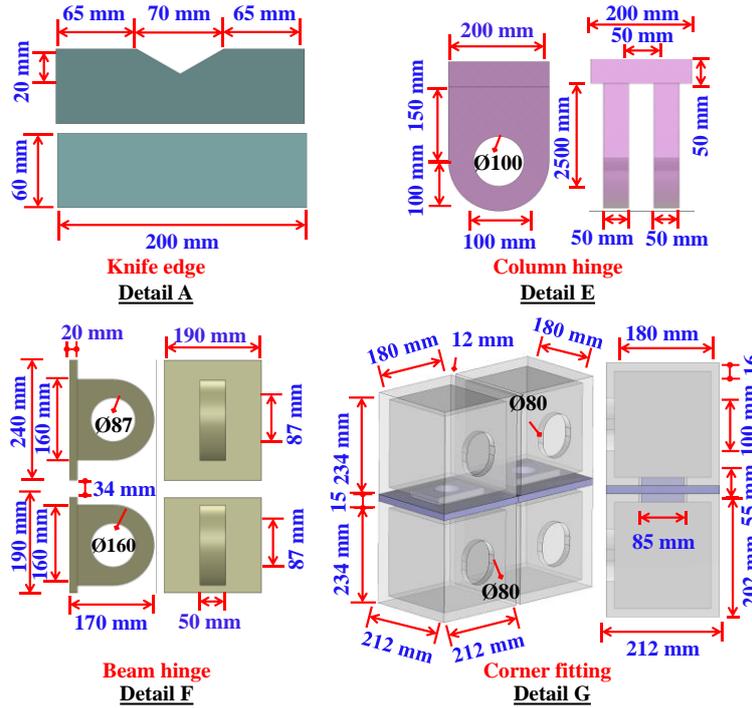
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(b) Rotary-type IMC's components design details (Prototype & Tests)



(c) Details of the selected components

174

175

Fig. 1 Details of tested sub-assembled rotary-connected SCSMIFs

176 **2.3 Material properties**

177 Steel coupons, made from the same material as the frames, were created following

178 GB/T228.1-2010 [92] for analyzing test outcomes and generating FEM. Measurements

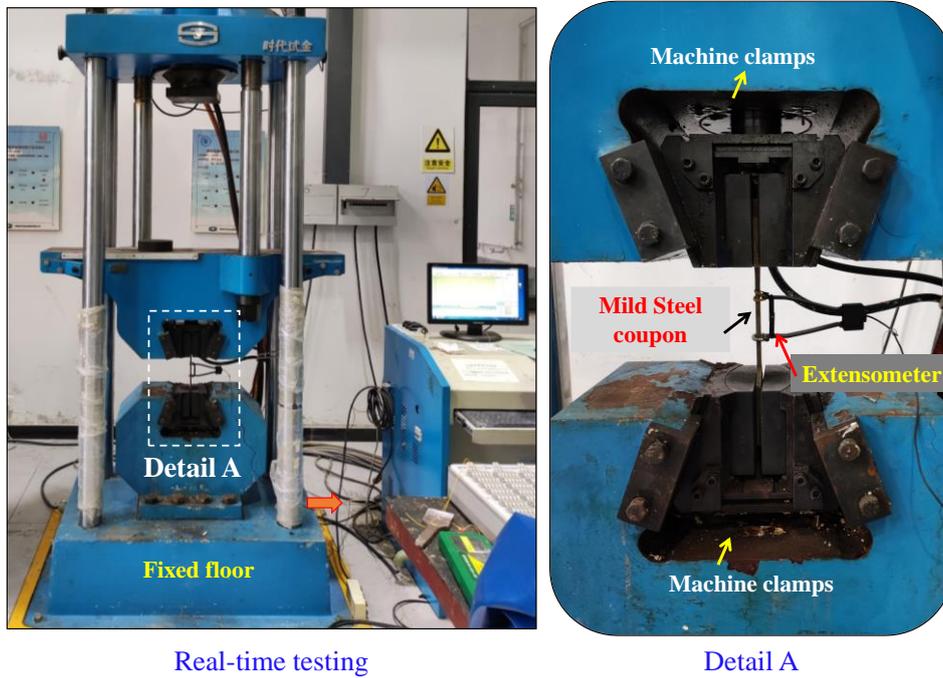
179 of the thicknesses of 15 coupons, with three for each of the five cross-sectional member

180 sizes, revealed variations that significantly impacted the strength, ductility, failure

181 modes, and yield plateau but not stiffness. The mean values of the obtained parameters

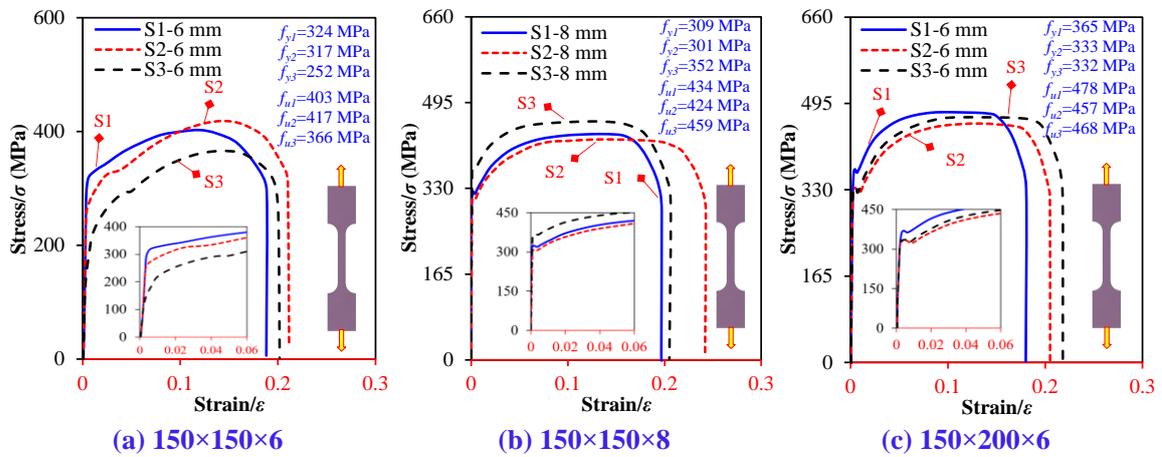
182 and thicknesses are presented in **Table 1**, while the test setup and tensile stress-strain

183 curves of the coupons are depicted in **Fig. 2** and **Fig. 3(a~e)**. These findings were
 184 employed to determine the material input parameters for the FEM.



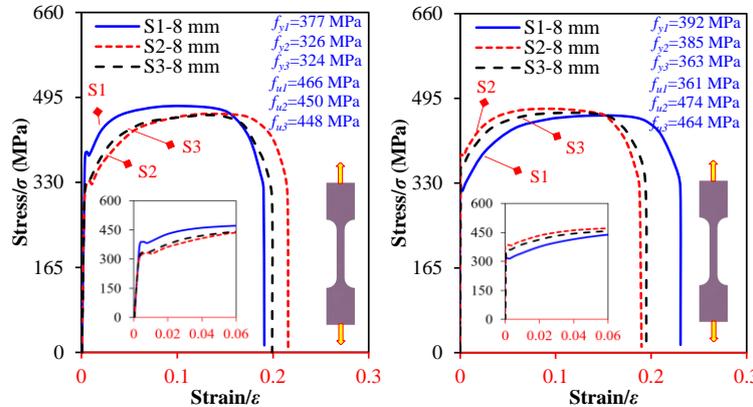
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Fig. 2 Description of steel coupons and setup for material testing



187

(a) 150×150×6 (b) 150×150×8 (c) 150×200×6



(d) 150×200×8 (e) 200×200×8

188
 189

Fig. 3 Tensile testing results of steel coupons

Table 1 Material properties, details, and findings of compression tests and FEMs on rotary-connected SCSMIFs

Item	D_{FB} (mm)	B_{FB} (mm)	t_{FB} (mm)	L_{FB} (m)	D_{CB} (mm)	B_{CB} (mm)	t_{CB} (mm)	L_{CB} (m)	$P_{u,Test}$ (kN)	$K_{e,Test}$ (kN/mm)	$\Delta_{u,Test}$ (mm)	DI_{Test} (Ratio)
RDD-1	150	200	8	1.2	150	150	6	1.2	3793	828	11.1	1.8
RDD-2	150	200	8	1.2	150	150	8	1.2	3646	778	7.2	1.5
Item	$P_{u,Test}$ (kN)	$P_{u,FE}$ (kN)	$\frac{P_{u,Test}}{P_{u,FE}}$	$K_{e,Test}$ (kN/mm)	$K_{e,FE}$ (kN/mm)	$\frac{K_{e,Test}}{K_{e,FE}}$	$\Delta_{u,Test}$ (mm)	$\Delta_{u,FE}$ (mm)	$\frac{\Delta_{u,Test}}{\Delta_{u,FE}}$	DI_{Test} (Ratio)	DI_{FE} (Ratio)	$\frac{DI_{Test}}{DI_{FE}}$
RDD-1	3793	3827	0.99	828	832	1.00	11.1	9.3	1.19	2.0	2.6	0.69
RDD-2	3646	3686	0.99	778	827	0.94	7.2	4.7	1.54	2.0	1.3	1.16
Mean			0.99			0.97			1.37			0.93
Cov			0			0.03			0.13			0.25
Item	Length (mm)	Width (mm)	Thickness (mm)	f_y (MPa)	f_u (MPa)	δ (%)	E_s (GPa)					
Beam-1	150	150	6(5.37)	298	395	22.2	201					
Beam-2	150	150	8(7.33)	321	439	23.6	209					
Beam-3	150	200	6(5.54)	344	468	24.9	208					
Beam-4	150	200	8(7.30)	342	455	23.5	210					
Column	200	200	8(7.34)	380	434	22.7	206					
Corner fittings ¹	-	-	16(15.80)	351	518	23.0	198					
IMC (ii, iii) ¹	-	-	-	360	580	34.0	206					
IMC (i, iv) ¹	-	-	-	360	610	16.0	206					

D_{FB} , B_{FB} , t_{FB} , L_{FB} ; D_{CB} , B_{CB} , t_{CB} , L_{CB} ; $P_{u,Test}$ ($P_{u,FE}$), $K_{e,Test}$ ($K_{e,FE}$), $\Delta_{u,Test}$ ($\Delta_{u,FE}$), DI_{Test} (DI_{FE}); and f_y , f_u , δ , E_s represent the floor and ceiling beam's width, depth, thickness, length; ultimate compressive resistance, initial stiffness, ultimate shortening, post-ultimate ductility index via test (FEM); material yield strength, ultimate strength, percentage elongation, elastic modulus. Note:¹ Material properties obtained according to the authors' previous Ref. [2] study. The thickness values in the bracket represent the average measured thickness of members.

Table 2 Rotary-connected SCSMIFs' buckling load comparison using tests-validated FEMs, parametric, and theoretical models

Specimen (#)	P_{cr}/FE (kN)	P_{cr}/PD (kN)	P_{cr}/SR (kN)	$\frac{P_{cr}/PD}{P_{cr}/FE}$	$\frac{P_{cr}/SR}{P_{cr}/FE}$
RDD-1	1403	939	1248	0.67	0.89
RDD-2	1367	1019	1337	0.75	0.98
Mean				0.71	0.93
Cov				0.05	0.05
FEM (#)	P_{cr}/FE (kN)	P_{cr}/PD (kN)	P_{cr}/SR (kN)	$\frac{P_{cr}/PD}{P_{cr}/FE}$	$\frac{P_{cr}/SR}{P_{cr}/FE}$
DR-1	1369	784	1210	0.57	0.88
DR-2	1369	850	1303	0.62	0.95
DR-3	1527	1155	1496	0.76	0.98
DR-4	1532	1228	1588	0.80	1.04
DR-5	1673	1272	1660	0.76	0.99
DR-6	1676	1343	1665	0.80	0.99
DR-7	1605	1228	1579	0.77	0.98
DR-8	1116	432	932	0.39	0.84
DR-9	1856	1565	1810	0.84	0.98
DR-10	1299	727	1201	0.56	0.92
DR-11	1768	1652	1862	0.93	1.05
DR-12	1316	840	1294	0.64	0.98
DR-14	851	275	350	0.32	0.41
DR-26	1429	862	1210	0.60	0.85
DR-28	1839	1092	1753	0.59	0.95
DR-40	1427	952	1405	0.67	0.98
DR-42	1746	1111	1805	0.64	1.03

DR-44	1829	1182	2017	0.65	1.10
DR-47	1415	862	1064	0.61	0.75
DR-48	1386	862	1064	0.62	0.77
DR-49	1379	1019	1337	0.74	0.97
DR-50	1347	1019	1337	0.76	0.99
DR-51	1331	1092	1478	0.82	1.11
DR-52	1342	1092	1478	0.81	1.10
DR-55	2086	1019	1337	0.49	0.64
DR-56	2078	1019	1337	0.49	0.64
DR-78	1457	1019	1337	0.70	0.92
DR-79	1348	1019	1337	0.76	0.99
DR-80	1370	1019	1337	0.74	0.98
DR-81	1342	1019	1337	0.76	1.00
DR-82	1344	1019	1337	0.76	0.99
DR-83	1129	1019	1337	0.90	1.18
DR-84	1368	1019	1337	0.74	0.98
DR-85	1375	1019	1337	0.74	0.97
DR-86	1423	1019	1337	0.72	0.94
DR-87	1087	1019	1337	0.94	1.23
Mean				0.69	0.95
Cov				0.20	0.16

P_{cr}/FE , P_{cr}/PD , and P_{cr}/SR define the buckling load of sub-assembled rotary-connected SCSMIFs via FEMs and theoretically-obtained pinned and semi-rigid IMCs models.

190 **2.4 Test setup**

191 Rotary-connected SCSMIF specimens were mounted on the compressive testing setup,
192 as shown in **Fig. 4(a,b)**. Before mounting the specimens on the setup, the left, right,
193 upper, and lower frame skeletons were joined on the ground using rotary-type IMCs
194 following Refs. [2,8–10,40,89]. A vertical hydraulic jack applied a compressive force
195 to the upper columns of specimens. Column roller supports were installed above the
196 jack to allow lateral movement of the specimen while maintaining the compressive load,
197 even during specimen shortening and lateral deflection. The jack base was fixed to a
198 load sensor using a plate and threaded bolts to record the reaction forces. The load
199 sensor was attached at both the top and bottom ends by a jack and double-column knife-
200 edge support using welded plates and threaded bolts. To ensure that the load was
201 transferred promptly to both adjacent upper columns while allowing for rotation, a
202 double-column knife-edge support was welded to the bottom end of the plate, keeping
203 the support flat. The double-column knife-edge support facilitated in-plane rotation,
204 while out-of-plane rotation was restricted. A pin cell supported the lower columns base,

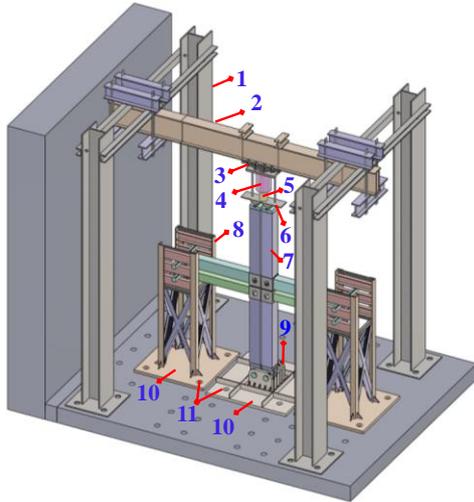
205 which provided double-column hinged support that prevented specimens from
206 translating in-plane and out-of-plane directions while allowing in-plane rotation. Roller
207 supports were installed on the ends of beams of both the right and left modules to restrict
208 vertical translation while allowing for in-plane translation and rotation. In Ref. [88], a
209 similar testing method was suggested as a standard for simulating the behavior of
210 SCSMIF under sway-frame conditions, as detailed in [2,8,10,36,46,48,89,91,93]. A
211 laser level ensured the specimen and load setup were aligned correctly. Once the
212 alignment was confirmed, the jack was pressed slightly to ensure they remained vertical.
213 Measuring devices were then installed before the formal testing began.

214 According to GB/T50344-2019 [94], the loading process was divided into preloading
215 and formal loading, with unloading occurring in both stages. The measuring devices'
216 precision was validated using a preload equivalent to $0.2P_u$ (SCSMIF's ultimate
217 compressive resistance). The specimens were kept at the preload level for two minutes
218 before being wholly unloaded for another two minutes. In order to account for
219 structures with unpredictable yield displacements, a loading approach that combined
220 force and displacement control was employed [95]. Following force loading until
221 yielding, a displacement loading rate of 0.05 mm/min was adopted until the load
222 dropped to 85% of P_u [96]. Once the non-linear segment of the load-shortening curves
223 began, displacement loading was accomplished using reaction forces captured through
224 the load sensor and shortening measured by the vertical LVDT (V5 in **Fig. 1(a)**).

225 Strain gauges were utilized to evaluate deformation and force transfer mechanisms [97].
226 As shown in **Fig. 5(a,b)**, several strain gauges were installed on the upper columns,
227 lower columns, floor beams, ceiling beams, and upper and lower corner fittings to
228 assess local elastic or plastic buckling that occurred either before or after material yield
229 [98]. Due to the susceptibility of columns to local buckling, strain gauges were

230 vertically positioned along their height and on the corner fittings. Furthermore, it was
231 anticipated that columns near the IMCs or the ends would be subjected to higher
232 stresses and ultimately fail. As a result, strain gauges were mounted in various positions
233 on the upper columns, including at the ends and midpoints. Similarly, strain gauges
234 were placed on the lower columns where higher stresses were expected, mainly near
235 the IMCs and ITCs zones up to mid-height. No strain gauges were attached to the mid-
236 to-bottom section of the lower columns, as stresses were less apparent in those regions.
237 As the beams were allowed to rotate and the stress levels near the ITCs and IMCs were
238 high, strain gauges were affixed to a distance of up to 200 mm from these locations.
239 Since the limited space available for work due to the small gap between the floor beam
240 and ceiling beam and the adjacent upper column and lower column, no strain gauges
241 were attached in those areas. A total of 73 strain gauges were utilized in RDD-1, while
242 RDD-2 had 66.

243 **Figure 1(a)** shows fourteen horizontal LVDTs placed vertically on adjacent modular
244 units' right and left sides, including lower columns, upper columns, floor beams, and
245 ceiling beams (H1-H14), to measure deflection, translation, sway, or buckling. A
246 vertical LVDT (V5) was also mounted on a double-column jack-fixed knife-edge
247 support to evaluate end-shortening. Four other LVDTs (V1-V4) near ITCs and IMCs
248 measured the vertical deflection of ceiling and floor beams. These deflection, end-
249 shortening, strain, and load measurements were recorded using a data recorder.

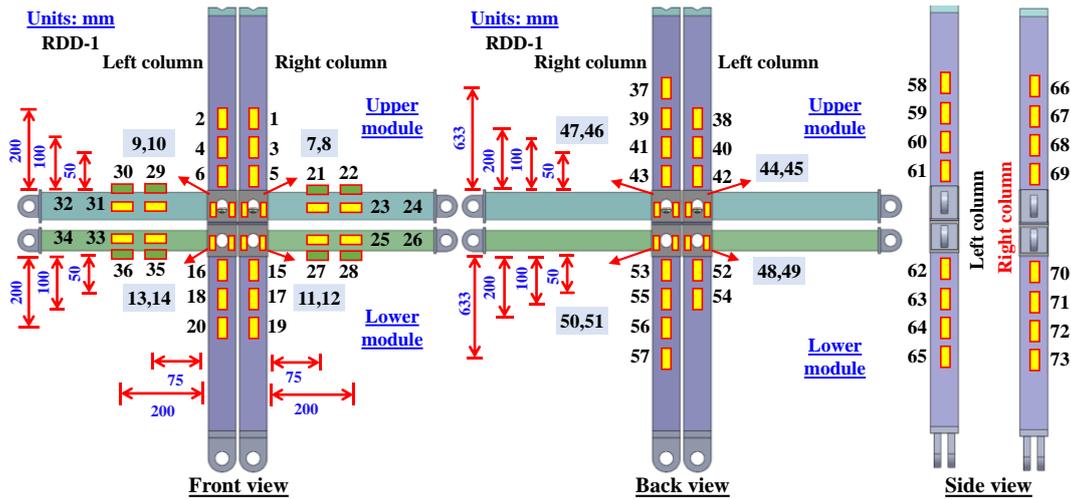


(a) Test setup schematic diagram

(b) Test setup and specimen before the test

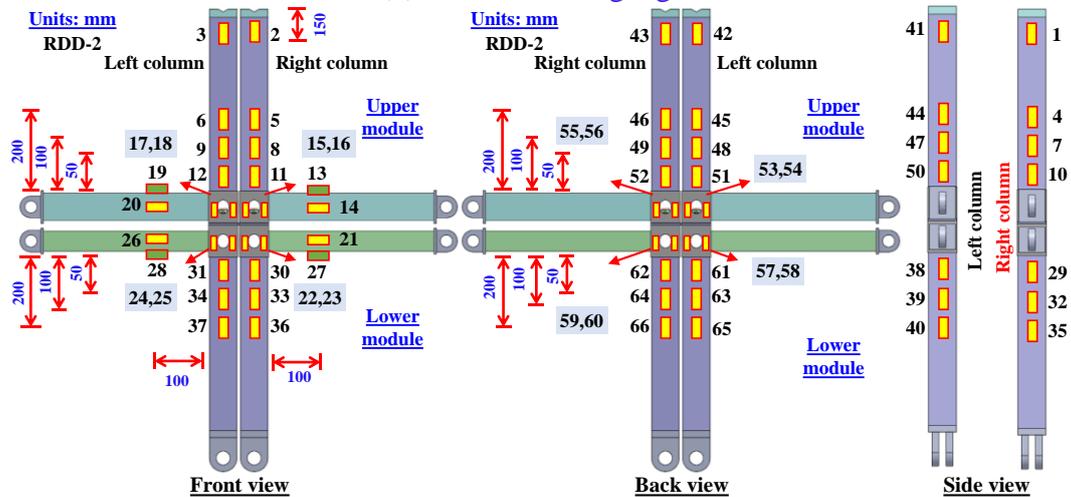
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Fig. 4 Generalized compression tests setup on sub-assembled rotary-connected SC SMIFs. (1-Reaction frame; 2-Reaction beam; 3-Column roller support; 4; Vertical hydraulic jack; 5-Load sensor; 6-Double-column knife-edge support; 7; SC SMIF specimen with rotary-type IMCs; 8-Beams roller supports; 9- Double-column hinged support; 10-Pedestal; 11-Anchor bolt holes)



(a) RDD-1 strain gauges

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(b) RDD-2 strain gauges

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Fig. 5 Specific strain gauge locations on rotary-connected SC SMIFs

259 **3 Experiment outcomes**

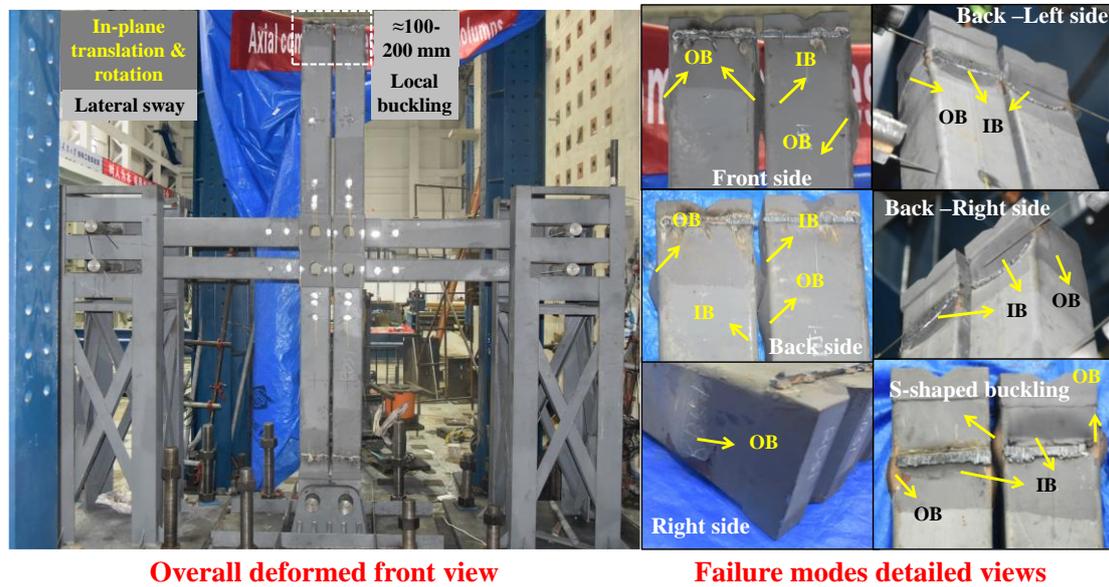
260 **3.1 Failure modes**

261 As depicted in **Fig. 6(a,b)**, the relative rigidity of the floor beams, ceiling beams, upper
262 columns, and lower columns substantially affected the in-plane translations of RDD-1
263 and RDD-2. The SCSMIF in RDD-2 was more flexible than in RDD-1, resulting in
264 increased lateral instability. Non-rigid constraints enabled the formation of gaps,
265 rotations, and translations of the upper and lower frame skeletons around the rotary-
266 type IMCs, which pinned or rigid assumptions cannot simulate. Upon reaching their
267 compression capacity, the SCSMIFs experienced simultaneous local buckling of
268 adjacent upper columns, forming S-shaped inward and outward patterns with similar
269 placements on adjacent columns. Once buckling occurred, the SCSMIFs could not
270 support additional load, necessitating a loading halt for safety purposes. Strain values
271 indicated that in RDD-1, local buckling happened at a distance of 100-200 mm on all
272 top faces of upper columns due to reduced sway, whereas in RDD-2, it occurred at the
273 base of the upper columns approximately 100-200 mm from the corner fittings due to
274 significant bending. Columns of RDD-1 bulged outward on their inner sides, while
275 columns of RDD-2 bulged inward, resulting in a double S-shaped local buckling and
276 preventing collisions on the inner sides.

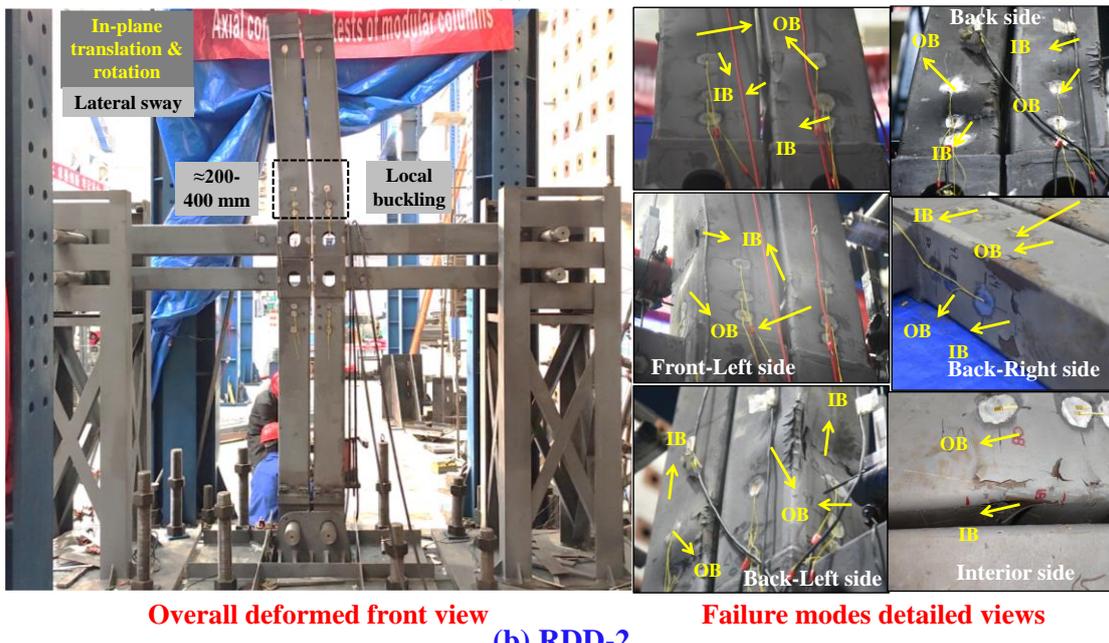
277 Comparisons of material yield in **Fig. 8(a~c)** revealed that the upper columns
278 experienced local elastic buckling in regions aligned with or opposite the bending
279 direction. In other areas, local plastic buckling was observed. The beams and corner
280 fittings did not buckle or yield before reaching the SCSMIF's capacity, but several
281 upper column regions yielded, indicating that the upper columns bore the primary load.

282 The absence of out-of-plane translation and rotation adjacent to the bending sides meant
283 that the compressive behavior of SCSMIF was primarily controlled in the in-plane
284 direction. Although the local buckling location varied between the two specimens,

285 IMCs could transmit force to adjacent members without any localized failure,
 286 suggesting that rotary-connected SCSMIFs could ensure the safety and integrity of
 287 MSBs under compressive loads.



288 **(a) RDD-1**



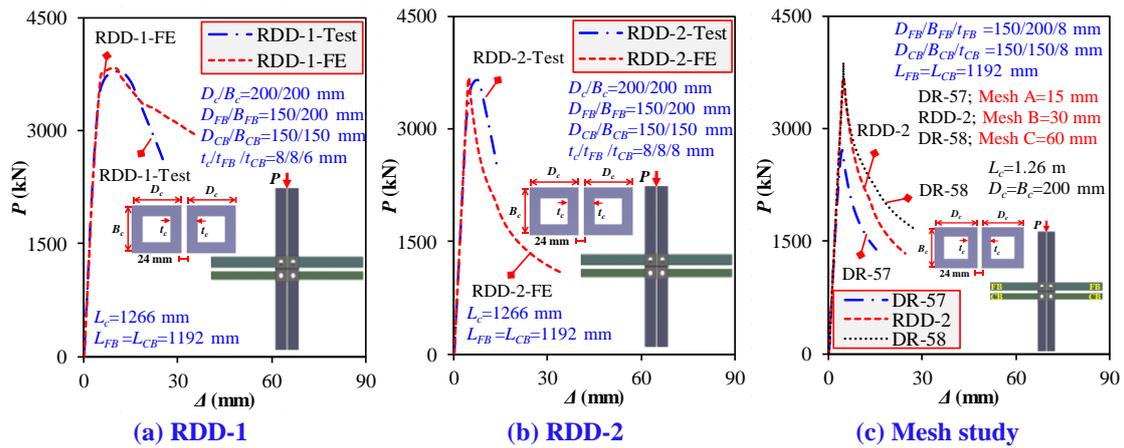
(b) RDD-2

Fig. 6 Failure modes of rotary-connected SCSMIFs (IB/OB, inward/outward buckling)

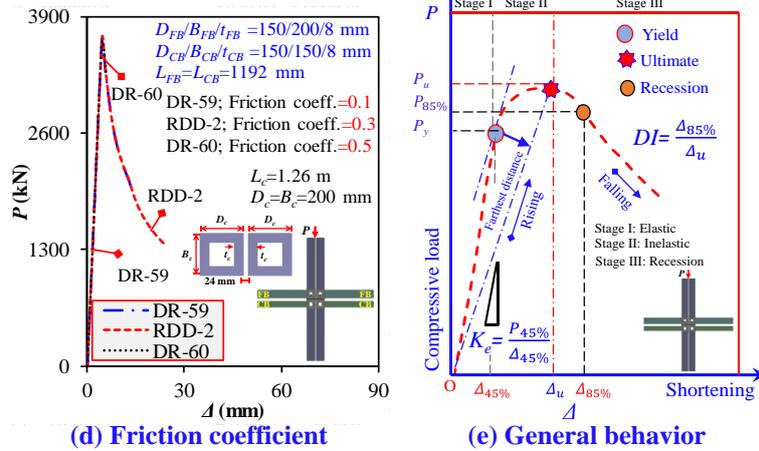
292 **3.2 Load-shortening curves**

293 **Figures 7(a,b) and 7(e)** illustrate load-shortening curves and the general behavior of
 294 RDD-1 and RDD-2 rotary-connected SCSMIFs. These curves reveal the elastic (I),
 295 inelastic (II), and recession (III) stages of the SCSMIFs, which can be used to calculate

296 their ultimate compressive resistance (P_u), ultimate shortening (Δ_u), initial stiffness (K_e),
297 and ductility index (DI) [99,100]. In stage I, the load increases proportionally with
298 shortening until the yield strength (P_y) is attained. After reaching P_y , the capacity
299 increases as the stiffness of curves decreases because buckling and bending stresses are
300 exceeded at various upper column locations, such as the top in the RDD-1 and the
301 bottom in the RDD-2. The curves take on a parabolic shape throughout stage II,
302 beginning from P_y and continuing until P_u . Meanwhile, a symmetrical inward and
303 outward local buckling pattern emerges simultaneously on both adjacent upper columns.
304 When comparing RDD-1 and RDD-2, it was found that the P_u of RDD-2 was 3.9%
305 lower, and K_e decreased by 6%. This suggests that RDD-2 was less rigid, leading to
306 more significant sway, bending stresses, and secondary moment effect, which
307 decreased SCSMIF strength and stiffness. Strain values confirmed that local elastic
308 buckling mainly resulted in noticeable SCSMIF compressive behavior reduction and
309 premature instability in the RDD-2. As shown by Δ_u , RDD-2 demonstrates 35% less
310 pre-ultimate ductility than RDD-1. This results from the SCSMIF's reduced flexibility,
311 which increases buckling strain and ductility. In contrast, RDD-1 can better resist
312 compressive forces, experience less sway and deformation, and minimize bending and
313 shear stresses in its members, attaining higher strength, stiffness, and ductility levels.
314 Stage III is characterized by decreased capacity, an abrupt increase in deflection, and
315 severe local buckling; thus, the DI of SCSMIFs are compared. In this post-ultimate
316 stage, the recession follows the ultimate stage and is marked by an abrupt decline in
317 capacity that may persist until a larger end-shortening [101–107]. RDD-1 had a 48%
318 higher ductility index and a superior recession stage than RDD-2, indicating it can
319 withstand more significant deformations while preserving its structural integrity and
320 preventing stress transmission to its components.



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Fig. 7 Load-shortening curves of rotary-connected SCSMIFs

324 **3.3 Load-strain curves**

325 **Figures 8(a~c) and 9(a~m)** depict load-strain curves for SCSMIFs' columns and strain
 326 magnitudes on upper and lower corner fittings, upper columns, lower columns, floor
 327 beams, and ceiling beams, highlighting the strain amount, yield strain, and local
 328 buckling sites. The curves' linear, non-linear, and recession sections identify test failure
 329 types and local buckling locations, whether apparent or not. As the load increases, the
 330 stresses increase until local buckling is indicated by the inversion, overturning, or rapid
 331 decline of strain curves and exceptionally high strain values. Curves that reverse before
 332 or around the yield strain show stresses below the material's yield strength, resulting in
 333 elastic buckling. In contrast, plastic buckling occurs when stresses exceed the yield
 334 strain. Additionally, the appearance of overturning curves during the recession phase
 335 following the yield strain indicates the emergence of severe local plastic buckling. The

336 failure modes, significant strain values, and curves indicate the presence of symmetrical
337 S-shaped local inward and outward buckling in the lower regions of both adjacent upper
338 columns in RDD-2 and top areas of RDD-1, with these buckling patterns occurring
339 circumferentially on all column faces at 100-200 mm. Despite evident local buckling
340 in upper columns, no buckling or yielding was detected in many areas of lower columns.
341 Furthermore, most other members, such as floor beams, ceiling beams, and lower and
342 upper corner fittings, did not yield because the maximum strain values recorded in tests
343 were generally low. This indicates no adverse localized deformation in the rotary-type
344 IMCs and SCSMIFs' other members, except for the columns, which were the primary
345 load-carrying members.

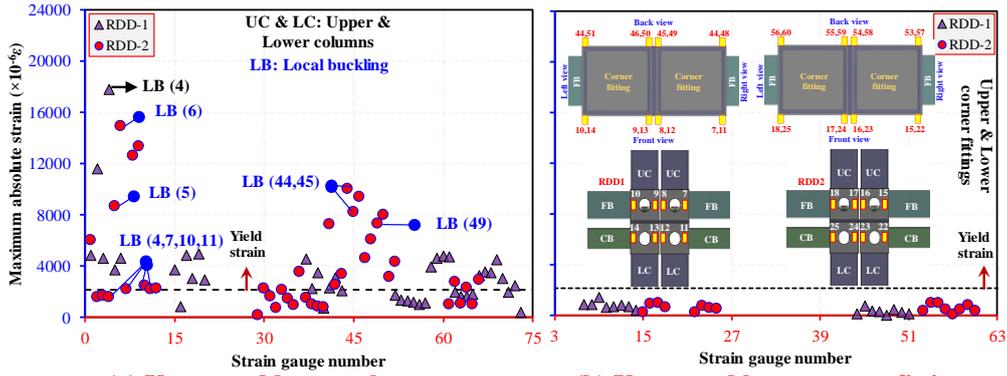
346 Since no strain gauge was on the RDD-1 local buckling location, more extensive strain
347 measurements or curve overturning were observed at top locations 2 and 4 in RDD-1.
348 Likewise, bottom locations 4, 5, 6, 7, 10, 11, 44, 45, and 49 in RDD-2 demonstrated
349 the existence of local buckling on each face of the upper columns, displaying both local
350 elastic and plastic buckling. In RDD-2, upper column portions experiencing bending
351 stresses in or opposite directions of the beams exhibited elastic buckling, while adjacent
352 upper column sides not exposed to bending underwent plastic buckling. For example,
353 locations 4, 7, and 10 in RDD-2 displayed local elastic buckling, indicating that bending
354 and the secondary moment effect prevented upper columns and other members from
355 completely yielding.

356 **3.4 Load-deflection curves**

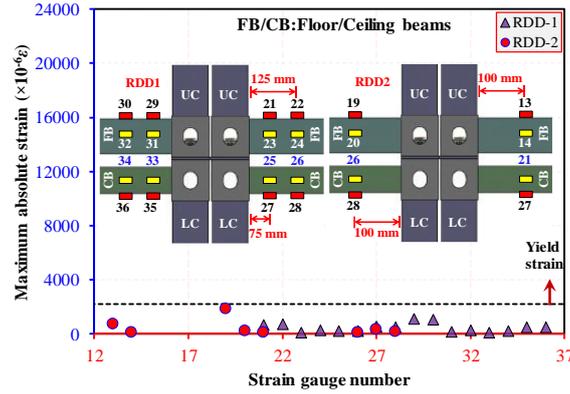
357 The load-deflection curves in **Fig. 10(a~h)** show linear, non-linear, and recession
358 phases followed by a curve drop. The length of the curves indicates in-plane translations
359 as measured by the deflection amount. The varying stiffness reduction of each curve
360 reveals that members respond differentially to the magnitude of the P -delta effect.

361 However, the relative members' stiffness effect caused SCSMIF to be more flexible in
362 RDD-2. This resulted in higher lateral deflection of members than in RDD-1. As the
363 load increases, the deflection also rises, stabilizing when the ultimate capacity is
364 reached, followed by a pause in load but deflection increments. The orderly increase in
365 deflection from lower to upper columns on the right and left frame skeletons indicates
366 the presence of SCSMIF sway and local buckling. The maximum deflection at the top
367 of the upper columns in RDD-2 suggests the instability of SCSMIFs due to local and
368 global failure. Non-identical deflections of floor and ceiling beam in vertical and lateral
369 directions and their difference imply a degree of relative rotation between the upper and
370 lower frame skeleton at rotary-type IMCs. This cannot be simulated as rigid or pinned
371 [32].

372 The deflection curve validates the test failure modes. The apparent deflection difference
373 between the top and bottom ends of the upper columns, as illustrated by H8/H9 and
374 H6/H11, indicates that local buckling began near H6/H11, followed by an increase in
375 lateral deflection at the top H8/H9 in RDD-2. In addition, the deflection curves of the
376 right and left frame skeletons exhibit an apparent resemblance in RDD-1 and RDD-2,
377 suggesting that both structures behaved symmetrically, resulting in identical local
378 buckling at corresponding locations on adjacent upper columns. These findings indicate
379 that the rotary-type IMCs transmitted forces and that the left frame skeleton had no
380 detrimental effect on the deformation behavior of the right frame, consistent with
381 existing literature [49].

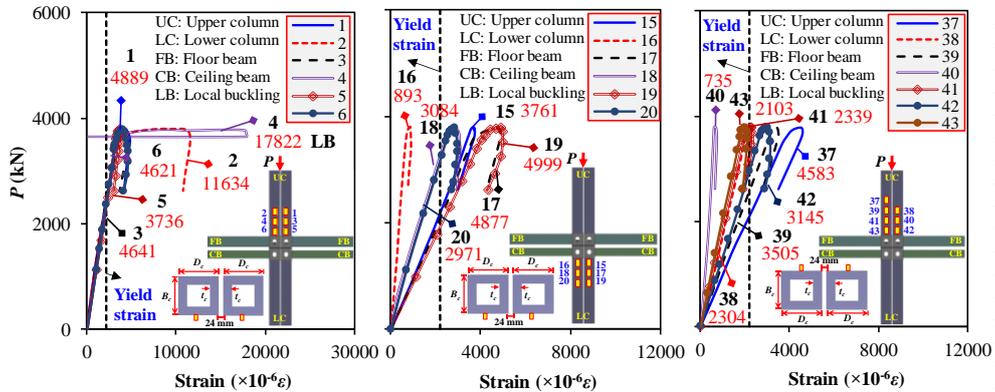


(a) Upper and lower columns (b) Upper and lower corner fittings

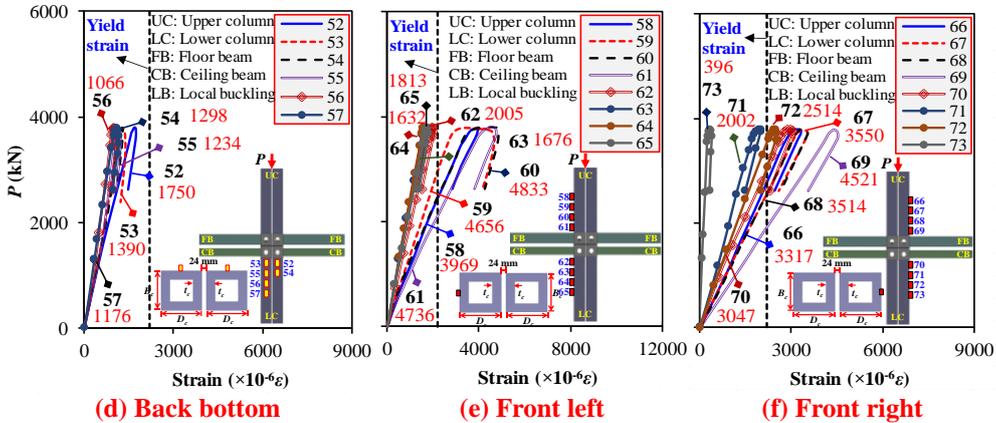


(c) Floor and ceiling beams

Fig. 8 Part-wise maximum absolute strain distribution of SCSMIFs



(a) Front top (b) Front bottom (c) Back top



(d) Back bottom (e) Front left (f) Front right

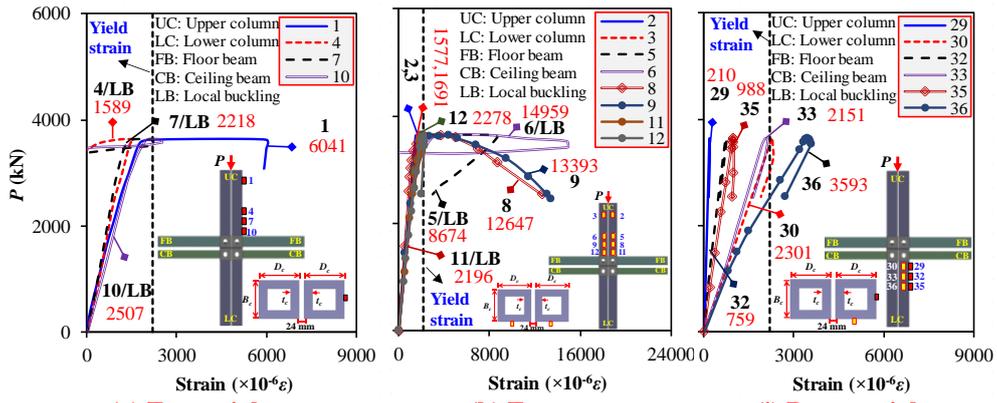
RDD-1 columns strain curves

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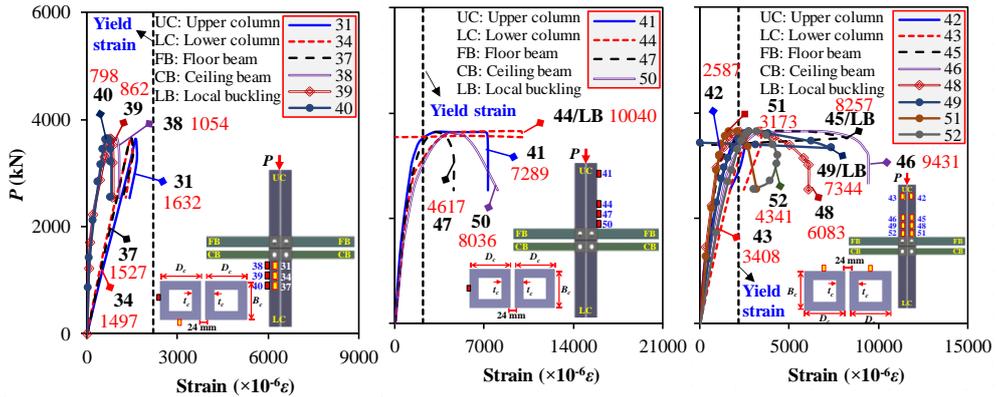


387

(g) Front right

(h) Front top

(i) Bottom right

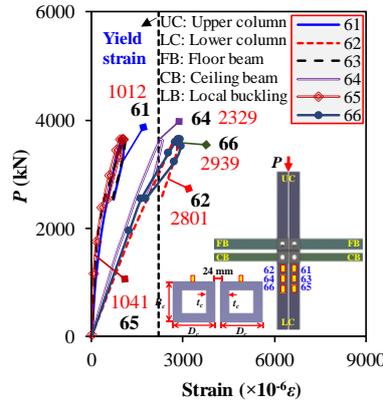


388

(j) Bottom left

(k) Back right

(l) Back top



(m) Back bottom

RDD-2 columns strain curves

Fig. 9 Load-strain curves at columns of SCSMIFs

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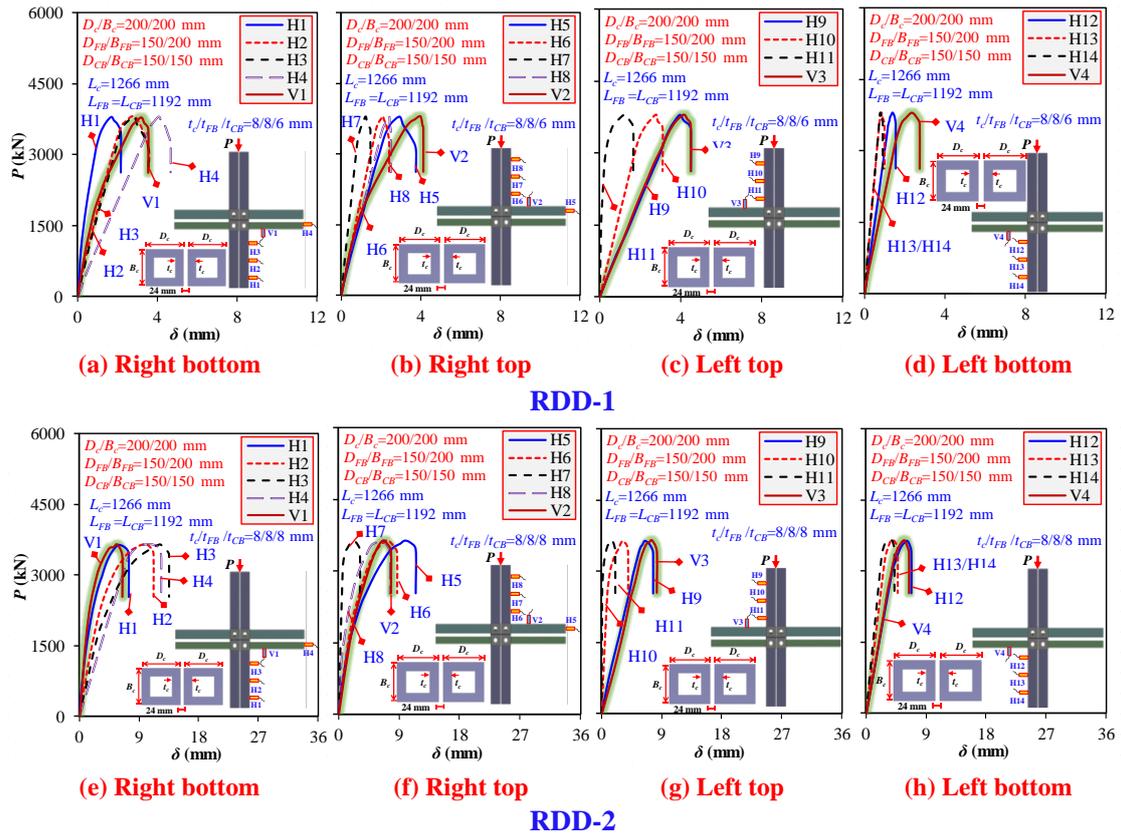


Fig. 10 Load-deflection curves at various parts of SCSMIFs

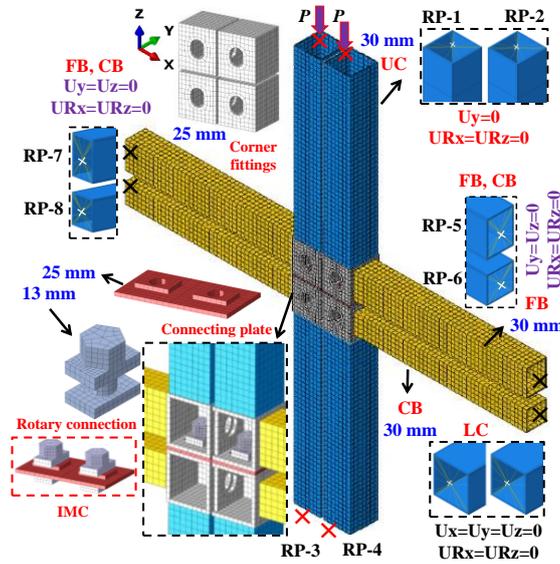
4 Finite element analysis of rotary-connected SCSMIFs

Although the tests provided valuable information on the behavior of the SCSMIFs, they did not fully assess their overall instability or the effect of varying parameters on their elastoplastic compressive behavior. A 3D non-linear FEM was developed to address these limitations using data extracted from load-shortening curves and failure modes observed during testing.

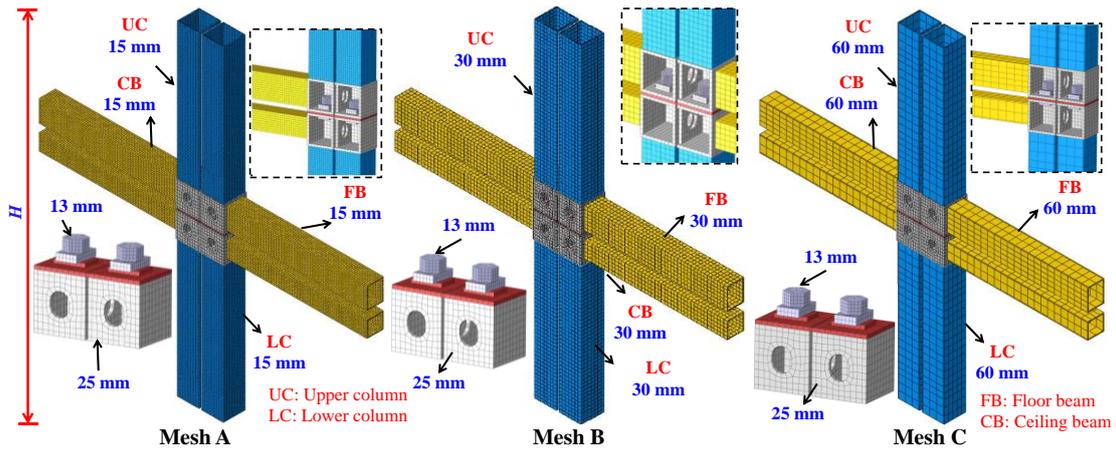
4.1 Development of finite element model

ABAQUS [108] was utilized for finite element modeling and analysis. Elastic buckling analysis was performed with the ABAQUS/Linear perturbation buckle-type solver and the subspace iteration approach to determine buckling loads and modes. Non-linear analysis was conducted with the ABAQUS/static Riks-type solver to investigate load-shortening behavior and failure mechanisms. The bilinear kinematic hardening and von

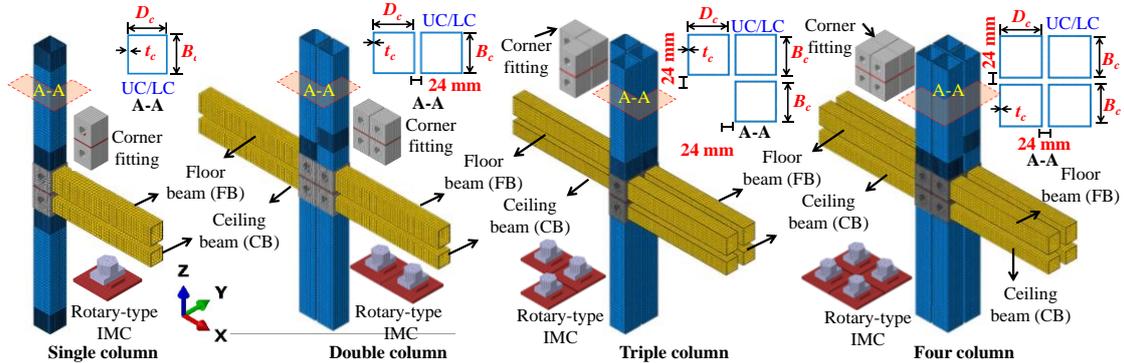
405 Mises yield criteria were applied to all components, with material properties taken from
 406 **Table 1** [32]. A Poisson's ratio was set to 0.3 [89].



407 (a) Verified FEM mesh details of tested specimens



408 (b) Mesh details for validation of tested specimens and parametric study



409 (c) Varying column number details of FEMs

410 **Fig. 11** Details of experimentally-validated and parametric FEMs of rotary-connected
 411 SCSSMIFs

412 4.2 Mesh modeling

413 **Figure 11(a)** shows the mesh model for RDD-1 and RDD-2, which includes corner
414 fittings, upper columns, lower columns, floor beams, ceiling beams, and rotary-type
415 IMCs. Different mesh sizes and the specifics of various column numbers are depicted
416 in **Figs. 11(b,c)**. All member dimensions were designed to be comparable to the actual
417 specimens and were modeled as 8-node linear brick, reduced integration with hourglass
418 control elements (C3D8R) [109]. The mesh convergence study performed to assess the
419 element size suitability involved comparing the results from mesh A, B, and C with test
420 $P-\Delta$ curves, as displayed in **Figs. 7(a~c)**. To more accurately replicate the observed
421 local buckling and deformation characteristics, the column locations at the upper edges
422 for RDD-1 and the lower area for RDD-2 were densely meshed at 100-200 mm with 5
423 mm, while other regions and parts uniformly meshed. The same technique was followed
424 in Refs. [8,32] to capture the formation of potential local buckling at members. The
425 corners of columns and beams were partitioned at their thickness to create the structured
426 mesh [31,110–112]. Mesh A and B mimicked local buckling and deformation more
427 precisely than Mesh C, as evidenced by the RDD-2 column bottoms in **Fig. 16(c)**. When
428 the mesh size increased from 15 to 30 and 60 mm, $P_u (K_e)$ increased by up to 36% (4%)
429 and 43% (8%), while Δ_u increased by 8%, and DI was reduced to 3%. Moreover, the
430 mesh refinement technique used in Refs. [8,32] capture the formation of prospective
431 local buckling at members did not affect the location of the failure mode where initial
432 imperfections play a crucial role in causing buckling failure. Nevertheless, mesh
433 refinement made the deformation patterns of previously identified test failure locations
434 more apparent. The impact of mesh sizes on compressive behavior was substantial,
435 revealing that Type B mesh yielded the most precise results, emphasizing compression
436 tests on rotary-connected SCSMIFs to determine the appropriate mesh density.

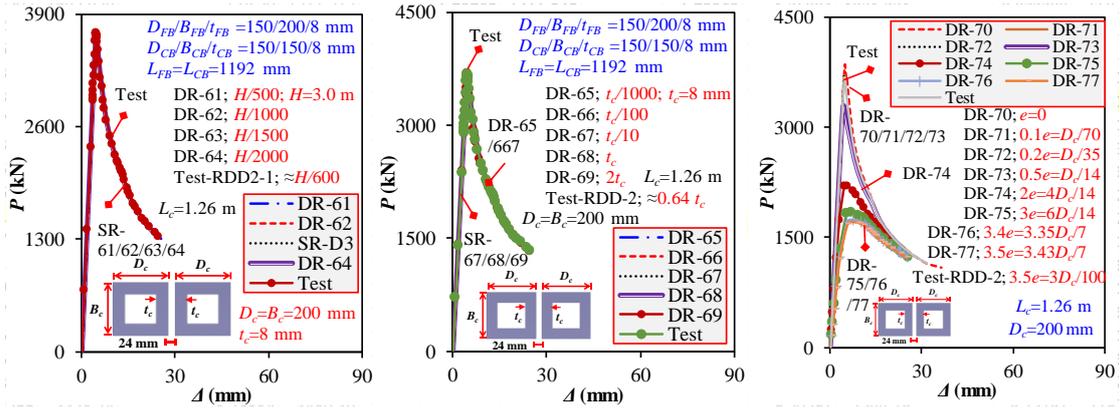
437 **4.3 Loading and boundaries**

438 The columns and beams were subjected to loading and boundary conditions by defining
439 reference points (RP-1~RP-8) on cross-sections and applying surface-based coupling
440 constraints to limit translations and rotations at the coupling nodes. The lower columns
441 were restricted in all directions, while the upper columns, floor beams, and ceiling
442 beams were permitted in-plane translation, but beam vertical translations were
443 restrained. The beams and columns were only allowed in-plane rotation, as their out-
444 of-plane rotation was restricted. The upper columns were subjected to an equal
445 compression force as a displacement-controlled loading at their respective reference
446 points to achieve shortening, and loading was determined by summing up both columns.
447 The ITCs were achieved by welding columns and beams to corner fittings using "tie
448 constraint" via surface-to-surface contact. The interaction between corner fittings,
449 connecting plate, and rotary-type IMC components were simulated as surface-to-
450 surface contact with "hard contact" as normal and "finite sliding" as tangential behavior,
451 using a friction coefficient of 0.3 displayed in **Fig. 7(d)** [109,113,114]. The specimens
452 used in this study were hot-rolled sections with low bending, welding deformation, and
453 residual stresses. Hence, the effects of bending, welding, and temperature residual stress
454 were not considered in the FEM analysis [115,116].

455 **4.4 Initial imperfections**

456 The rotary-connected SCSMIFs consisted of upper columns, lower columns, floor
457 beams, ceiling beams, corner fittings, and IMCs, all of which may have imperfections
458 before and after installation that are challenging to measure with conventional
459 techniques [100]. For reliable outcomes, design standards advise imperfections
460 between $t_c/500$ to $t_c/200$ and $L_c/1000$ to $L_c/1996$ [117]. However, in SCSMIFs,
461 imperfections can be attributed to local and global factors, i.e., column thickness (t_c),
462 frame height (H), and eccentricity (e) [27,71,72,118]. This study selected specific

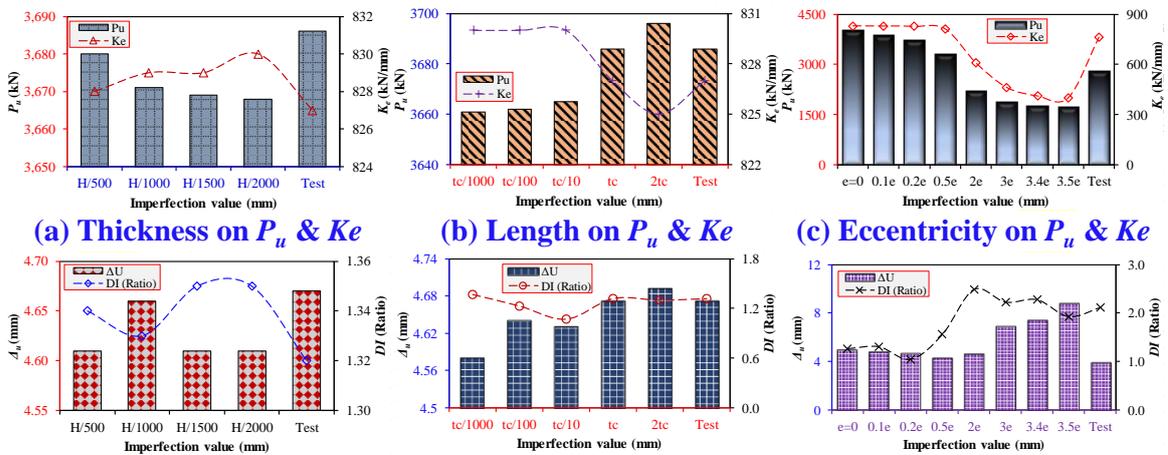
463 values for height imperfection, thickness imperfection, and load eccentricities,
464 including $H/500$, $H/1000$, $H/1500$, and $H/2000$ for height imperfections; $t_c/1000$, $t_c/100$,
465 $t_c/10$, t_c , and $2t_c$ for thickness imperfection; and 0 , $D_c/70$, $D_c/35$, $3D_c/100$, $D_c/14$, $4D_c/14$,
466 $6D_c/14$, and $3.43D_c/7$ for load eccentricities. The buckling modes depicted in **Fig.**
467 **14(a~f)** were determined using an Eigenvalue analysis. A non-linear Riks analysis was
468 performed, including imperfections and load eccentricities per test failure mode shown
469 in **Fig. 6(a,b)**, such as in the frame sway direction. In comparing the buckling modes
470 obtained from Eigenvalue analysis to the failure modes in Riks analysis, the lowest
471 buckling mode (Mode 1) was selected for RDD-1 and RDD-2 to incorporate
472 imperfections, taking into account their precision as recommended in Ref. [73]. The
473 critical buckling loads and accompanying mode shapes were then compared with the
474 loads at which failure occurred in the Riks analysis for a reliable description of the
475 structure's behavior. The imperfection amplitude determined in **Fig. 12(a~c)** was used
476 for rotary-connected SCSMIFs and FEMs of parametric studies in **Supplementary**
477 **Table A1**. The local imperfection of $H/600$ or $0.64t_c$ and global imperfection of
478 $e=3D_c/100$ produced the closest results to the test outcomes of RDD-2. Like RDD-1's
479 test results, the local imperfection of $H/600$ or $0.64t_c$ and global imperfection of
480 $e=7D_c/500$ produced the most comparable outcomes. **Figures 12(a~c)** and **13(a~f)**
481 demonstrate that increasing H or t_c imperfection values had no noticeable impact on P_u
482 (K_e) and Δ_u (DI). The imperfection values, particularly the load eccentricity values,
483 were influential in determining the failure location in the specimens and reorganizing
484 them on top of the upper columns in RDD-1 and the bottom of the upper columns in
485 RDD-2. Since translation or rotation is permitted in SCSMIFs, the eccentricity effect
486 on $P-\Delta$ curves, P_u , K_e , Δ_u , DI , and failure modes was substantial, as shown in
487 **Supplementary Figure B1**.



488
489

(a) Local imperfection H (b) Local imperfection t_c (c) Global imperfection e

Fig. 12 Impact of initial imperfections amplitudes on load-shortening curves

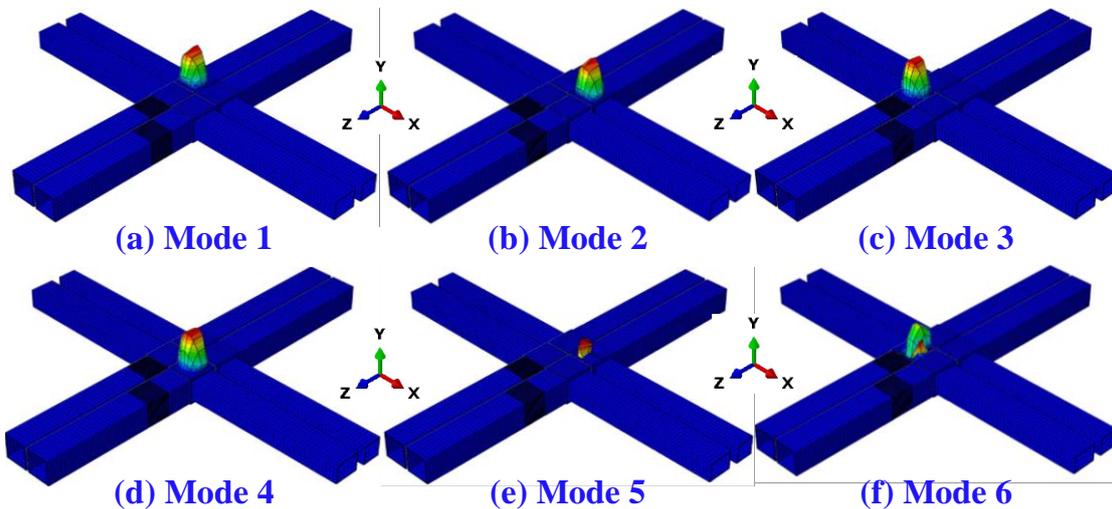


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491

(a) Thickness on P_u & K_e (b) Length on P_u & K_e (c) Eccentricity on P_u & K_e

(d) Thickness on A_u & DI (e) Length on A_u & DI (f) Eccentricity on A_u & DI

Fig. 13 Influence of initial imperfections on compressive behaviors of RDD-2



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Fig. 14 First six buckling modes of SCSMIFs

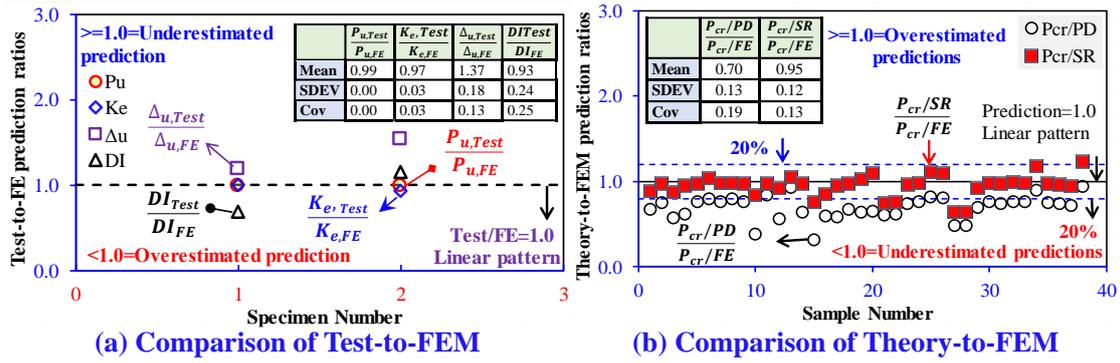
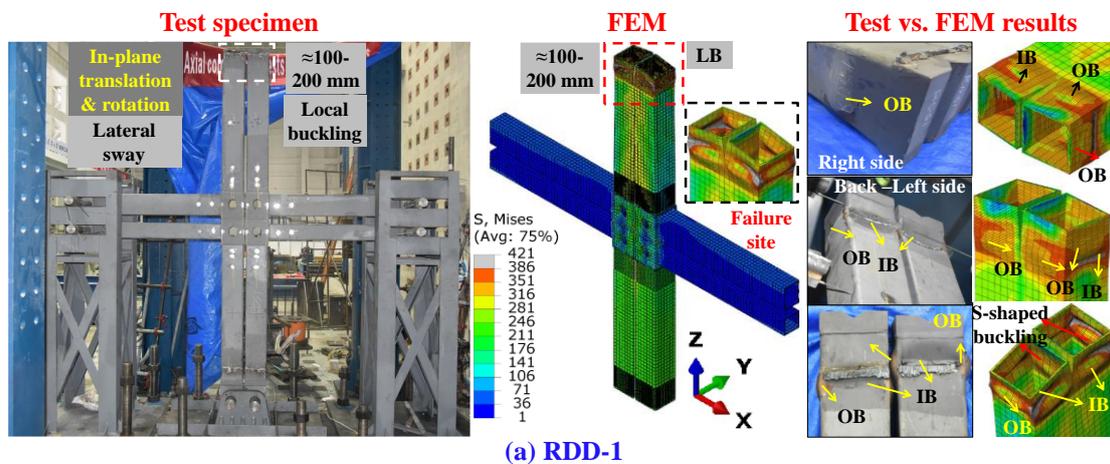


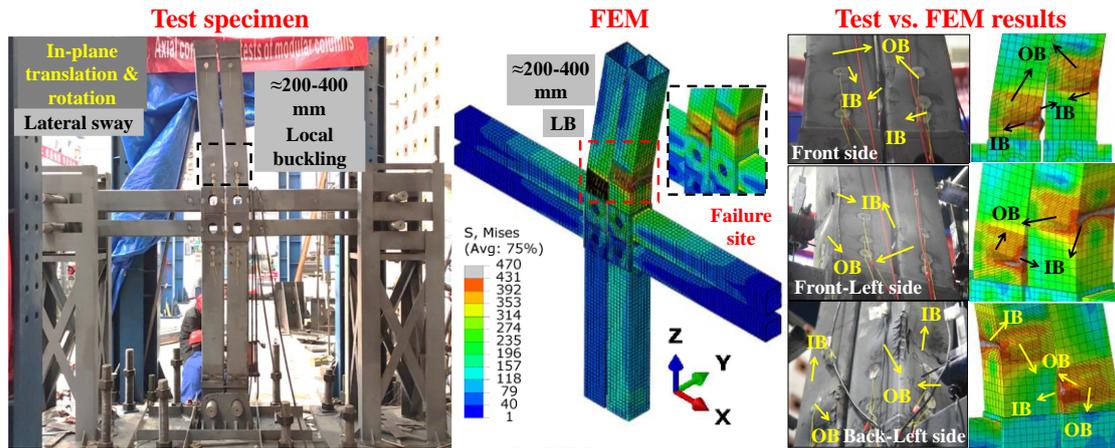
Fig. 15 Validations of FEMs and theoretical models outcomes

4.5 Validations

Figures 7(a,b) and 15(a), as well as Table 1, show the average estimates for $P-\Delta$ curves, P_u , K_e , Δ_u , and DI produced by the FEMs for two tests. The results indicate that the FEMs exhibited average modest prediction errors of 1%, 3.2%, and 7.9% for P_u , K_e , and DI , respectively. However, there was a significant scattering of 36.6% for Δ_u , mainly due to FEMs simplifications, soft supports, material modeling, and variations in imperfections. The developed FEM can accurately simulate the deformed shapes of SCSMIFs, including S-shaped inward and outward local buckling on all faces at the adjacent upper columns top areas in RDD-1 and lower zones with equal sway in RDD-2, as depicted in Fig. 16(a,b). These validations demonstrate that the proposed FEMs can reliably predict the compressive behavior of rotary-connected SCSMIFs and can be used for further extensive parametric and theoretical investigations.

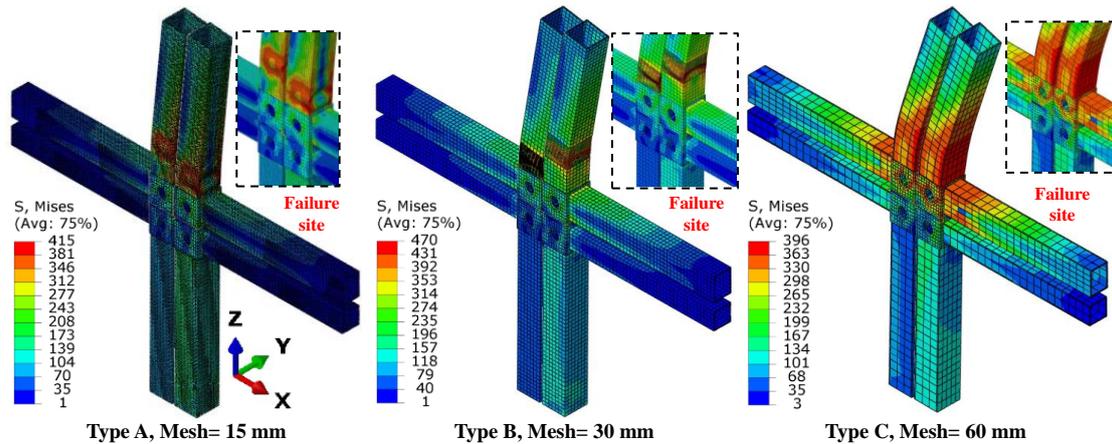


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(b) RDD-2



(c) Mesh effect on RDD-2

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Fig. 16 Comparison of test and FE-predicted failure modes

512

5 Parametric analysis

513

Experimental and numerical validations revealed that RDD-2, with significant sway of the SCSMIFs, possessed lower strength, stiffness, and ductility than RDD-1. Therefore,

514

RDD-2 FEM was used for further parametric investigation. It allows for a more

515

conservative design, better accounting for uncertainties and variations in material

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properties, fabrication, and installation, ensuring safety in design. The validated FEM

517

produced data for 87 rotary-connected SCSMIFs, maintaining the dimensions of the

518

rotary-type IMCs and corner fittings, 30 mm Type B mesh, local imperfection of $H/600$

519

or $0.64t_c$, and global imperfection of $e=3D_c/100$. The parametric analysis involved

520

beam and column sizes, lengths, gaps, quantities, and connecting plate thicknesses. The

521

load-shortening behavior of parameters is presented in Fig. 17(a~i). Supplementary

522

Figures B1(a~l), B2(a~i), and B3(a~i) illustrate the failure modes, $P_u(K_e)$ trends, and

523

failure modes, $P_u(K_e)$ trends, and

524 Δ_u (DI) trends observed in all 87 SCSMIFs. Moreover, **Fig. 18(a~f)** classifies typical
525 behaviors observed in parametric studies, which resemble the test results, indicating
526 that local buckling and lateral sway are predominantly linked to observed behaviors.
527 Similarly, **Fig. 19(a~d)** depicts the failure modes of SCSMIFs with varying quantities
528 of columns. **Supplementary Table A1** outlines each parameter's design details and
529 values for P_u , K_e , Δ_u , and DI .

530 **5.1 Beams cross-sections** ($D_{FB} \times B_{FB} \times t_{FB}$; $D_{CB} \times B_{CB} \times t_{CB}$)

531 **Figure 17(a)** shows how SCSMIF compressive behavior is affected by D_{FB} , D_{CB} , B_{FB} ,
532 B_{CB} , t_{FB} , and t_{CB} , measuring 150 and 200 and 6 and 8 mm while retaining other members'
533 dimensions as prototype design. The outcomes reveal that increasing beam cross-
534 sections improves SCSMIF performance by enhancing P_u and Δ_u in the 4~9% and 3~10%
535 range but lowers DI by 1~20%. When beams' width, depth, and thickness are raised, it
536 boosts structural integrity and prevents premature buckling, allowing SCSMIFs to
537 deform more before reaching their capacity (as seen in **Supplementary Figure B1(a)**).
538 However, there may be a compromise with post-ultimate ductility; therefore,
539 synchronizing these factors is essential for maximizing the performance of SCSMIFs.

540 **5.2 Beams lengths** (L_{FB} ; L_{CB})

541 **Figure 17(b)** illustrates increasing beam lengths from 0.6 to 1.2 and 3 m for a given
542 D_{FB} , D_{CB} , B_{FB} , and B_{CB} of 150 and 200 mm and t_{FB} and t_{CB} of 8 mm while keeping other
543 members consistent with prototype design can negatively impact SCSMIFs P_u (K_e) by
544 impairing them up to the range of 2~16% (1~2%). This is attributed to the increased
545 slenderness of longer beams, which reduces their bending resistance and can lead to
546 premature buckling, hindering the SCSMIFs from reaching their ultimate capacity.
547 Additionally, it can marginally impair ductility by reducing Δ_u (DI) up to 13% (4%).
548 This is because premature instability can reduce the ability to deform plastically,
549 leading to reduced ductility, as shown in **Supplementary Figure B1(b)**. This highlights

550 the significance of maintaining sufficient ductility for effective energy dissipation and
551 resilience of SCSMIF under severe load conditions.

552 **5.3 Columns lengths (L_c)**

553 **Figure 17(c)** reveals that elongating the columns from 0.6 to 1.2 and 3 m for a given
554 L_{FB} and L_{CB} of 0.6, 1.2, and 3 m, D_{FB} , D_{CB} , B_{FB} , and B_{CB} of 150 and 200 mm, and t_{FB}
555 and t_{CB} of 8 mm while keeping other members constant can impair the frames'
556 compressive behavior by decreasing $P_u (K_e)$ up to 68% (76%). This is because longer
557 columns become more slender, reducing their resistance to buckling and bending and
558 increasing deflection and bending stresses, lowering the overall load-carrying capacity
559 of SCSMIFs. It can increase Δ_u by up to 105% due to force redistribution, allowing for
560 more lateral deflection before approaching the buckling limit, as seen in
561 **Supplementary Figure B1(c)**. This highlights the need for careful column length
562 selection in coordination with other members during design.

563 **5.4 Columns cross-sections ($D_c \times B_c$)**

564 **Figure 17 (d)** indicates that, while keeping other members constant, increasing the
565 column cross-sections from 150 to 180, 200, and 210 mm for a given L_c , L_{FB} , and L_{CB}
566 of 1.2, 2.5, and 3.6 m and t_c , t_{FB} , and t_{CB} of 8 mm can improve SCSMIF compressive
567 performance by raising their $P_u (K_e)$ by up to 155% (116%). However, it might also
568 have adverse effects as it can lower $\Delta_u (DI)$ by as much as 16% (41%). Increased D_c
569 and B_c improve the columns' resistance to buckling and bending and reduce their
570 slenderness, strengthening SCSMIFs. However, increasing D_c/t_c , as shown in
571 **Supplementary Figure B1(d)**, reduces the frame members' flexibility, lowers buckling
572 strain, and diminishes the SCSMIFs' ductility.

573 **5.5 Columns thickness (t_c)**

574 As depicted in **Fig. 17(e)**, column thickness variations can influence the compression
575 behavior of SCSMIFs. Results indicate that increasing the thickness of the cross-section

576 of columns from 6 to 8 and 10 mm for a given D_c and B_c of 150, 180, 200, and 210 mm
577 can enhance the performance of SCSMIFs by increasing their P_u (K_e) by up to 188%
578 (93%) and Δ_u (DI) by up to 60% (96%). This is a result of the reduction in D_c/t_c , which
579 increases the buckling and bending resistance of the columns while decreasing their
580 slenderness. **Supplementary Figure B1(e)** reveals that SCSMIFs can withstand more
581 significant plastic deformations before failure, ultimately increasing their compressive
582 strength, buckling strain, and ductility. This emphasizes the importance of considering
583 column dimensions carefully during SCSMIFs' design for robust and resilient MSBs.

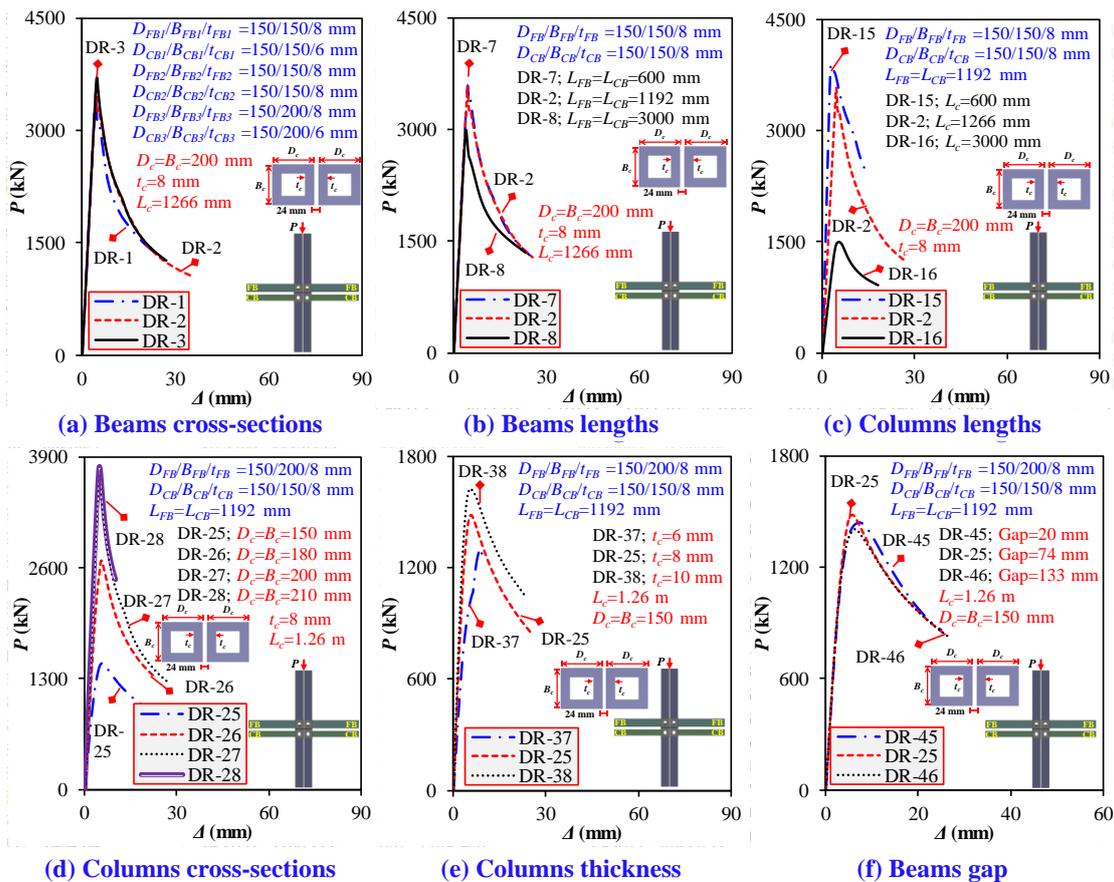
584 **5.6 Beams gap, connecting plate thickness, and columns gap**

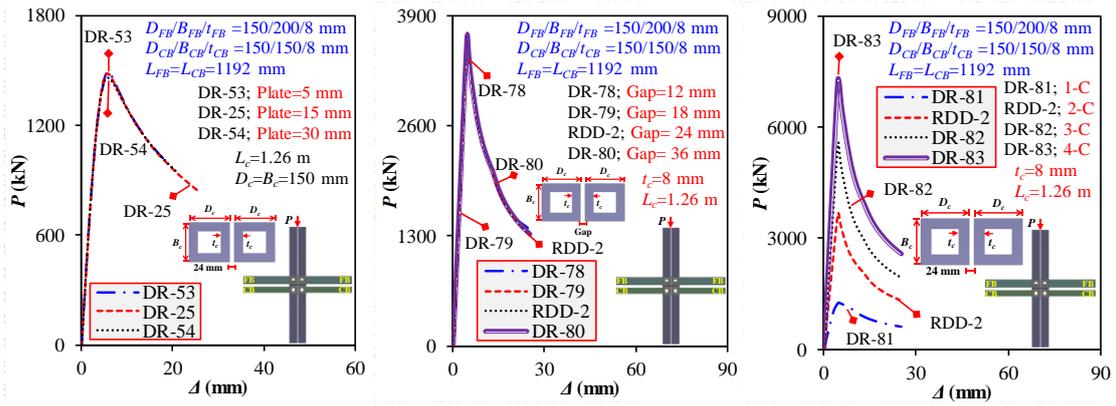
585 **Supplementary Table A1** and **Figs. 17(f~h)** and **Supplementary Figure B1(f,g,k)**
586 demonstrate that as long as the SCSMIFs retain the stiffness of members and IMCs per
587 the prototype design, limited increases in the gap between beams (from 20 to 74 and
588 133 mm), the thickness of the connecting plate (from 5 to 15 and 30 mm), and the gap
589 between columns (from 12 to 24 and 36 mm) do not have a significant impact on the
590 compressive behavior of SCSMIFs. They resulted in a maximum 3% decrease in
591 strength. Columns in SCSMIFs are capable of withstanding compressive forces,
592 preventing excessive deformation or local buckling of other members. Despite the
593 impact of modular gaps on lateral stability, SCSMIFs can still resist compressive forces.
594 Moreover, increasing the gap between columns increases the ductility by up to 2%,
595 allowing lateral movement. These findings demonstrate that rotary-type IMCs transfer
596 compressive stresses effectively, preserving the integrity of SCSMIFs [27,88].

597 **5.7 Columns quantity**

598 **Figures 17(i)** and **19(a~d)** illustrate the $P-\Delta$ curves and failure modes, highlighting that
599 P_u and K_e improve linearly as the column number increases from 1 to 2, 3, and 4. For
600 given D_{FB} and D_{CB} (150 mm), B_{FB} and B_{CB} (200 and 150 mm), and varying t_{FB} and t_{CB}

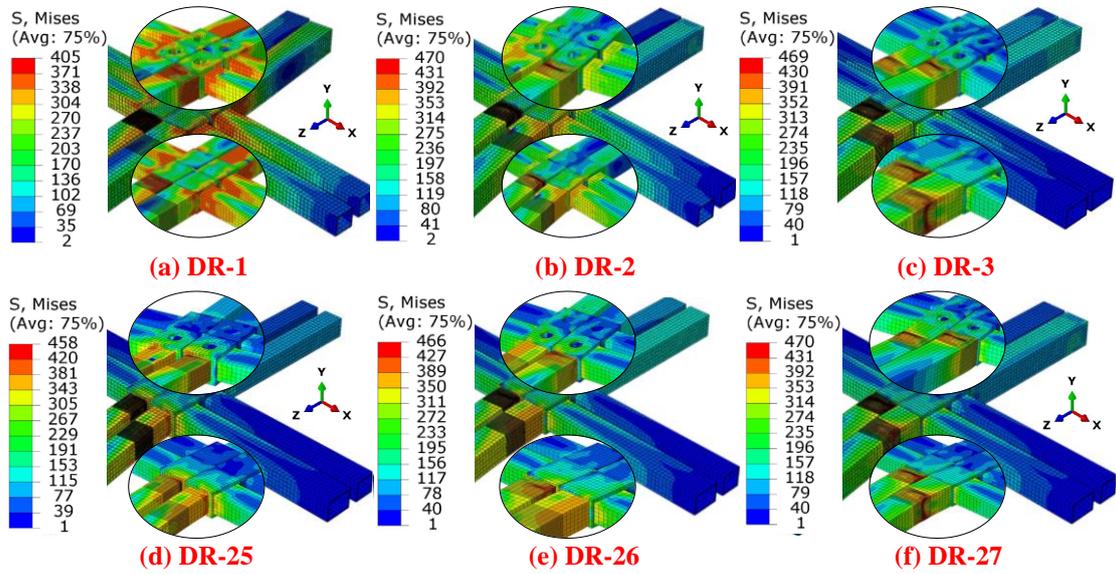
601 (8 and 6 mm), P_u (K_e) raises by 193% (142%), 345% (264%), 482% (384%) and 191%
 602 (132%), 341% (251%), 477% (367%). The relationship between Δ_u is less pronounced
 603 due to data fluctuation. However, DI is significantly reduced by up to 38% and 40%.
 604 The substantial increase in P_u by 2.9, 4.4, and 5.8 times, and K_e by 2.3, 3.5, and 4.7
 605 times confirms the positive effect of members' grouping. Increasing the number of
 606 columns enhances the compressive behavior of the SCSMIF by more than 2, 3, and 4
 607 times when advancing from a single to a double, triple, or quadruple-grouped frame
 608 skeleton. The failure mode remains on the lower end of the upper columns, indicating
 609 that increasing the number of frame skeletons distributes the load, reduces the stress on
 610 individual components, and increases the overall capacity of the SCSMIF without
 611 altering the failure pattern.





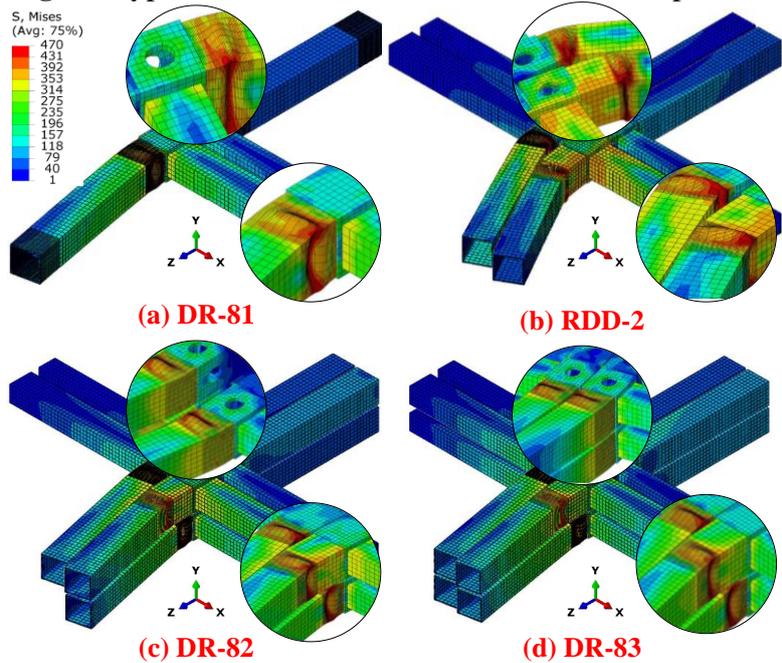
614
615

(g) Connecting plate thickness (h) Columns gap (i) Columns quantity
Fig. 17 Influence of varying parameters on typical load-shortening curves



616
617

Fig. 18 Typical failure modes SCSCMIFs under compression



618
619

Fig. 19 Influence of varying column quantity on failure modes of SCSCMIFs

620 **6 Theoretical investigation of buckling load for rotary-connected SCSMIFs**

621 The observed failure mechanism disclosed an S-shaped pattern marked by local inward
622 and outward buckling, exhibiting both elastic and plastic failure modes. It reveals that
623 the upper columns did not satisfy the EC3 Class 3 slenderness requirements, as elastic
624 buckling was not permitted, stopping the cross-section from achieving complete
625 yielding. Local buckling is believed to substantially affect the cross-sectional and
626 member capacities, regardless of whether inelastic or elastic [6]. Therefore, global
627 strength accounting for the radius of gyration, elastic buckling stress, and strength
628 reduction under these conditions produces more conservative results than the cross-
629 sectional strength [66]. Identical design practices were observed in other studies, like
630 Refs. [119,120] employed global strength prediction equations for member design,
631 taking yield strength failure with local buckling of Class 3 steel columns into account.
632 Ref. [121] used local buckling reduction factors for fixed-ended short columns.
633 Moreover, the member buckling strength was the primary design strength criterion used
634 in Ref. [122]. In addition, the global buckling strength model was applied to simple-
635 supported, concentrically compressed members in Ref. [123]. According to IS800
636 [54,55], NZS 3404[56], EC3:1-1 [57], CSA S16-19 [58], AISC360-16 [59], and GB
637 50017-2017 [60], the effective length factor played a crucial role in stability design,
638 which was dependent on the degree of elastic restraint at the column's ends. Because
639 MSB columns have semi-rigid connections at their ends, making them unique.
640 Therefore, their effective length factor and buckling load vary based on the relative
641 joint and member bending stiffness ratio and the rigidity of vertical and horizontal IMCs.
642 Considering the global strength rather than the cross-sectional strength is always
643 preferred to account for these factors effectively [49]. Thus, it is reasonable to
644 determine the buckling strength to account for the local elastic and plastic buckling.

645 Chen et al. [70], further highlighted that insufficient connection stiffness would result
646 in a greater slenderness of the MSB columns, highlighting the need for stability analysis
647 to evaluate buckling performance and determine influential design factors for columns
648 in MSBs. Therefore, the stability design methods have utilized the SCSMIFs' global
649 strength to achieve a buckling load more conservatively than a cross-section resistance
650 design.

651 Assuming the ITCs are fixed, the sub-assembled SCSMIFs shown in **Fig. 20(a)** were
652 analyzed for buckling load using three-story full-scale models in **Fig. 20(b)**. Pin and
653 semi-rigid IMCs were employed, along with stability functions from Eqns—1 and 2
654 introduced in Ref. [124] and buckling load from Eqn. 3 in Ref. [83].

$$S_{ii} = \frac{\left(\frac{\pi}{\mu}\right)^2 - \frac{\pi}{\mu} \sin \frac{\pi}{\mu}}{2 - 2 \cos \frac{\pi}{\mu} - \frac{\pi}{\mu} \sin \frac{\pi}{\mu}}; S_{ij} = \frac{\frac{\pi}{\mu} \sin \frac{\pi}{\mu} - \left(\frac{\pi}{\mu}\right)^2 \cos \frac{\pi}{\mu}}{2 - 2 \cos \frac{\pi}{\mu} - \frac{\pi}{\mu} \sin \frac{\pi}{\mu}} \text{ for } c_1, c_2, c_3, c_4, c_5, \text{ and } c_6 \quad (1)$$

$$S_{ii} = 4; S_{ij} = 2 \text{ for } b_1, b_2, b_3, b_4, b_5, b_6, b_7, \text{ and } b_8 \quad (2)$$

$$P_{cr} = \left[\frac{\pi^2 EI_{c2}}{(2\mu L_{ct})^2} \right] \quad (3)$$

655 **6.1 Pinned IMCs**

656 The moments in each member via Eqns. 4~7 and their equilibrium at joints A, B, and
657 sway for the target column c_2 in **Fig. 20(c)** are determined using Eqn. 8 according to
658 Chen et al.'s model [83] as follows:

$$(M_A)_{c2} = \left(\frac{EI_{c2}}{L_{c2}} \right) [S_{ii}\theta_A + S_{ij}\theta_B - (S_{ii} + S_{ij})\Delta_c/L_{ct}] \quad (4)$$

$$(M_B)_{c2} = \left(\frac{EI_{c2}}{L_{c2}} \right) [S_{ij}\theta_A + S_{ii}\theta_B - (S_{ii} + S_{ij})\Delta_c/L_{ct}] \quad (5)$$

$$(M_A)_{b2} = \left(\frac{EI_{b2}}{L_{b2}} \right) [4\theta_A + 2\theta_B] = \left(\frac{EI_{b2}}{L_{b2}} \right) [6\theta_A] \quad (6)$$

$$(M_B)_{b3} = \left(\frac{EI_{b3}}{L_{b3}} \right) [4\theta_B + 2\theta_A] = \left(\frac{EI_{b3}}{L_{b3}} \right) [6\theta_B] \quad (7)$$

$$(M_A)_{c2} + (M_A)_{b2} = 0; (M_B)_{c2} + (M_B)_{b3} = 0; (M_A)_{c2} + (M_B)_{c2} + P\Delta_c = 0 \quad (8)$$

$$\theta_A(S_{ii} + 6G_C) + \theta_B(S_{ij}) + \Delta_c/L_{ct} [-(S_{ii} + S_{ij})] = 0; G_C = \frac{EI_{b2}/L_{b2}}{EI_{c2}/L_{ct}} \quad (9)$$

$$\theta_A(S_{ij}) + \theta_B(S_{ii} + 6G_D) + \Delta_c/L_{ct} [-(S_{ii} + S_{ij})] = 0; G_D = \frac{EI_{b3}/L_{b3}}{EI_{c2}/L_{ct}} \quad (10)$$

$$\theta_A(S_{ii} + S_{ij}) + \theta_B(S_{ii} + S_{ij}) - \Delta_c/L_{ct} [2(S_{ii} + S_{ij}) - (\pi^2/\mu^2)] = 0 \quad (11)$$

659 where $P = \frac{\pi^2 EI_{c2}}{\mu^2 L_{c2}^2}$. By solving Eqns. 9~11 using a determinant, μ is calculated from Eqn.
660 12 and then substituted in Eqn. 3 to obtain the buckling load (P_{cr}/PD) of a pinned sub-
661 assembled rotary-connected SCSMIF.

$$\begin{aligned} (S_{ii} + 6G_C) \left[\left\{ \left(\frac{\pi^2}{\mu^2} \right) - 2(S_{ii} + S_{ij}) \right\} \times \{S_{ii} + 6G_D\} + (S_{ii} + S_{ij})^2 \right] \\ - (S_{ij}) \left[\{S_{ij}\} \times \left\{ \left(\frac{\pi^2}{\mu^2} \right) - 2(S_{ii} + S_{ij}) \right\} + (S_{ii} + S_{ij})^2 \right] \\ - (S_{ii} + S_{ij}) \left[(S_{ij}) \times (S_{ii} + S_{ij}) - (S_{ii} + S_{ij}) \times (S_{ii} + 6G_D) \right] = 0 \end{aligned} \quad (12)$$

662 6.2 Semi-rigid IMCs

663 6.2.1 Vertical rotary-type IMCs

664 According to Li et al.'s model [76], the rotary-connected SCSMIFs shown in **Fig. 20(d)**
665 experience double curvature bending, which results in equal beam end rotations, i.e.,
666 $\theta_B = \theta_G$; $\theta_C = \theta_H$; $\theta_D = \theta_I$; $\theta_E = \theta_J$. Additionally, the column end rotations are given by
667 $\theta_A = \theta_C - \frac{M_B}{R_{1v}} \times \frac{\theta_D}{\theta_C}$, $\theta_B = \theta_G = \theta_C - \frac{M_B}{R_{1v}}$, $\theta_E = \theta_J = \theta_D - \frac{M_E}{R_{2v}}$, and $\theta_F = \theta_D - \frac{M_E}{R_{2v}} \times \frac{\theta_C}{\theta_D}$. The
668 moments of the members can be determined via Eqns. 13~20 using slope-deflection
669 equations as follows:

$$(M_{BA})_{c1} = EI_{c1}/L_{ct} \left[S_{ii} \left(\theta_C - \frac{M_B}{R_{1v}} \right) + S_{ij} \left(\theta_D - \frac{M_B}{R_{1v}} \times \frac{\theta_D}{\theta_C} \right) - (S_{ii} + S_{ij}) \Delta_c / L_{ct} \right] \quad (13)$$

$$(M_{CD})_{c2} = EI_{c2}/L_{ct} \left[S_{ii} \theta_C + S_{ij} \theta_D - (S_{ii} + S_{ij}) \Delta_c / L_{ct} \right] \quad (14)$$

$$(M_{DC})_{c2} = EI_{c2}/L_{ct} \left[S_{ii} \theta_D + S_{ij} \theta_C - (S_{ii} + S_{ij}) \Delta_c / L_{ct} \right] \quad (15)$$

$$(M_{EF})_{c3} = EI_{c3}/L_{ct} \left[S_{ii} \left(\theta_D - \frac{M_E}{R_{2v}} \right) + S_{ij} \left(\theta_C - \frac{M_E}{R_{2v}} \times \frac{\theta_C}{\theta_D} \right) - (S_{ii} + S_{ij}) \Delta_c / L_{ct} \right] \quad (16)$$

$$(M_{BG})_{b1} = 6 \left(EI_{b1} / L_{b1} \right) \theta_B = 6 \left(EI_{b1} / L_{b1} \right) \left(\theta_C - \frac{M_B}{R_{1v}} \right) \quad (17)$$

$$(M_{CH})_{b2} = 6 \left(EI_{b2} / L_{b2} \right) \theta_C \quad (18)$$

$$(M_{DI})_{b3} = 6 \left(EI_{b3} / L_{b3} \right) \theta_D \quad (19)$$

$$(M_{EJ})_{b4} = 6 \left(EI_{b4} / L_{b4} \right) \theta_E = 6 \left(EI_{b4} / L_{b4} \right) \left(\theta_D - \frac{M_E}{R_{2v}} \right) \quad (20)$$

670 The equilibrium of moments at joints C and D and the sway can be determined by
671 considering c_2 as the objective column, using Eqns. 21 to 23, as shown below:

$$(M_{BA})_{c1} + (M_{BG})_{b1} + (M_{CH})_{b2} + (M_{CD})_{c2} = 0 \quad (21)$$

$$(M_{EF})_{c3} + (M_{EJ})_{b4} + (M_{DI})_{b3} + (M_{DC})_{c2} = 0 \quad (22)$$

$$(M_{CD})_{c2} + (M_{DC})_{c2} + P \Delta_c = 0 \quad (23)$$

672 When $P = \frac{\pi^2 EI_{c2}}{\mu^2 L_{c2}^2}$ is substituted into Eqn. 23, the resulting equation enables the
673 calculation of $\Delta_c/L_{ct} = \frac{\mu^2(S_{ii}+S_{ij})(\theta_c+\theta_D)}{-\pi^2+2(S_{ii}+S_{ij})\mu^2}$. Introducing Δ_c/L_{ct} , the equations (Eqns. 13~20),
674 the relative members', i.e., $G_{1v} = \frac{EI_{b1}/L_{b1}}{EI_{c3}/L_{ct}}$, $G_{2v} = \frac{EI_{b2}/L_{b2}}{EI_{c2}/L_{ct}}$, $G_{3v} = \frac{EI_{b3}/L_{b3}}{EI_{c2}/L_{ct}}$, $G_{4v} = \frac{EI_{b4}/L_{b4}}{EI_{c3}/L_{ct}}$, and the
675 IMC-to-members stiffnesses ratios, i.e., $J_{1v} = \frac{R_{1v}}{EI_{c1}/L_{ct}}$, $J_{2v} = \frac{R_{1v}}{EI_{c2}/L_{ct}}$, $J_{3v} = \frac{R_{2v}}{EI_{c2}/L_{ct}}$, $J_{4v} =$
676 $\frac{R_{2v}}{EI_{c3}/L_{ct}}$ into Eqns. 21 and 22. Then, by rearranging the equations in terms of θ_c^2 , θ_D^2 ,
677 and $\theta_c\theta_D$, the resulting equations (Eqns. 24 and 25) are obtained.

$$\begin{aligned} & \theta_c^2 \left[(6G_{1v} + S_{ii})(6G_{2v} + S_{ii}) + (6G_{1v}J_{2v} + 6G_{2v}J_{1v} + S_{ii}J_{2v} + S_{ii}J_{1v}) - \right. \\ & \quad \left. \frac{\mu^2}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} (6G_{1v} + S_{ii} + J_{1v} + J_{2v})(S_{ii} + S_{ij})^2 \right] + \theta_D^2 \left[S_{ij}^2 - \right. \\ & \quad \left. \frac{\mu^2}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} S_{ij}(S_{ii} + S_{ij})^2 \right] + \theta_c\theta_D \left[S_{ij}(J_{2v} + J_{1v} + 6G_{2v} + 2S_{ii} + 6G_{1v}) - \right. \\ & \quad \left. \frac{\mu^2}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} (6G_{1v} + S_{ii} + J_{1v} + J_{2v})(S_{ii} + S_{ij})^2 - \frac{\mu^2}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} S_{ij}(S_{ii} + \right. \\ & \quad \left. S_{ij})^2 \right] = 0 \end{aligned} \quad (24)$$

$$\begin{aligned} & \theta_D^2 \left[(6G_{3v} + S_{ii})(6G_{4v} + S_{ii}) + (6G_{4v}J_{3v} + 6G_{3v}J_{4v} + S_{ii}J_{4v} + S_{ii}J_{3v}) - \right. \\ & \quad \left. \frac{\mu^2}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} (6G_{4v} + S_{ii} + J_{4v} + J_{3v})(S_{ii} + S_{ij})^2 \right] + \theta_c^2 \left[S_{ij}^2 - \right. \\ & \quad \left. \frac{\mu^2}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} S_{ij}(S_{ii} + S_{ij})^2 \right] + \theta_c\theta_D \left[S_{ij}(J_{3v} + J_{4v} + 6G_{3v} + 2S_{ii} + 6G_{4v}) - \right. \\ & \quad \left. \frac{\mu^2}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} (6G_{4v} + S_{ii} + J_{4v} + J_{3v})(S_{ii} + S_{ij})^2 - \frac{\mu^2}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} S_{ij}(S_{ii} + \right. \\ & \quad \left. S_{ij})^2 \right] = 0 \end{aligned} \quad (25)$$

$$(\theta_c + \beta_1\theta_D)(\beta_2\theta_c + \beta_3\theta_D) = 0; (\beta_4\theta_c + \theta_D)(\beta_5\theta_c + \beta_6\theta_D) = 0 \quad (26)$$

678 Eqn. 26 provides a simplified version of Eqns. 24 and 25, which yield four possible
679 general solutions: $\begin{vmatrix} 1 & \beta_1 \\ \beta_4 & 1 \end{vmatrix} = 0$, $\begin{vmatrix} 1 & \beta_1 \\ \beta_5 & \beta_6 \end{vmatrix} = 0$, $\begin{vmatrix} \beta_2 & \beta_3 \\ \beta_4 & 1 \end{vmatrix} = 0$, and $\begin{vmatrix} \beta_2 & \beta_3 \\ \beta_5 & \beta_6 \end{vmatrix} = 0$. These general
680 solutions are utilized to obtain the simplified expressions presented in Eqns. 27~30 after
681 solving the determinant. The maximum value obtained from Eqns. 27~30 determines
682 the maximum μ value, which is then inserted into Eqn. 3 to calculate the minimum
683 buckling load (P_{cr}/SR) of a sub-assembled rotary-connected SCSMIF with vertical
684 IMC.

$$685 \quad 1 - \left[\frac{2[S_{ij}^2 - D]}{[S_{ij}(E) - C - D] + \sqrt{[S_{ij}(E) - C - D]^2 - 4[S_{ij}^2 - D][A + B - C]}} \right] \left[\frac{2[S_{ij}^2 - I]}{[S_{ij}J - H - I] + \sqrt{[S_{ij}J - H - I]^2 - 4[F + G - H][S_{ij}^2 - I]}} \right] = 0 \quad (27)$$

$$686 \quad [F + G - H] - \left[\frac{2[S_{ij}^2 - D]}{[S_{ij}(E) - C - D] + \sqrt{[S_{ij}(E) - C - D]^2 - 4[S_{ij}^2 - D][A + B - C]}} \right] \left[\frac{[S_{ij}J - H - I] + \sqrt{[S_{ij}J - H - I]^2 - 4[F + G - H][S_{ij}^2 - I]}}{2} \right] =$$

$$687 \quad 0 \quad (28)$$

$$688 \quad [A + B - C] - \left[\frac{[S_{ij}(E) - C - D] + \sqrt{[S_{ij}(E) - C - D]^2 - 4[S_{ij}^2 - D][A + B - C]}}{2} \right] \left[\frac{2[S_{ij}^2 - I]}{[S_{ij}J - H - I] + \sqrt{[S_{ij}J - H - I]^2 - 4[F + G - H][S_{ij}^2 - I]}} \right] =$$

$$689 \quad 0 \quad (29)$$

$$690 \quad [A + B - C][F + G - H] -$$

$$691 \quad \left[\frac{[S_{ij}(E) - C - D] + \sqrt{[S_{ij}(E) - C - D]^2 - 4[S_{ij}^2 - D][A + B - C]}}{2} \right] \left[\frac{[S_{ij}J - H - I] + \sqrt{[S_{ij}J - H - I]^2 - 4[F + G - H][S_{ij}^2 - I]}}{2} \right] = 0 \quad (30)$$

692 Coefficients A , B , C , D , E , F , G , and H are obtained from Eqns. 31~34, which are
693 defined in terms of θ_c^2 , θ_D^2 , and $\theta_c\theta_D$ in Eqns. 24 and 25. The value of $R_{1v} =$
694 $R_{2v} = 2391.49$ kNm/rad is the rotational stiffness of a rotary-type IMC, as stated in Refs.
695 [9].

$$A = (6G_{1v} + S_{ii})(6G_{2v} + S_{ii}); B = (6G_{1v}J_{2v} + 6G_{2v}J_{1v} + S_{ii}J_{2v} + S_{ii}J_{1v}) \quad (31)$$

$$C = \frac{\mu^2(6G_{1v} + S_{ii} + J_{1v} + J_{2v})(S_{ii} + S_{ij})^2}{-\pi^2 + 2(S_{ii} + S_{ij})\mu^2}; D = \frac{\mu^2 S_{ij}(S_{ii} + S_{ij})^2}{-\pi^2 + 2(S_{ii} + S_{ij})\mu^2} \quad (32)$$

$$E = (J_{2v} + J_{1v} + 6G_{2v} + 2S_{ii} + 6G_{1v}); F = (6G_{3v} + S_{ii})(6G_{4v} + S_{ii}) \quad (33)$$

$$G = (6G_{4v}J_{3v} + 6G_{3v}J_{4v} + S_{ii}J_{4v} + S_{ii}J_{3v}); H = \frac{\mu^2(6G_{4v} + S_{ii} + J_{4v} + J_{3v})(S_{ii} + S_{ij})^2}{-\pi^2 + 2(S_{ii} + S_{ij})\mu^2} \quad (34)$$

696 6.2.2 Horizontal rotary-type IMCs

697 In order to find the impact of adjacent module members' and horizontal IMCs' stiffness
698 per Li et al.'s model [76], the end rotations of beams are equal, specifically $\theta_L =$
699 θ_Q ; $\theta_M = \theta_R$; $\theta_N = \theta_S$; $\theta_O = \theta_T$. Additionally, column end rotations are $\theta_M = \theta_c + \frac{M_C}{R_{1h}}$,
700 $\theta_N = \theta_D - \frac{M_D}{R_{2h}}$. Moments of the members can be expressed with slope-deflection

701 equations using Eqns. 35~37 as follows:

$$(M_{MR})_{b6} = 6 \left(\frac{EI_{b6}}{L_{b6}} \right) \left(\theta_c + \frac{M_C}{R_{1h}} \right) \quad (35)$$

$$(M_{NS})_{b7} = 6 \left(\frac{EI_{b7}}{L_{b7}} \right) \left(\theta_D + \frac{M_D}{R_{2h}} \right) \quad (36)$$

$$(M_{MN})_{c5} = \frac{EI_{c5}}{L_{ct}} \left[S_{ii} \left(\theta_c + \frac{M_C}{R_{1h}} \right) + S_{ij} \left(\theta_D + \frac{M_D}{R_{2h}} \right) - (S_{ii} + S_{ij}) \Delta_c / L_{ct} \right] \quad (37)$$

702 As depicted in **Fig. 20(d)**, when c_5 is selected as the objective column, the moments'
703 equilibrium at joints M and N can be determined through Eqns. 38 and 39 as follows:

$$(M_{MR})_{b6} + (M_{MN})_{c5} + (M_{CH})_{b2} + (M_{CD})_{c2} = 0 \quad (38)$$

$$(M_{MN})_{c5} + (M_{NS})_{b7} + (M_{DC})_{c2} + (M_{DI})_{b3} = 0 \quad (39)$$

704 By inputting $\Delta_c/L_{ct} = \frac{\mu^2(S_{ii}+S_{ij})(\theta_c+\theta_D)}{-\pi^2+2(S_{ii}+S_{ij})\mu^2}$ obtained from Eqn. 23, along with members'

705 relative stiffness ratios, such as $G_{1h} = \frac{EI_{b6}/L_{b6}}{EI_{c5}/L_{ct}}$, $G_{2h} = \frac{EI_{b7}/L_{b7}}{EI_{c5}/L_{ct}}$, $G_{2v} = \frac{EI_{b2}/L_{b2}}{EI_{c2}/L_{ct}}$, $G_{3v} = \frac{EI_{b3}/L_{b3}}{EI_{c2}/L_{ct}}$,

706 and $K = \frac{EI_{c2}/L_{ct}}{EI_{c5}/L_{ct}}$, and the IMC-to-member relative stiffness ratios, i.e., $J_{1h} = \frac{R_{1h}}{EI_{c2}/L_{ct}}$ and

707 $J_{2h} = \frac{R_{2h}}{EI_{c5}/L_{ct}}$ into Eqns. 38 and 39, it is possible to rearrange equations in terms of θ_c and

708 θ_D to obtain Eqns. 40 and 41.

$$\begin{aligned} & \theta_c \left[J_{1h}J_{2h}(6G_{1h} + S_{ii} + 6G_{2v}K + KS_{ii}) + J_{2h}(36G_{1h}G_{2v} + 6G_{1h}S_{ii} + 6G_{2v}S_{ii} + \right. \\ & S_{ii}^2) - J_{1h}S_{ij}^2 - \frac{\mu^2(S_{ii}+S_{ij})}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} \{ J_{1h}J_{2h}(S_{ii} + S_{ij} + KS_{ii} + KS_{ij}) + J_{2h}(6G_{1h}S_{ii} + \\ & 6G_{1h}S_{ij} + S_{ii}^2) + S_{ii}S_{ij}(J_{2h} - J_{1h}) - J_{1h}S_{ij}^2 \} \left. \right] + \theta_D \left[J_{1h}J_{2h}S_{ij}(1 + K) + \right. \\ & J_{2h}S_{ij}(6G_{1h} + S_{ii}) - J_{1h}S_{ij}(6G_{3v} - S_{ii}) - \frac{\mu^2(S_{ii}+S_{ij})}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} \{ J_{1h}J_{2h}(S_{ii} + S_{ij} + \\ & KS_{ii} + KS_{ij}) + J_{2h}(6G_{1h}S_{ii} + 6G_{1h}S_{ij} + S_{ii}^2) + S_{ii}S_{ij}(J_{2h} - J_{1h}) - J_{1h}S_{ij}^2 \} \left. \right] = 0 \\ & \theta_c \left[J_{1h}J_{2h}(1 + K)S_{ij} + J_{2h}S_{ij}(6G_{2v} + S_{ii}) - J_{1h}S_{ij}(S_{ii} + 6G_{2h}) - \right. \\ & \frac{\mu^2(S_{ii}+S_{ij})}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} \{ J_{1h}J_{2h}(S_{ii} + S_{ij} + KS_{ii} + KS_{ij}) - J_{1h}(S_{ii}^2 + 6S_{ii}G_{2h} + 6S_{ij}G_{2h}) + \\ & J_{2h}(S_{ij}^2) + S_{ii}S_{ij}(J_{2h} - J_{1h}) \} \left. \right] + \theta_D \left[J_{1h}J_{2h}(S_{ii} + 6G_{2h} + 6KG_{3v} + KS_{ii}) - \right. \\ & J_{1h}(6G_{3v}S_{ii} + S_{ii}^2 + 36G_{3v}G_{2h} + 6S_{ii}G_{2h}) + J_{2h}S_{ij}^2 - \\ & \frac{\mu^2(S_{ii}+S_{ij})}{-\pi^2+2(S_{ii}+S_{ij})\mu^2} \{ J_{1h}J_{2h}(S_{ii} + S_{ij} + KS_{ii} + KS_{ij}) - J_{1h}(S_{ii}^2 + 6S_{ii}G_{2h} + 6S_{ij}G_{2h}) + \\ & J_{2h}(S_{ij}^2) + S_{ii}S_{ij}(J_{2h} - J_{1h}) \} \left. \right] = 0 \\ & \theta_c[\xi_1] + \theta_D[\xi_2] = 0; \theta_c[\beta_3] + \theta_D[\beta_4] = 0 \end{aligned} \quad (40)$$

$$\theta_c[\xi_1] + \theta_D[\xi_2] = 0; \theta_c[\beta_3] + \theta_D[\beta_4] = 0 \quad (41)$$

709 Eqn. 42 is a simplified form of Eqns. 40 and 41, and its general solution is $\begin{bmatrix} \xi_1 & \xi_2 \\ \xi_3 & \xi_4 \end{bmatrix} = 0$.

710 After determining the general solution using Eqn. 42, the μ value can be obtained and

711 then inserted into Eqn. 3 to calculate the buckling load (P_{cr}/SR) of a sub-assembled

712 SCSMIF considering horizontal IMCs and adjacent modular frame members stiffnesses

713 effect. The coefficients ξ_1 , ξ_2 , ξ_3 , and ξ_4 are derived from Eqns. 43~46, which are

714 expressed in terms of θ_c and θ_D as defined in Eqns. 40 and 41.

$$\begin{aligned} \xi_1 = & J_{1h}J_{2h}(6G_{1h} + S_{ii} + 6G_{2v}K + KS_{ii}) \\ & + J_{2h}(36G_{1h}G_{2v} + 6G_{1h}S_{ii} + 6G_{2v}S_{ii} + S_{ii}^2) - J_{1h}S_{ij}^2 - Z_1 \end{aligned} \quad (43)$$

$$\xi_2 = J_{1h}J_{2h}S_{ij}(1 + K) + J_{2h}S_{ij}(6G_{1h} + S_{ii}) - J_{1h}S_{ij}(6G_{3v} - S_{ii}) - Z_1 \quad (44)$$

$$\xi_3 = J_{1h}J_{2h}(1 + K)S_{ij} + J_{2h}S_{ij}(6G_{2v} + S_{ii}) - J_{1h}S_{ij}(S_{ii} + 6G_{2h}) - Z_2 \quad (45)$$

$$\xi_4 = J_{1h}J_{2h}(S_{ii} + 6G_{2h} + 6KG_{3v} + KS_{ii}) - J_{1h}(6G_{3v}S_{ii} + S_{ii}^2 + 36G_{3v}G_{2h} + 6S_{ii}G_{2h}) + J_{2h}S_{ij}^2 - Z_2 \quad (46)$$

$$Z_1 = \frac{\mu^2(S_{ii} + S_{ij})}{-\pi^2 + 2(S_{ii} + S_{ij})\mu^2} \{J_{1h}J_{2h}(S_{ii} + S_{ij} + KS_{ii} + KS_{ij}) + J_{2h}(6G_{1h}S_{ii} + 6G_{1h}S_{ij} + S_{ii}^2) + S_{ii}S_{ij}(J_{2h} - J_{1h}) - J_{1h}S_{ij}^2\} \quad (47)$$

$$Z_2 = \frac{\mu^2(S_{ii} + S_{ij})}{-\pi^2 + 2(S_{ii} + S_{ij})\mu^2} \{J_{1h}J_{2h}(S_{ii} + S_{ij} + KS_{ii} + KS_{ij}) - J_{1h}(S_{ii}^2 + 6S_{ii}G_{2h} + 6S_{ij}G_{2h}) + J_{2h}(S_{ij}^2) + S_{ii}S_{ij}(J_{2h} - J_{1h})\} \quad (48)$$

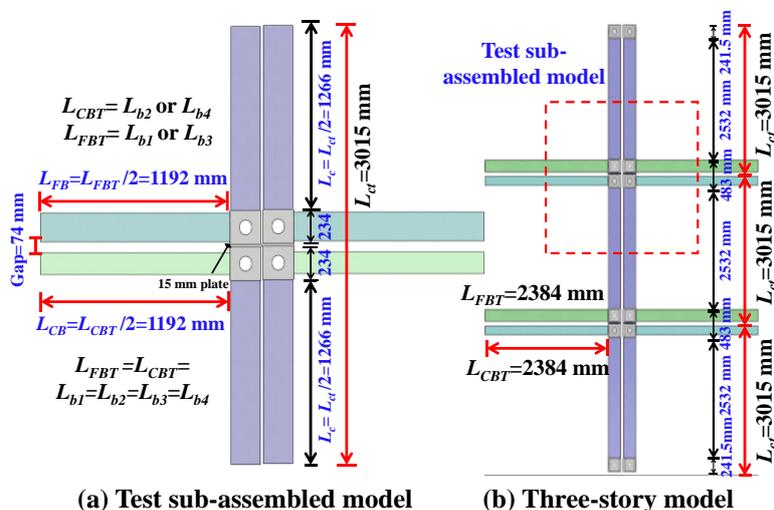
715 The coefficients Z_1 and Z_2 are obtained from Eqns. 47 and 48, and $R_{1h} = R_{2h} = 151$
716 kN/mm represents the shear stiffness of a horizontal rotary-type IMC as given in Refs.
717 [9,40].

718 6.3 Validations

719 The details of the FEMs' buckling loads (P_{cr}/FE) are compared with the theoretical
720 results (P_{cr}/PD and P_{cr}/SR) in **Fig. 15(b)** and **Table 2**. P_{cr}/PD represents the
721 buckling load obtained using the pinned IMC's Eqn. 12, while P_{cr}/SR corresponds to
722 the minimum buckling load obtained by solving vertical rotary-type IMCs' Eqn. 26 or
723 horizontal IMCs' Eqn. 42. Based on Refs. [9,40], $R_{1h} = R_{2h}$ is typically lower than
724 other components and vertical IMC rotational stiffness. Consequently, horizontal IMCs'
725 stiffness does not directly impact the buckling behavior, and its μ value has minimal
726 influence on the buckling behavior of SCSMIF. This suggests that the stability of
727 SCSMIF is primarily determined by the members' or the vertical IMC-to-members
728 relative stiffnesses and that adjacent frames behave independently. When the columns
729 are comparatively slender and the members' stiffnesses are lesser than horizontal IMC,
730 it is crucial to consider the effect of the horizontal IMC as it could reduce effective
731 length and increase buckling load. These findings are consistent with tests and literature,
732 indicating that adjacent frame behave independently without any deteriorating impact

733 [49]. The average (Cov) prediction ratios of the tested and parametric FEMs for P_{cr}/PD
734 and P_{cr}/SR are 0.71(0.05)/0.93(0.05) and 0.69(0.20)/0.95(0.16), offering conservative
735 results. However, P_{cr}/PD had a large scatter and underestimated findings over the
736 larger range, while P_{cr}/SR produced more accurate outcomes with minimal scatter.
737 Additionally, a safety factor could further improve slight overestimations. The results
738 showed that accurately anticipating the compressive behavior of sub-assembled rotary-
739 connected SCSMIFs could be achieved by considering the semi-rigidity and rotational
740 stiffness of rotary-type IMCs, as determined in Refs. [9]. Conversely, assuming rotary-
741 type IMCs to be pinned could not reflect their actual behavior and could result in an
742 uneconomical design.

743 Assuming pinned IMC results in conservative estimates for the buckling load that
744 applies to all SCSMIFs. On the other hand, considering the semi-rigidity of IMCs yields
745 more accurate predictions for rotary-connected SCSMIFs, including those for exterior,
746 middle, or interior frames. These findings were consistent with tests implying that
747 relative rotation between the upper and lower frame skeleton at rotary-type IMCs
748 cannot be simulated as pinned. However, these outcomes are specific to the models
749 used and require additional validation. Moreover, these results cannot be directly
750 applied to non-sway or special frames with welded IMCs or shear-keyed columns.



751

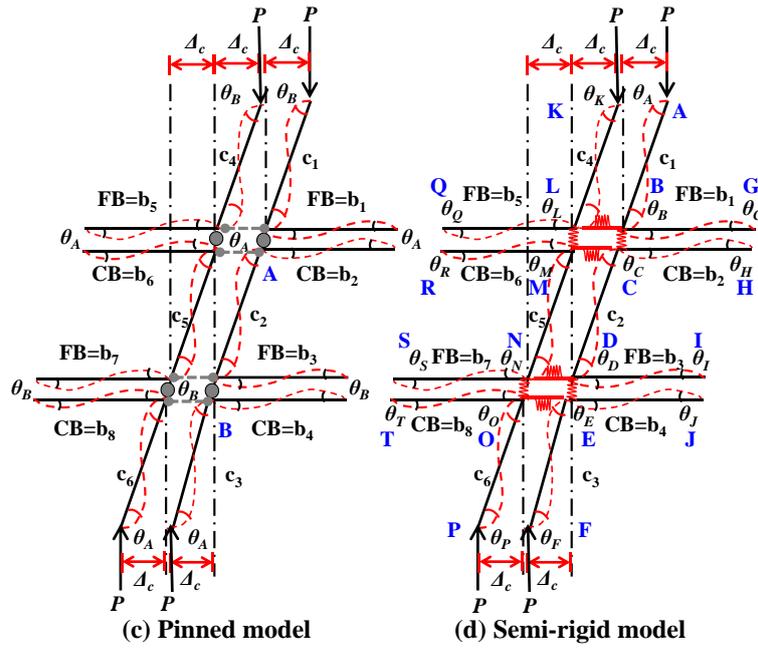


Fig. 20 Theoretical buckling load sub-assembled models; Per Chen et al.'s model [83] for pinned and Li et al.'s model [76] for semi-rigid rotary-connected SCSMIFs

7 Design guidelines and recommendations

Experimental, numerical, and theoretical results on the compressive performance of SCSMIFs with rotary-type IMCs add significantly to the structural integrity and stability of MSBs, allowing for conservative and cost-effective design. These results provide valuable insights into modular frame columns' global and local buckling under compression loads, contributing to significant revelations about these structures' failure modes and load-carrying capacity. When incorporating rotary-type IMCs into structural designs, it is essential to count on acquiring precise mechanical properties, such as rotational stiffness, strength, and behavioral characteristics of the vertical and horizontal IMCs from Refs. [9,40]. Selecting structural members with an appropriate cross-sectional dimension or shape and higher rigidity is crucial as this offers greater adaptability for meeting cross-sectional needs. The parametric analysis exposes that the primary strength-enhancing parameter is the cross-sectional sizes and thicknesses of columns and beams and adding more columns or adjacent modules. Moreover, applying IMCs with strengthened geometrical designs could optimize the load-bearing capacity

770 of the structures. Reliable decisions can be made concerning selecting and optimizing
771 the rigidities, geometries, and material properties of structural members and
772 connections to ensure structural integrity and satisfy the building's specific performance
773 requirements. The study generated FEMs and theoretical models to describe the
774 behavior of pinned and semi-rigid IMCs, which were rigorously analyzed and validated
775 using experimental and numerical data, resulting in accurate predictions of buckling
776 loads and potential failure modes. Using these predictive models in the design process
777 permits the identification of problem areas, such as high-stress concentrations or
778 unstable regions, thereby optimizing the design to guarantee the reliability and
779 structural integrity of the MSB. Consideration of the pinned nature produces a large
780 scatter, whereas the semi-rigidity of IMCs yields more precise predictions with less
781 scatter. Nevertheless, it is essential to acknowledge that the semi-rigid behavior of
782 IMCs in modular frames brings significant challenges in their design and can contribute
783 to complexities in modeling and analyzing these structures, which consumes
784 considerable time and resources [80,81]. Complex joint configurations can even result
785 in unpredictable behavior [82]. This study's theoretical models supply a categorization
786 system for pinned and semi-rigid IMCs, which can increase interconnection behavior
787 prediction and modular system dependability. These models are beneficial for
788 navigating the complexities and unpredictability of joint behavior, and they can guide
789 the efficient control of relative stiffnesses for a conservative and cost-effective design.
790 Even though this study provides essential guidance for designing rotary-connected
791 SCSMIFs, other studies can be carried out to compare the performance of different
792 types of IMCs.

793 **8 Conclusions**

794 This study conducted two sub-assembled tests, analyzed 87 parametric FEMs, and

795 developed theoretical buckling models using pinned and semi-rigid IMCs to estimate
796 the buckling load of rotary-connected SCSMIFs. The investigation aimed to understand
797 the compressive behavior of SCSMIFs and their effect on adjacent members. The study
798 resulted in the following findings:

- 799 1. The load-shortening of SCSMIFs exhibited elastic, inelastic, and recessional
800 properties. As the compressive resistance increased from yield to ultimate, local
801 buckling on the upper columns primarily occurred from the bending direction
802 towards the adjacent faces of the columns, associated with sway depending on
803 the relative rigidity of the members. After attaining their ultimate strength,
804 capacity decreased, and buckling became more intense.
- 805 2. The SCSMIFs displayed symmetrical S-shaped buckling patterns on all upper
806 column faces, occurring on either the top edges or lower areas at 100-200 mm.
807 The buckling pattern was similar on opposite faces but opposite on adjacent
808 faces of nearby columns. Additionally, the inner sides of grouped columns
809 either bulged out or in, resulting in double S-shaped buckling that prevented
810 collisions on the interior sides. However, the degree of sway and buckling
811 location depended on the members' relative stiffness.
- 812 3. The strain curves of the SCSMIFs revealed that S-shaped local buckling only
813 occurred on upper adjacent columns at the same location, either inward or
814 outward, symmetrically on opposite sides and oppositely on adjacent sides. The
815 columns in the direction of bending or the opposite direction demonstrated
816 elastic buckling, whereas the other areas showed plastic buckling. Other
817 members and rotary-type IMCs exhibited no yielding, fracturing, or failure,
818 indicating that the upper columns are the primary load-bearing members.

- 819 4. Increasing the cross-section of beams and columns improves the compressive
820 resistance of SCSMIFs, but lengthening them impairs. Greater member rigidity
821 can reduce ductility, but a significant difference in the relative stiffness of the
822 beam and column can lead to premature instability. Raising the column number
823 from 1 to 2, 3, and 4 can enhance strength by 2.9, 4.4, and 5.8 times and stiffness
824 by 2.3, 3.5, and 4.7 times. Changes in beam gap, connecting plate thickness, or
825 column gap did not significantly affect the compressive behavior, while rotary-
826 type IMCs sustained loading without failure.
- 827 5. With average prediction errors of 1% and 3.2% for P_u and K_e , the developed
828 FEMs with a mesh of 30 mm, local imperfection of $H/600$ or $0.64t_c$, and global
829 imperfection of $e=3D_c/100$ appropriately replicated SCSMIFs compression
830 behavior, demonstrating their ability to identify SCSMIFs' elastoplastic
831 behavior.
- 832 6. The mean (Cov) theory-to-FEM buckling loads were 0.70(0.19) and 0.9(0.13)
833 for pinned and semi-rigid SCSMIFs, suggesting that the semi-rigid model is a
834 more accurate predictor of compressive behavior for rotary-connected
835 SCSMIFs, as it produced more precise results with less scatter than the pinned
836 model, which had significant scatter. Non-identical deflections in tests imply a
837 relative rotation between the upper and lower frame skeleton at rotary-type
838 IMCs that can be better simulated as semi-rigid.

839 This study focused on standard member cross-sections of rotary-connected SCSMIFs,
840 with possible application to other semi-rigid IMCs using their vertical and horizontal
841 IMC stiffnesses or to all types of IMCs with developed pinned models. Future research
842 will investigate simplified FEMs for rotary-connected SCSMIFs based on the findings

843 of this study, utilizing vertical and horizontal IMCs as spring models to design more
844 practical multi-story MSBs.

845 **Acknowledgment**

846 The authors thank the National Natural Science Foundation of China (Grant No.
847 51978457 and 52008292) for their financial research support.

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